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Monterey, California



## THESIS

SHIP OUTLINE FEATURE SELECTION  
USING B-SPLINE FUNCTION

by

Werawong Thavamongkon

December 1984

Thesis Advisor:

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Ship Outline Feature Selection  
Using B-spline Function

by

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## ABSTRACT

This thesis presents two methods of ship classification with IR images. The first method is the Fourier Coefficient method which transform the sample points of a superstructure profile of a ship to the spatial frequency components. The shape of the coefficient curve can be used to classify the type of a ship from 8 different categories. But, the differences is so minor that it is difficult to implement a computer program to recognize it. The second method is a B-spline Coefficient method which uses the uneven spaced spline coefficients to find the beginning, the peak, and the area of the lumps of a ship for classification. This method is better than the Fourier Coefficient method. The study of These methods is presented here.

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## I. INTRODUCTION

It is quite important to classify the ships according to their types. Classification can be done in a number of different ways. The simplest is the visual method that is prone to error. Other methods for classification are the Fourier Coefficient method and the B-spline Coefficient method. In these methods the necessary information for classification can be obtained from the superstructure profile.

The Fourier Coefficient method samples the superstructure profile at every chosen points. The function values at the sampling points are transformed into the spatial components. The logarithm of the magnitude of these components is plotted and compared with the standard plot to recognize the type of the ship.

In the B-spline Coefficient method, the spline coefficients along the X axis and the Y axis are used to reconstruct the superstructure profile. The shape of the curve of the spline coefficients is in some way similar to the shape (the position of the lumps) of the superstructure profile. The ship classification may be achieved by recognizing the beginning, the peak, and the area of the lumps.

## II. PREPROCESSING

Preprocessing is the procedure to obtain the superstructure profile of a ship from the IR image. Then, Fourier coefficient or spline coefficient methods can be used. The details of the preprocessing procedures are as follows.

### A. DATA COLLECTION

The data consists of the IR image of eight different types of ships.

1. DD - Destroyer; "HALL" class.
2. Container; The class is unidentified.
3. Freighter; The class is unidentified.
4. AOR - Replenishment oiler; "WICHITA" class.
5. LST - Tank landing ships; "NEWPORT" class.
6. FF - Frigate; "GARCIA" class.
7. CGN - Guided missile cruiser (Nuclear propulsion); "BAINBRIDGE" class.
8. DDG - Guided missile destroyer; "CHARLES F. ADAMS" class.

These images are taken from an aircraft which is flown at a height of 500 feet with a speed of 400 knots toward the side of the ship. The aspect angles for these images are 90 degrees which may be slightly off in some images. The inaccuracy of the aspect angle arises from the fact that the photos are taken while the aeroplane keeps on moving. All data of the images are stored in a digital magnetic tape with 256 by 64 bytes per image. Thus, the number of bytes required for each image is 16384 and each byte represents the intensity of a pixel. For each record of the image, a label is coded in the last 8 bytes as follows:

1. Byte (16377) = Run number in each flight which passes

the ship.

2. Byte (16378) = Video tape time code when the data is taken; in minutes.
3. Byte (16381) = Video tape time code; in seconds.
4. Byte (16382) = Video tape time code; in thirtieth of a second.
5. Byte (16379) = Range in kilo-feet which is the distance measured from the radar it may have an error 1 to 2 kilo-feet.
6. Byte (16380) = Aspect angle; degrees from the bow of the ship.
7. Byte (16383) = Ship class.
8. Byte (16384) = ID, 1 = for training, 2, 3, 4, 5 = for testing.

The run number and the time code together uniquely define a specific image that represents a single TV frame with no averaging. In addition, the time interval between the end of one image to the beginning of the other is approximately 1.5 seconds. Also, there are inherent random noise in the record which arises from the photo instrument and the process of storing them on to the digital magnetic tape.

## B. SOBEL OPERATOR

The Sobel Operator technique is used to find the edge. To determine the edge, the Sobel Operator uses the difference of gray levels of the pixels in a 3 by 3 matrix as shown in Figure 2.1.

a,b,c,d,e,f,g,h, and i are the values of the gray levels at the position of (x-1,y), (x,y-1), (x+1,y), (x,y), (x+1,y), (x-1,y+1), (x,y+1), and (x+1,y+1) respectively. The Laplacian estimate is defined as

$$\frac{\partial^2 f}{\partial x^2} \simeq f(x,y) - f(x+1,y) - [f(x+1,y) - f(x+2,y)] \quad (2.1)$$

The basis vector for all directions are  $(a-2b+c)$ ,  $(g-2h+i)$ ,  $(a-2d+g)$ , and  $(c-2f+i)$ . Furthermore, each of the basis vector is convolved with the image as follows:

along the x axis

$$d_x = [f(x-1, y-1) + 2f(x, y-1) + f(x+1, y-1)] - [f(x-1, y+1) + 2f(x, y+1) + f(x+1, y+1)] \quad (2.2)$$

along the y axis

$$d_y = [f(x+1, y-1) + 2f(x+1, y) + f(x+1, y+1)] - [f(x-1, y-1) + 2f(x-1, y) + f(x-1, y+1)] \quad (2.3)$$

Since the magnitude of the resulting vector is the absolute value of the convolved results, the edge magnitude  $S(x, y)$  [Ref. 1],

$$S(x, y) = (d_x^2 + d_y^2)^{1/2} \quad (2.4)$$

Note that the Sobel operator does not use the gray level at the position  $(x, y)$ . The advantage of using a Sobel operator over others is that the resulting edge is smoother due to a 3 by 3 matrix approach. If we compare the Sobel operator with the Laplacian operator, it is seen that the Sobel operator using the four basis vector as shown above will provide more accurate reading because of noise reduction in the original image. Hence, The Sobel operator is often used in the preprocessing operation.



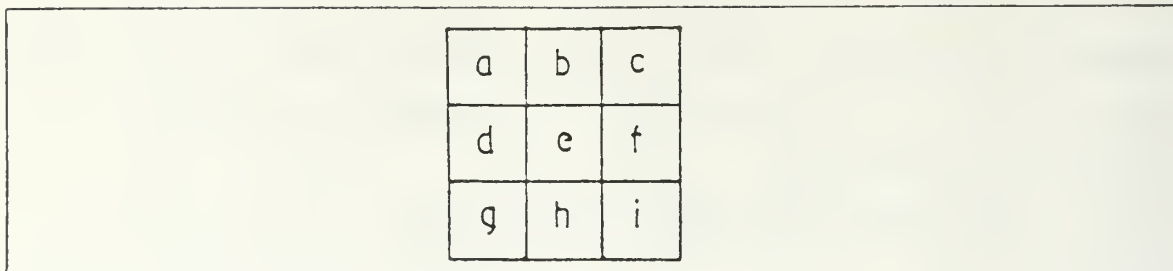


Figure 2.1 Sobel Operator.

### C. THE USE OF A SOBEL OPERATOR

When a Sobel operator is used at the edge of the image frame, the pixel level which lies out of the frame will be set equal to that of the adjacent pixel within the frame. The original image of a guided missile cruiser is shown in Figure 2.2. Since the result of the Sobel operator is numerically greater than 8 bits range, we have to rescale the result back to 8 bits range. This is achieved by determining the maximum and the minimum of the gray level. They are then used to rescale the gray level in the results of the Sobel operator as shown in Figure 2.3. In some instances, the original image is very poor as shown in Figure 2.4. Attempts to determine the edge of this image failed as shown in Figure 2.5.



Figure 2.2 A CGN at a Range of 45000 feet.

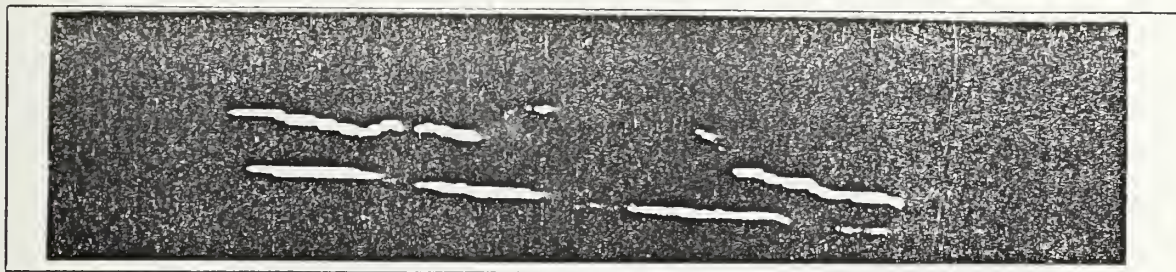


Figure 2.3 Image from a Sobel Operator.

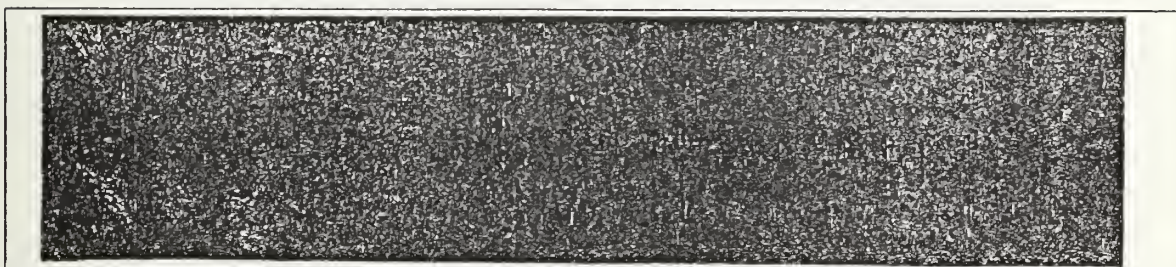


Figure 2.4 Blured Image of a LST at a Range 67000 feet.

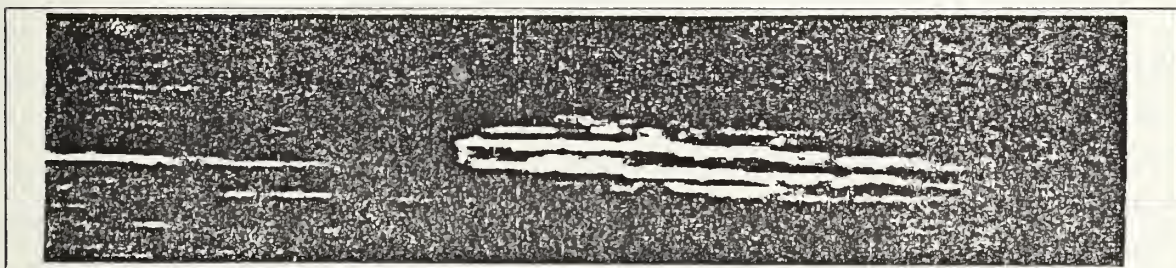


Figure 2.5 Noisy Image from a Sobel Operator.

#### D. EDGE THRESHOLD STRATEGIES

Edge threshold strategies are used to extract the edge profiles from the Sobel results. In this case, we use only the pronounced value of an edge element at  $x$  if  $g(x)$  is greater than certain threshold value [Ref. 2].

$$G(x,y) \begin{cases} = g(x,y) & \text{if } g(x,y) \geq \text{threshold} \\ = 0 & \text{otherwise} \end{cases} \quad (2.5)$$

To increase the contrast of the image to a silhouette form,  $G(x,y)$  is defined as

$$G(x,y) \begin{cases} = 255 & \text{if } g(x,y) \geq \text{threshold} \\ = 0 & \text{otherwise} \end{cases} \quad (2.6)$$

The choice of the threshold value is based upon the histogram of the edge image as shown in Figure 2.6. The estimated critical gray level is chosen so that a majority number of the pixels with value between 0 to 255 will fall below the critical value. Alternatively, histogram equalization may be used to determine the desired threshold level as shown in Figure 2.7. In this case, trial and error method was used in conjunction with the above method to obtain the threshold so that the correct profile is ascertained.



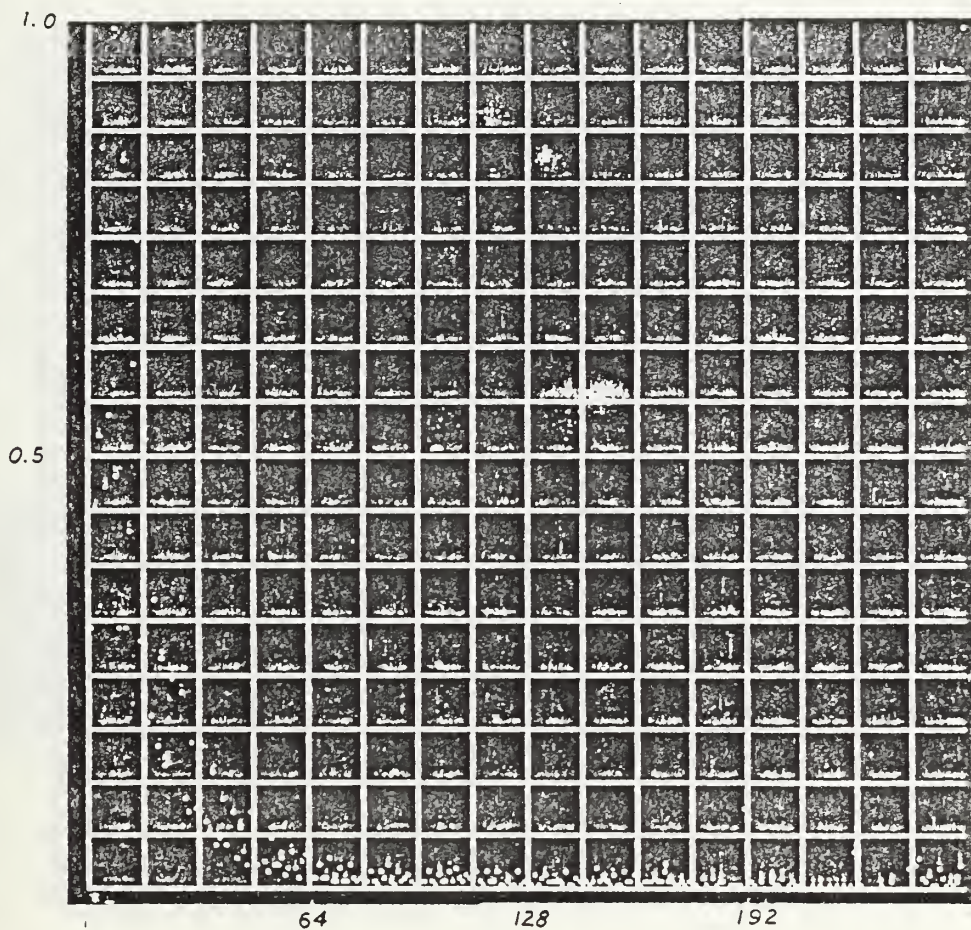


Figure 2.6 Histogram of Figure 2.3.

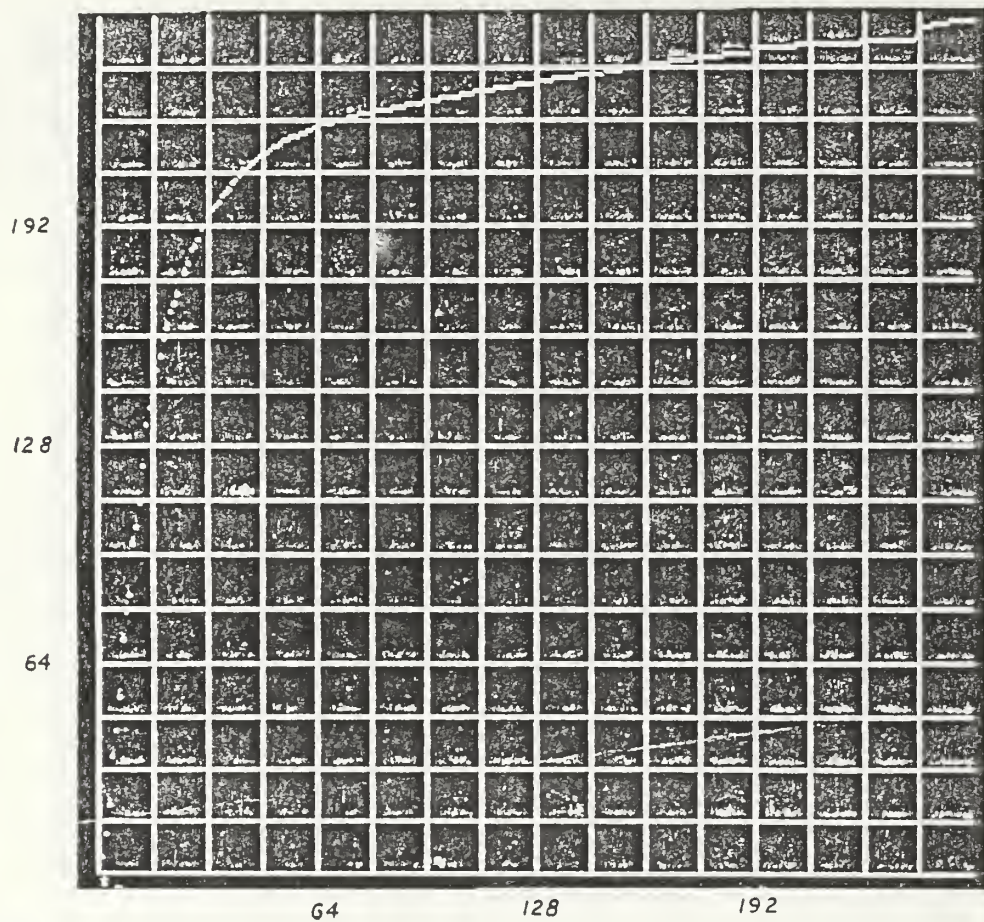


Figure 2.7 Cumulative Distribution of the Histogram  
in Figure 2.6.



#### E. HOW TO EXTRACT THE PROFILE

The edge image of the guided missile cruiser shows little variations in the gray level as shown in Figure 2.3. These variations are caused by the noise in the original image. In this case, the choice of the threshold value is based upon both the histogram and the cumulative distribution of the edge image so that it contains 90 % of the pixels. Therefor, the chosen gray level is 110 and the result is shown in Figure 2.8.

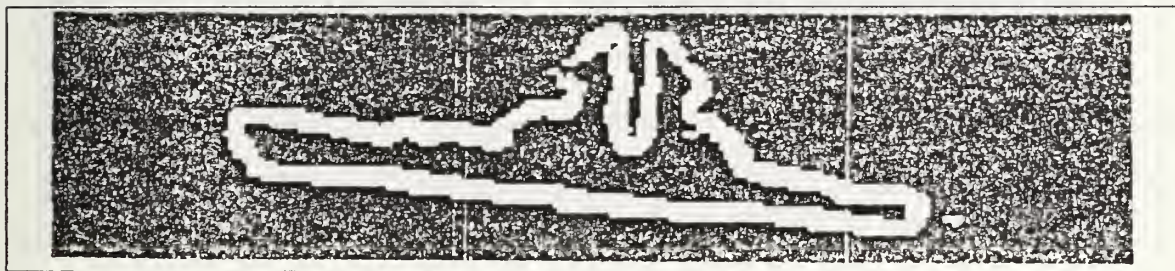


Figure 2.8 Silhouette of a CGN in Figure 2.3.

#### F. HOW TO OBTAIN THE SUPERSTRUCTURE

The original image of the ship is taken from the aeroplane with different displacement from waterline to the superstructure in a rough sea, so that the profile of the ship with respect to the sea surface varies. Thus, we have to eliminate some information in the image of the ship by considering the superstructure only. In considering the overall ship structure, it is obvious that the largest distance is between the bow and stern span. Therefore, the bow and stern points are located first in the program as shown in Appendix A. Then we consider the slope of the bow and stern of the ship and set all the gray values below the

slope line to zero. Hence, it results in a superstructure of the ship as shown in Figure 2.9. However, in some images, there is a lot of noise in the background which cause difficulty in locating the bow and stern. Under these circumstances, the dimensions of a ship are estimated by trial and error method. Then the gray level which lies outside is set to zero and an estimate of the bow and stern slope can be made.

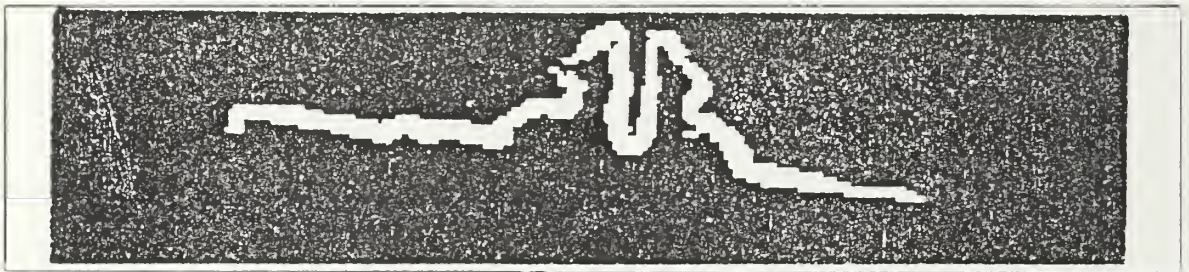


Figure 2.9 Superstructure Profile of Figure 2.8.

#### G. CONTOUR FOLLOWING

The noise in the background of the original image, yields wide edge structure. In considering this factor, the contour following procedure tracking the inner part of the image in Figure 2.9 gives the superstructure profile. The objective of this contour following is to describe the bow and stern points of the ship, the direction, and the position of the edge of the superstructure. The contour tracing is done in the counter-clockwise direction which compares pixel value of 0 or 255 in a 3 by 3 matrix in the following manner.

Starting from the leftmost point in the superstructure image in Figure 2.9 the contour profile tracing is accomplished by examining the neighbors of a 3 by 3 kernel located at the curser position as shown in Figure 2.10. The curser is moved along the profile. All successive positions of the curser constitute the contour profile of the ship. The testing procedure is explained below.

1. Initialize the curser position to the beginning of the thresholded image with the gray level of 255 and the estimate direction of the next position.
2. Check for reaching the end of the profile, if it is at the ending of the profile then stop, if not go to 3.
3. Check for the estimate to see whether it is in the direction of, North, East, South, or West. If the direction it is North then go to 4, if it is East then go to 5, if it is South then go to 6, and if it is West then go to 7.
4. Determine the next position with the gray value of 255 in the counter-clockwise direction as shown in Figure 2.11. Store the position found and move the curser to that position. Update the estimate direction to the one last found; then go to 2.
5. The procedure is the same as that in step 4 except search pattern is shown in Figure 2.12.
6. The procedure is the same as that in step 4 except search pattern is shown in Figure 2.13.
7. The procedure is the same as that in step 4 except search pattern is shown in Figure 2.14.

The flow chart of the procedure is shown in Figure 2.15, and the detail of each procedure are included in Appendix B. The testing procedure have to be performed in such a maner that the resulting contour is a good representation of the superstructure line. The result of the contour

image is shown in Figure 2.16. If the superstructure in Figure 2.9 has wide edge, we can not find the contour image. We have to use the additional step which is called the "Closing" operation.

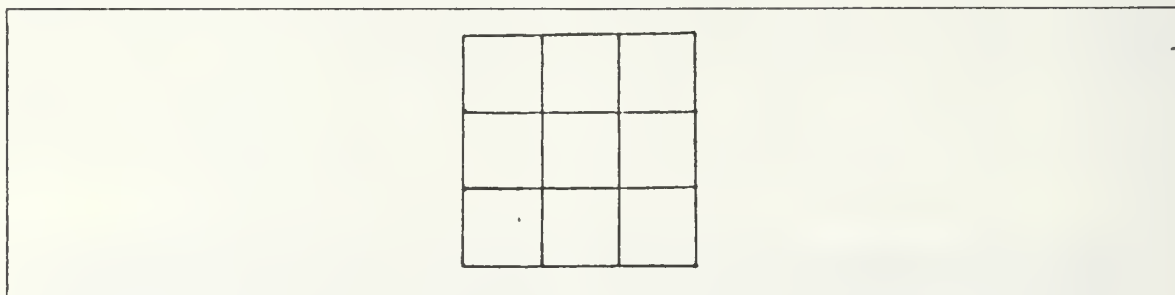


Figure 2.10 The 3 by 3 Kernel.

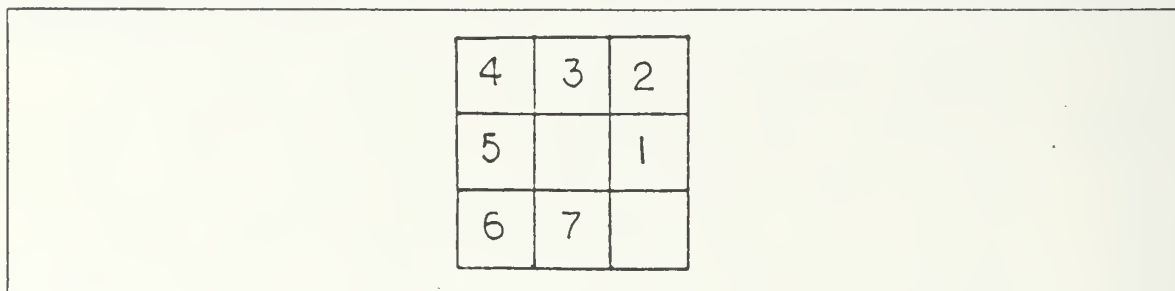


Figure 2.11 The Step Checks in the North Direction.

6	5	4
7		3
	1	2

Figure 2.12 The Step Checks in the East Direction.

		7	6
	1		5
	2	3	4

Figure 2.13 The Step Checks in the South Direction.

	2	1	
	3		7
	4	5	6

Figure 2.14 The Step Checks in the West Direction.



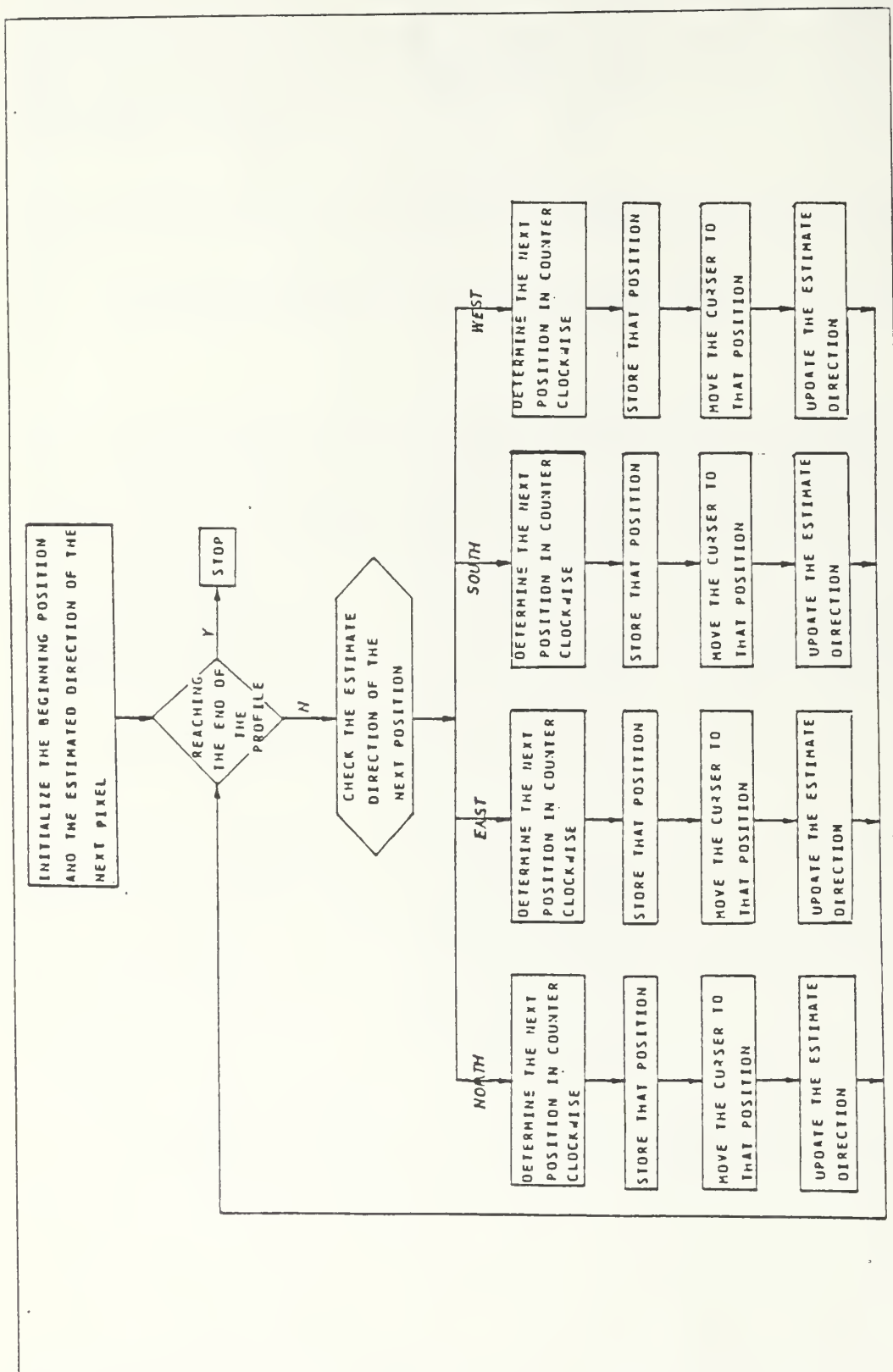


Figure 2.15 Flow Chart of Contour Following.

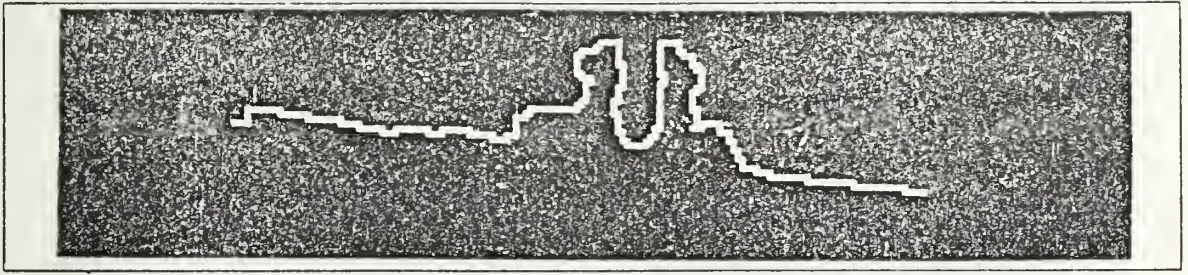


Figure 2.16 Contour Image of a CGN at  
a Range of 45000 feet.

#### H. CLOSING OPERATION

Some results from the superstructure extraction process are discontinuous because the gray level of those areas of the structure is less than the threshold value. If we decrease the threshold value, the details of the superstructure are effected. It is necessary to use the "Closing" operation which consists of the "dilation" process to smooth the superstructure profile used the "Erosion" process. All direction are dilated similarly which causes an smoothing effect on the edges, The superstructure increases in total area. Then, use the "Erosion" process to shrink (subtract) the dilated part in all directions, thus obtain the smoothed superstructure with the appropriate size. The Closing process employs the dilation process which yields an output image from an input image. The Dilation results is shown in Figure 2.17.

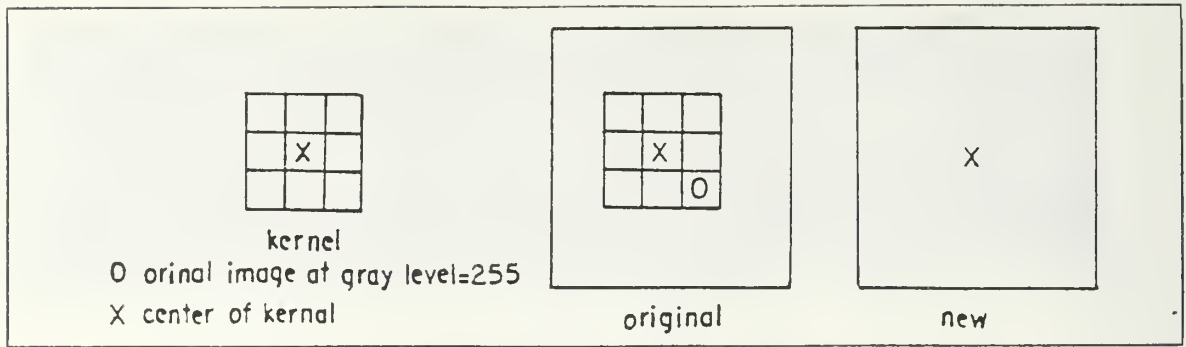


Figure 2.17 The Process of Dilation.

We examine the gray value of each pixel in the original image. Beginning from the pixel at the first row and the first column. The procedures are the following.

1. Considering one pixel in the original image with the kernel (B) centered there. If at least one pixel in the kernel has a value of 255, we let the gray level of the output image at the center of the kernel be 255.
2. We shift the center of the kernel 1 column to the right. Then following the same procedure as in step 1 until the last column is reached.
3. We shift the position of the kernel to the next row and starting from the first column. Then, following the same procedure as in step 1 and step 2 until the last row and the last column is reached.

The result obtained from the dilation process is an image with enlarged structure. The second procedure in the Closing operation is the Erosion process. The Erosion process perform the same procedures as the dilation process except for step 1. If every pixel in the kernel are 255, we let the gray level in the new image at the center of the kernel be 255. Otherwise, it will be 0. The output image obtained will have smooth edge with minor change occurring

in the edge detail as shown in Figure 2.18. In this case, we use an kernel of size 3 by 3. If we increase the size of the kernel, the details of the image are decreased.

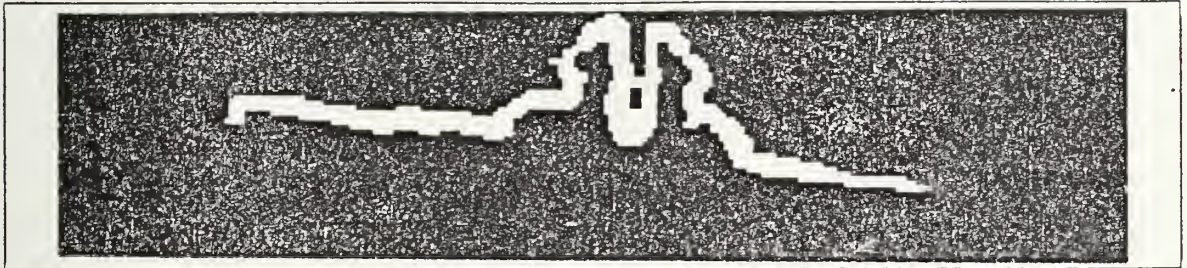


Figure 2.18 The Profile after Dilation and Erosion (Closing Process).

## I. PROFILE ROTATION

Often in the contour image, the first and the last point of the superstructure are not at the same horizontal level. We have to rotate the contour image by setting the two points to the same horizontal level. How to rotate it from the Y, X axis to the Y', X' axis is shown in Figure 2.19.

If the angle value  $\theta$  is positive, it will be

$$\begin{bmatrix} X' \\ Y' \end{bmatrix} = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} \begin{bmatrix} X \\ Y \end{bmatrix}$$

If the angle value  $\theta$  is negative, it will be

$$\begin{bmatrix} X' \\ Y' \end{bmatrix} = \begin{bmatrix} \cos\theta & \sin\theta \\ -\sin\theta & \cos\theta \end{bmatrix} \begin{bmatrix} X \\ Y \end{bmatrix}$$

For some of the contour profile, part of the profile after rotation will be out of the image frame. Then we have to shift this contour down by 20 pixels position. The rotated profile is shown in Figure 2.20.



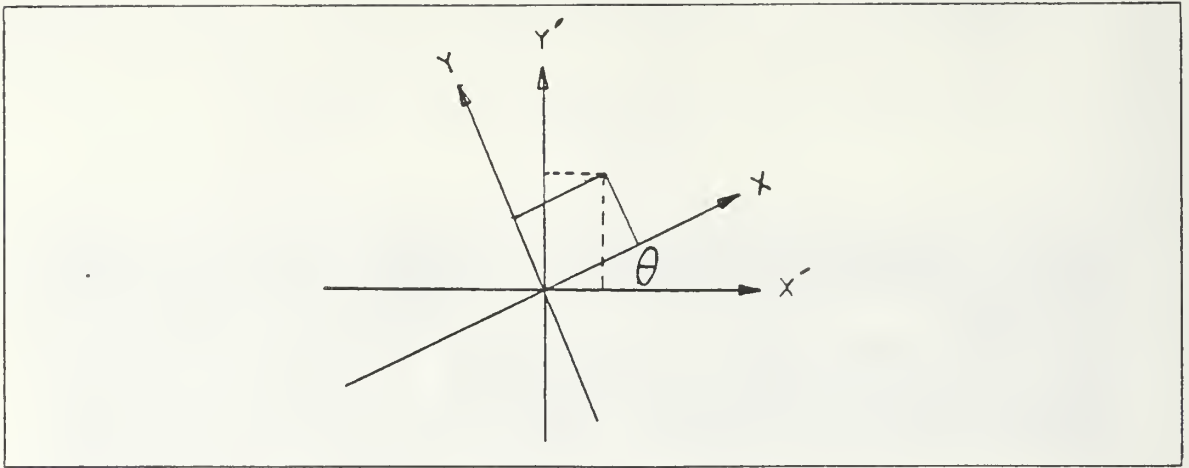


Figure 2.19 Rotation Process.

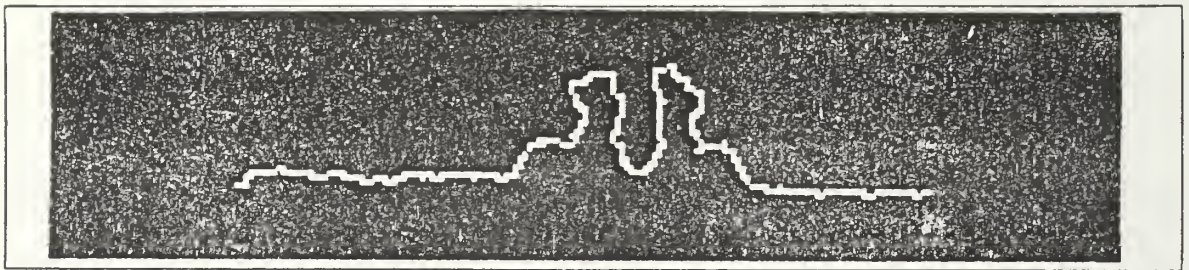


Figure 2.20 Rotated Profile of a CGN at  
a Range of 45000 feet.

### III. FOURIER COEFFICIENT METHOD

We have obtained the ship profile in the previous chapter. To extract features out of this profile for classification purposes, we will use the Fourier Transform method.

The Fourier transform of the ship profile showed that the transform coefficients depend upon the ship's dimensions, its superstructure, and the distance between the camera and the ship. If the profile  $f(x)$  is a discrete function with 128 sample points. The discrete Fourier transform can be written as

$$F(u) = \frac{1}{N} \sum_{x=0}^{M-1} f(x) \exp(-j2\pi u x / N) \quad (3.1)$$

For  $u, x = 0, 1, 2, \dots, N-1$ .  $N$  is the total number of samples.

If the direct calculation of the discrete Fourier transform is chosen, the number of complex multiplication and addition will be equal to  $N$ ; i.e. to obtain  $F(0)$  would require complex multiplication and addition  $N$  times. In order to reduce the computation, we use the fast Fourier transform algorithm. Thus, equation 3.1 [Ref. 2] can be separated into  $F_{\text{even}}(u)$  and  $F_{\text{odd}}(u)$

$$F_{\text{even}}(u) = \frac{1}{M} \sum_{x=0}^{M-1} f(2x) W_M^{u x} \quad (3.2)$$

$$F_{\text{odd}}(u) = \frac{1}{M} \sum_{x=0}^{M-1} f(2x+1) W_M^{u x} \quad (3.3)$$



$$W = \exp(-j2\pi/M) \quad (3.4)$$

$$M = \frac{N}{2} \quad (3.5)$$

$$F(u) = \frac{1}{2} [F_{even}(u) + F_{odd}(u) W_{2M}^u] \quad (3.6)$$

$$F(u+M) = \frac{1}{2} [F_{even}(u) - F_{odd}(u) W_{2M}^u] \quad (3.7)$$

Using this method, we have reduced the total number of complex multiplication and addition to  $N \log N$ . In this case, we have  $N = 128$ , thus, the total number of complex multiplication and addition will be 896.

First, we divide the rotated profile image into 128 divisions. Since the distance of each division is equal to the amount of the pixel between the bow and the stern of the ship divided by 128, we use the distance perpendicular from the horizontal line between bow and stern to the highest point of the superstructure as the sampled values. The result of the fast Fourier transform is a complex number. Then we use

$$M(u) = \log[1 + G(u)] \quad (3.8)$$

$G(u)$  is the magnitude of  $F(u)$ . A value of 1 is added to the magnitude to avoid negative logarithm result, the results obtained are as shown in Table I and

1. Figure 3.1 - Destroyer
2. Figure 3.2 - Container
3. Figure 3.3 - Freighter
4. Figure 3.4 - Replenishment oiler
5. Figure 3.5 - Tank landing ship

6. Figure 3.6 - Frigate
7. Figure 3.7 - Guided missile cruiser
8. Figure 3.8 - Guided missile destroyer

On the results minor difference of the shape of the Fourier coefficients can be noticed visually. But it is difficult to implement a program to detect these minor difference in shape. Therefore, a Second method was used to handle this problem.

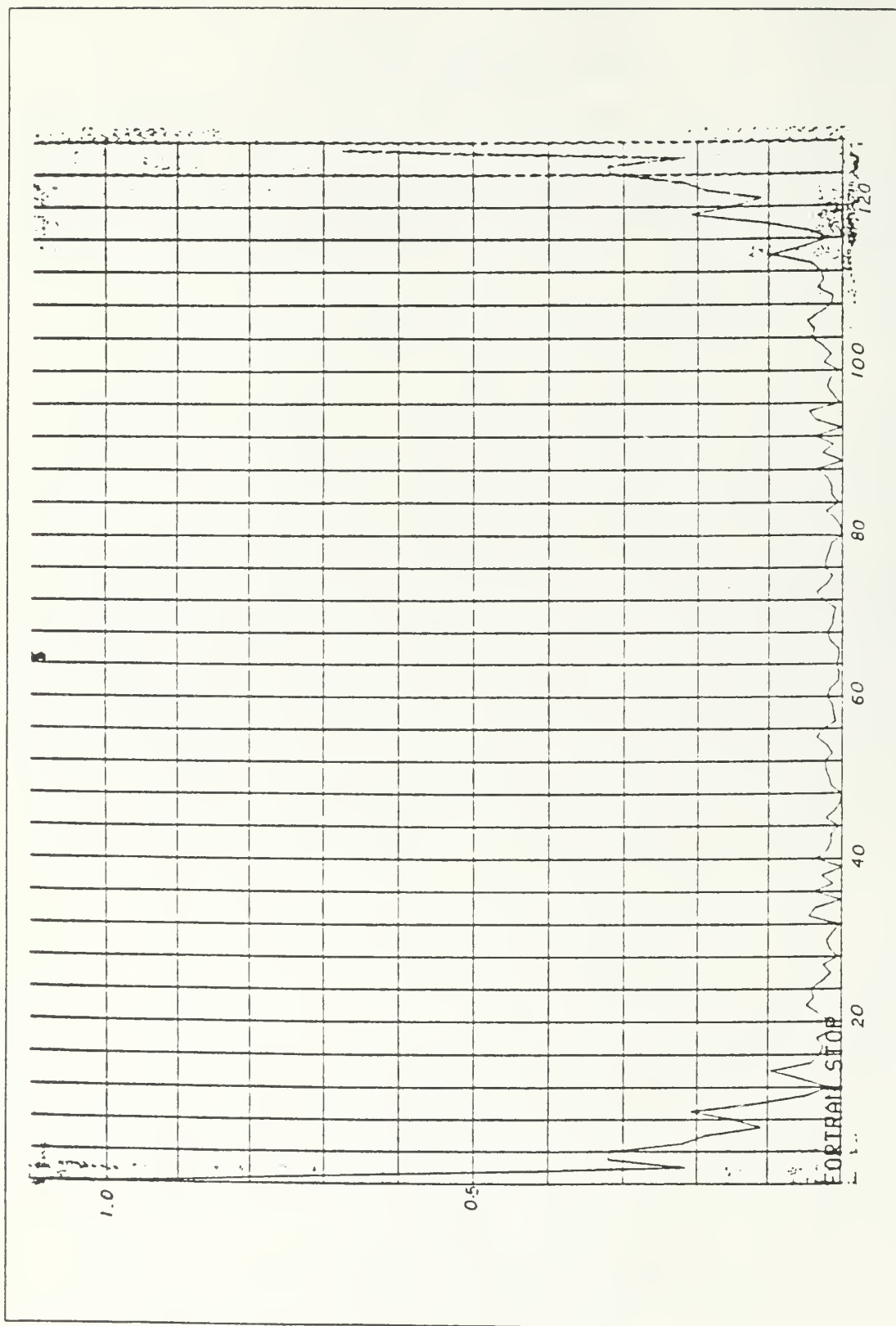


Figure 3.1 Logarithmic Magnitude of the Fourier Transform of a DD.

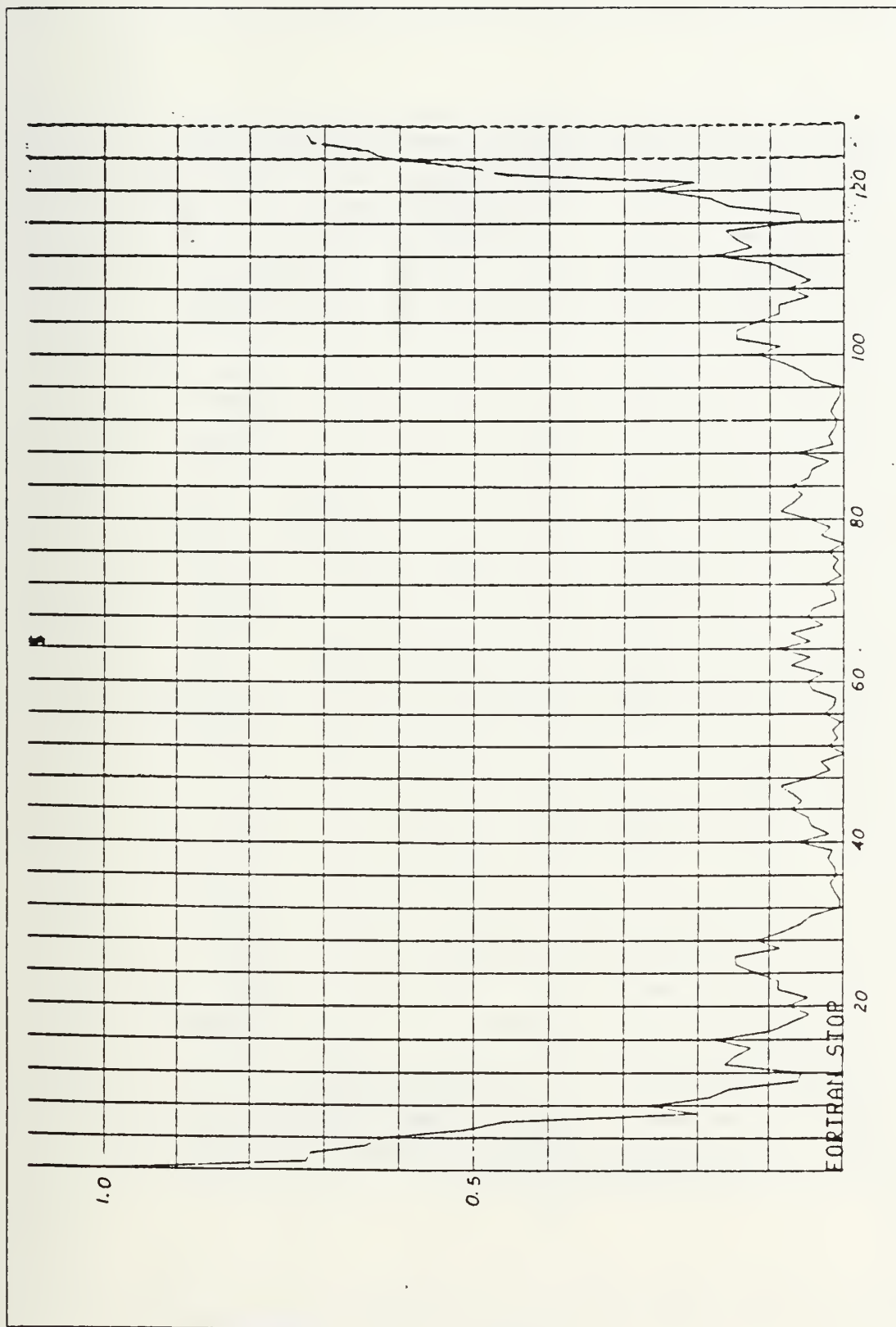


Figure 3.2 Logarithmic Magnitude of the Fourier Transform of a Container.

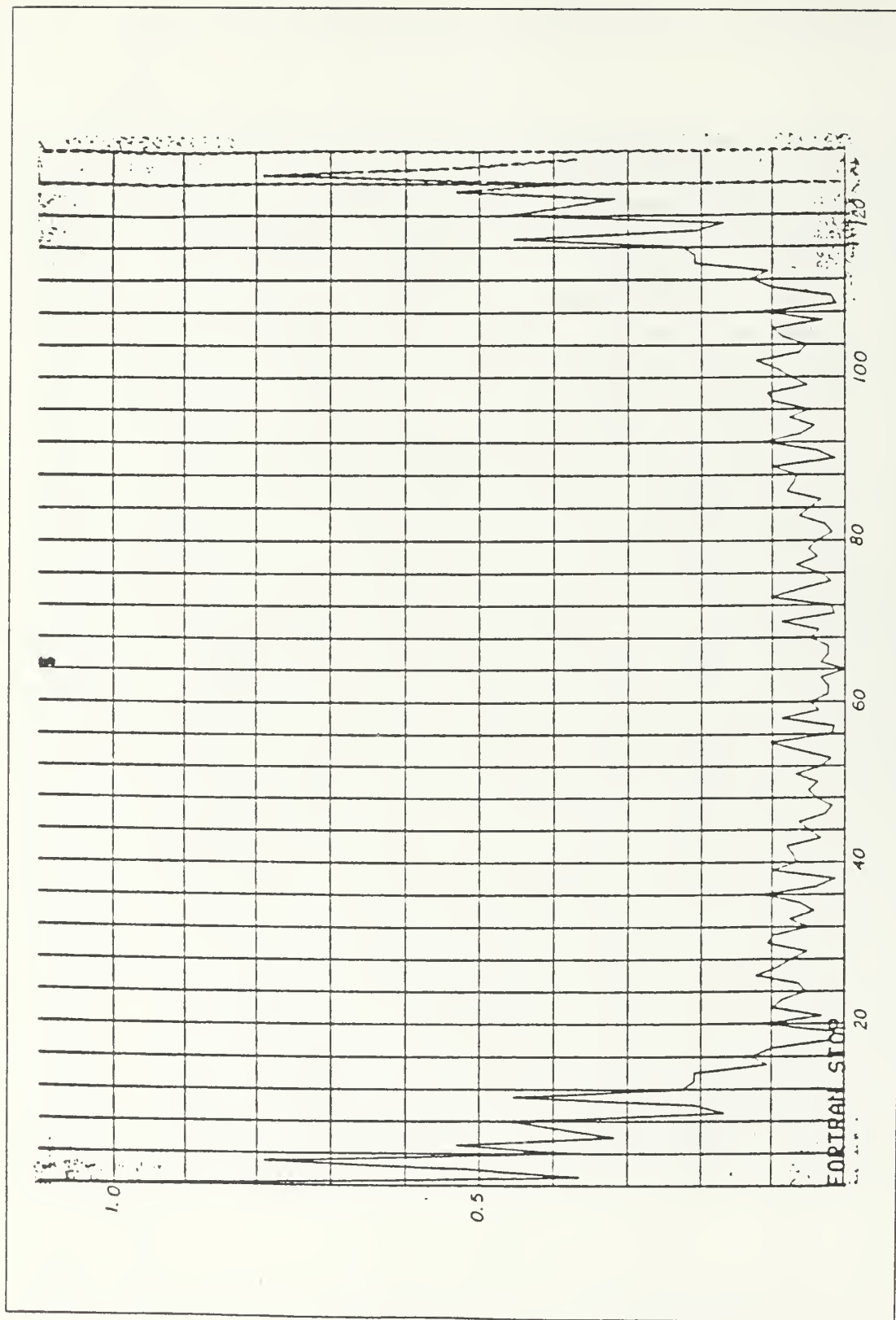


Figure 3.3 Logarithmic Magnitude of the Fourier Transform of a Freighter.

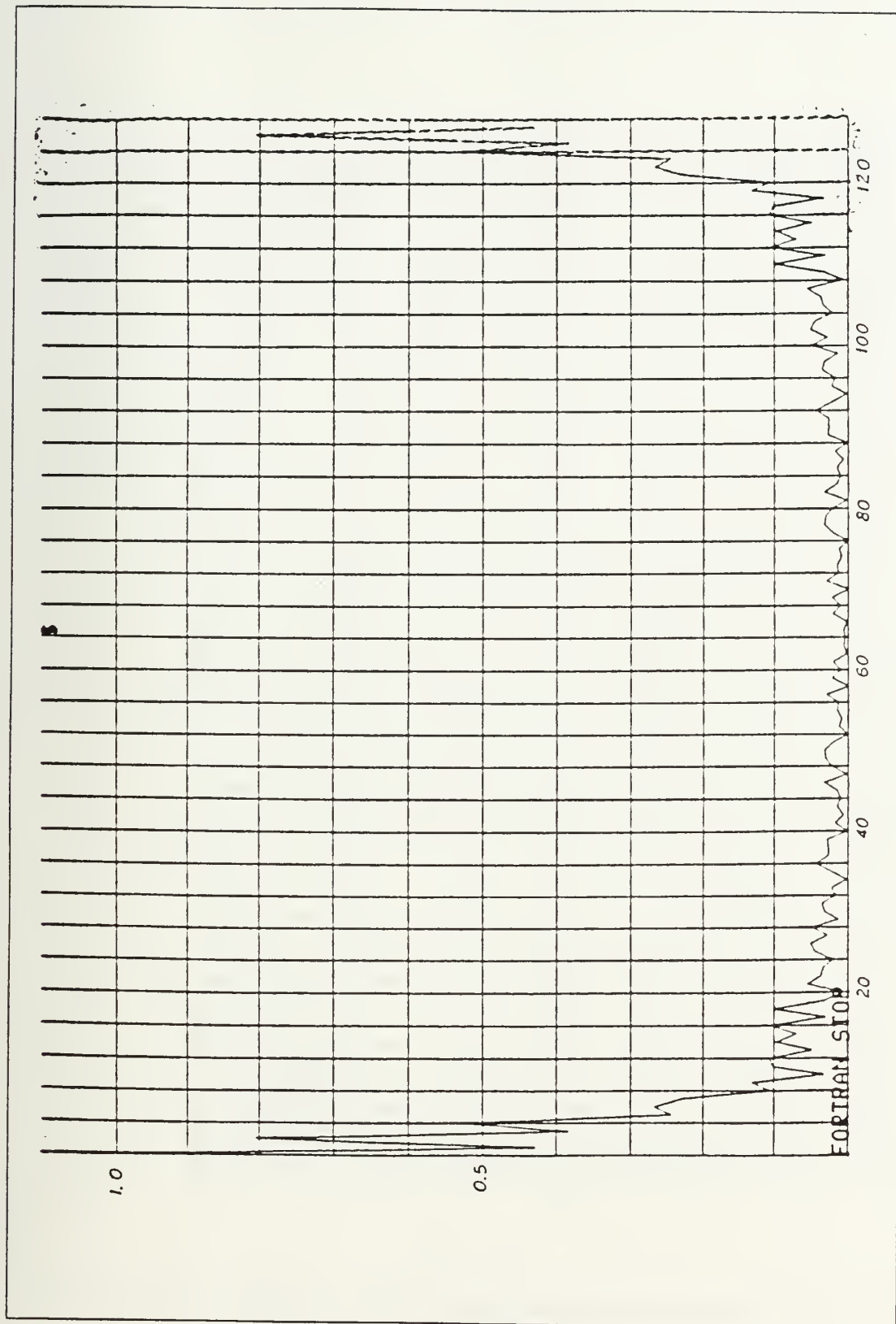


Figure 3.4 Logarithmic Magnitude of the Fourier Transform of a AOR.



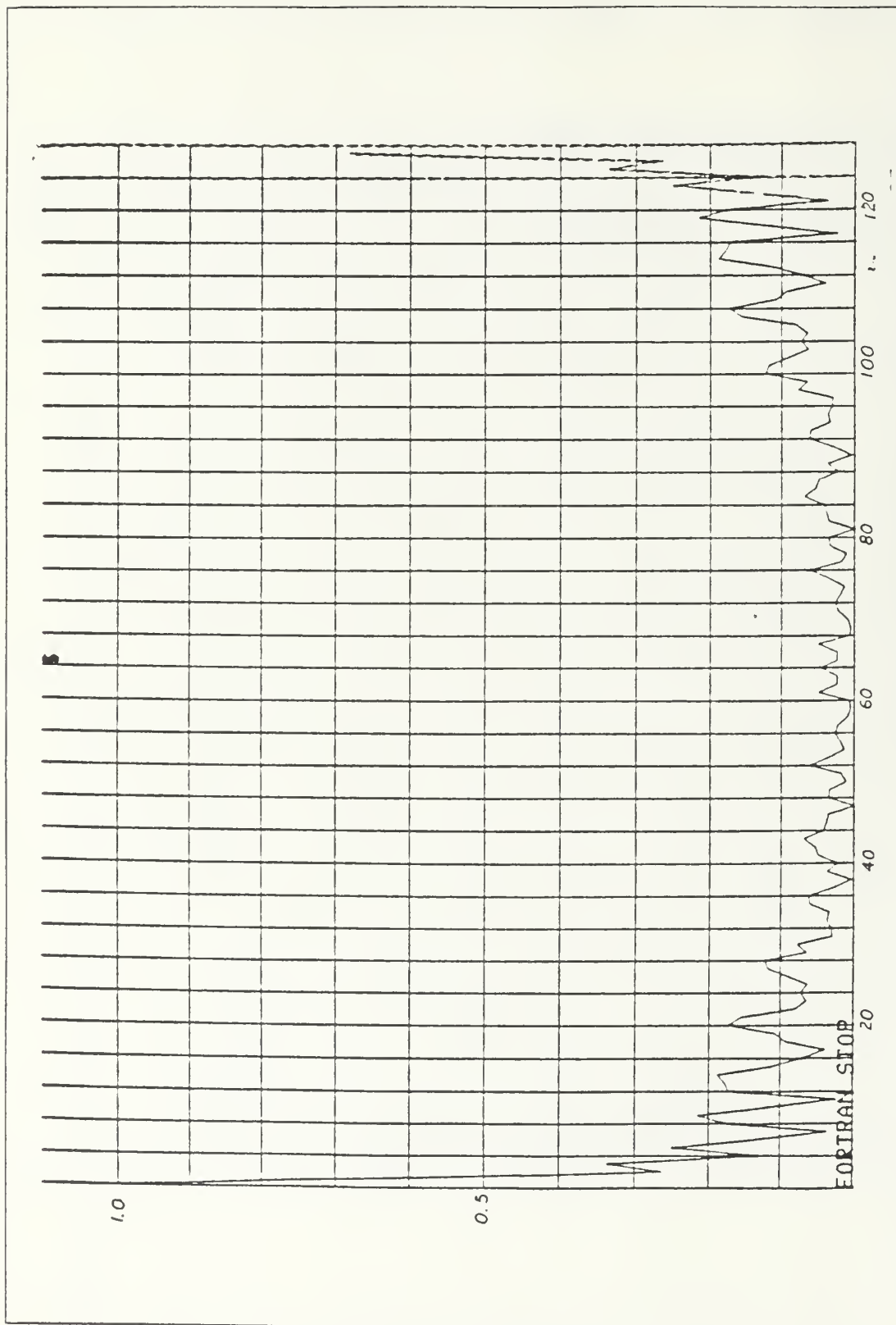


Figure 3.5 Logarithmic Magnitude of the Fourier Transform of a LST.

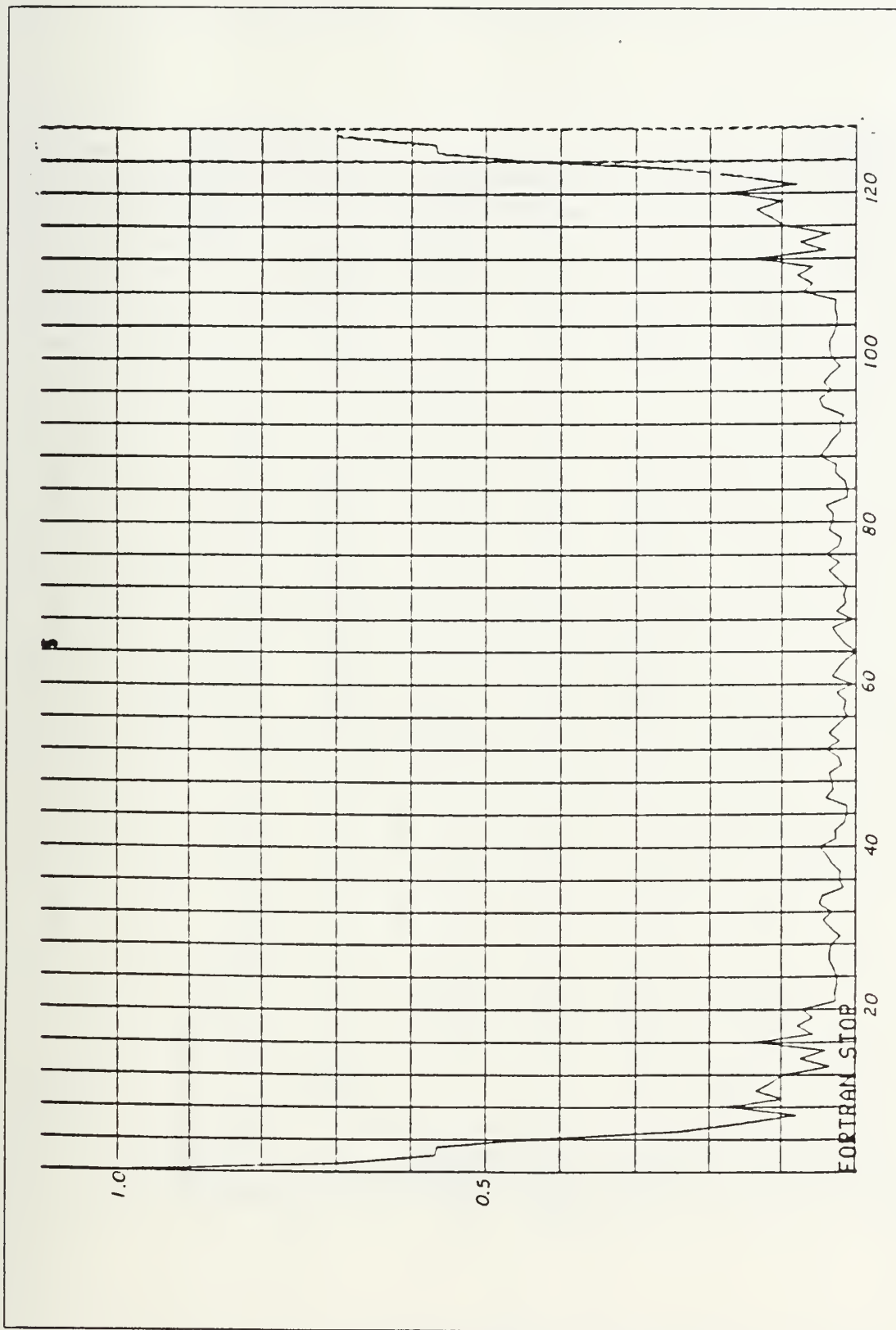


Figure 3.6 Logarithmic Magnitude of the Fourier Transform of a FF.

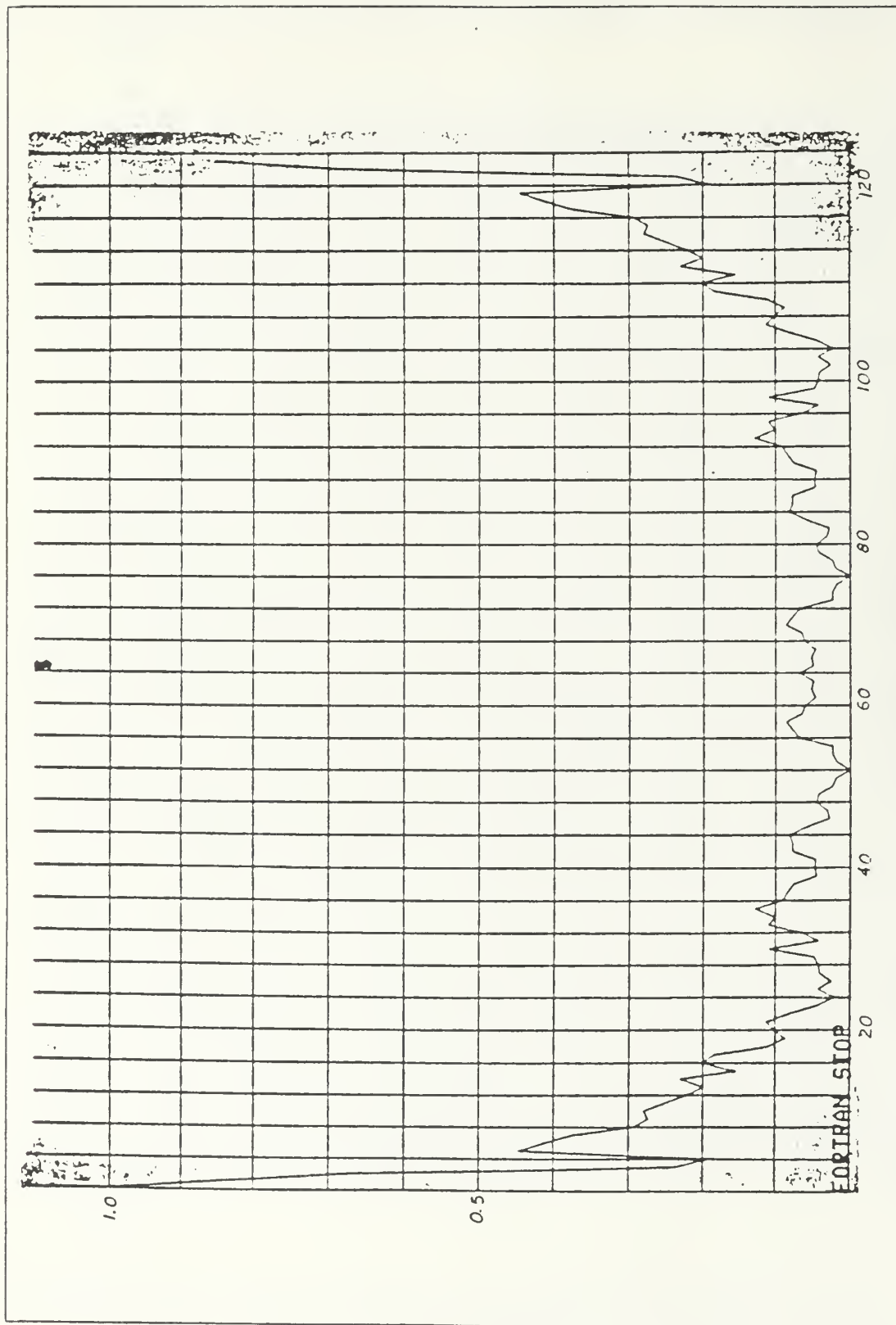


Figure 3.7 Logarithmic Magnitude of the Fourier Transform of a CGN.

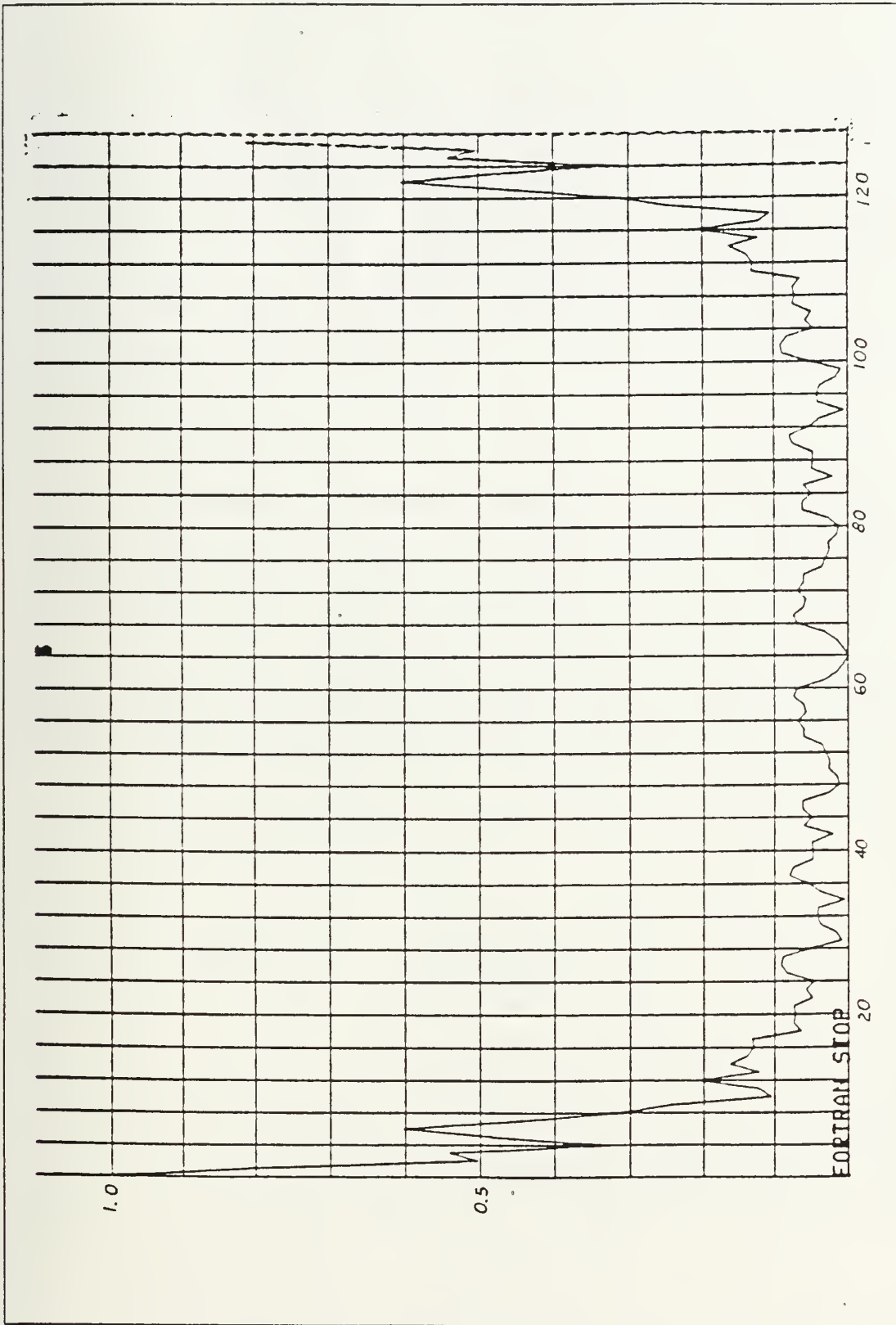


Figure 3.8 Logarithmic Magnitude of the Fourier Transform of a DDG.

TABLE I  
Logarithmic Magnitude of Ships

F(N)	DD	Cont.	Frig.	AOR	LST	FF	CGN	DDG
(1E-1)	(1E-1)	(1E-1)	(1E-1)	(1E-1)	(1E-1)	(1E-1)	(1E-1)	(1E-1)
DC	1.E1	1.E1	1.E1	1.E1	1.E1	1.E1	1.E1	1.E1
F(1)	6.74	7.23	3.66	4.28	6.81	6.97	7.78	8.10
F(2)	2.17	7.18	5.17	8.04	2.64	5.67	6.37	5.04
F(3)	3.21	6.45	7.88	3.84	3.35	5.64	2.38	5.40
F(4)	3.17	6.18	3.76	5.28	1.28	4.43	7.6E-1	3.26
F(5)	2.20	5.09	5.31	2.45	2.49	2.40	3.53	4.81
F(6)	1.88	4.57	3.17	2.66	1.36	1.52	3.48	6.02
F(7)	1.10	2.03	3.90	2.28	3.6E-1	8.1E-1	3.44	4.50
F(8)	1.50	2.65	4.58	1.07	1.74	1.76	3.35	3.14
F(9)	2.07	1.83	1.67	1.32	2.14	9.9E-1	3.40	2.46
F(10)	1.13	1.57	2.07	3.39E1	1.19	1.13	3.24	1.05
F(11)	4.6E-1	6.0E-1	4.53	1.03	2.46	1.14	2.62	1.23
F(12)	1.6E-1	5.6E-1	2.23	1.07	1.73	9.8E-1	1.80	2.09
F(13)	5.4E-1	1.62	2.09	5.0E-1	1.78	3.5E-1	3.5E-1	1.23
F(14)	9.6E-1	1.48	2.09	9.8E-1	1.87	7.4E-1	6.7E-1	1.62
F(15)	4.1E-1	1.25	1.07	7.0E-1	1.14	4.1E-1	1.19	1.38
F(16)	2.9E-1	1.77	1.30	1.05	7.1E-1	1.36	1.25	1.30

DD = Destroyer at range 77000 feet.

Cont = Container at range 28000 feet.

Frig= Freighter at range 40000 feet.

AOR = Replenishment oiler at range 78000 feet.

LST = Tank landing ship at range 51000 feet.

FF = Frigate at range 49000 feet.

CGN = Guided missile cruiser at range 45000 feet.

DDG = Guided missile destroyer at range 41000 feet



## A. VARIATION OF SHIP SUPERSTRUCTURE WITH RANGE

One of the practical problem of using the ship profile is that it is sensitive to range variations. Close ship profile has more details than the far away ship profile. The dependency of the geometric size of the profile with the range is discussed in this section.

Assume that the object is centered on the camera axis. The field of view of the camera, the number of the pixel in the image, the size of the image, and the size of the object are known. Our problem is to determine the distance between the camera and the object.

### 1. Background

The camera system is similar to a human eyes system. The light reflected from the object goes into the eyes. The image of the object falls on the retina, the signal is sent to the brain in some electrical from, and the brain changes it to a from that human can perceive. But, in the camera system film or sensor are used to pick up the image. The function of a camera is shown in Figure 3.9

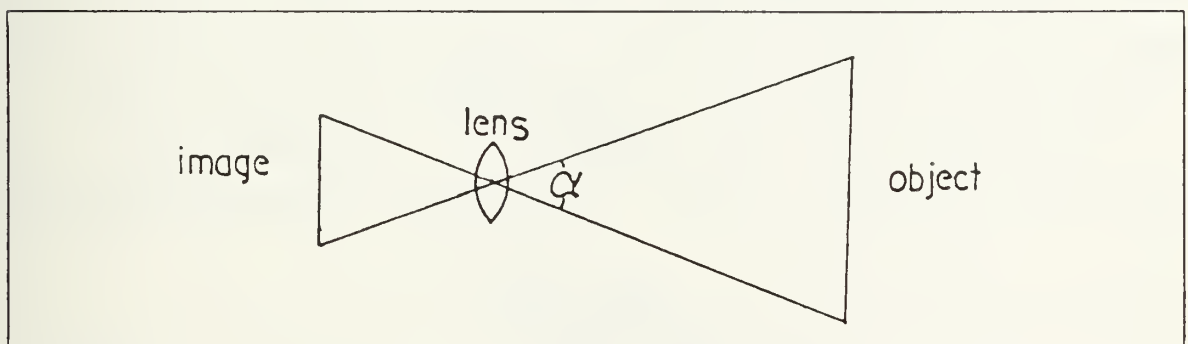


Figure 3.9 The System of the Camera.

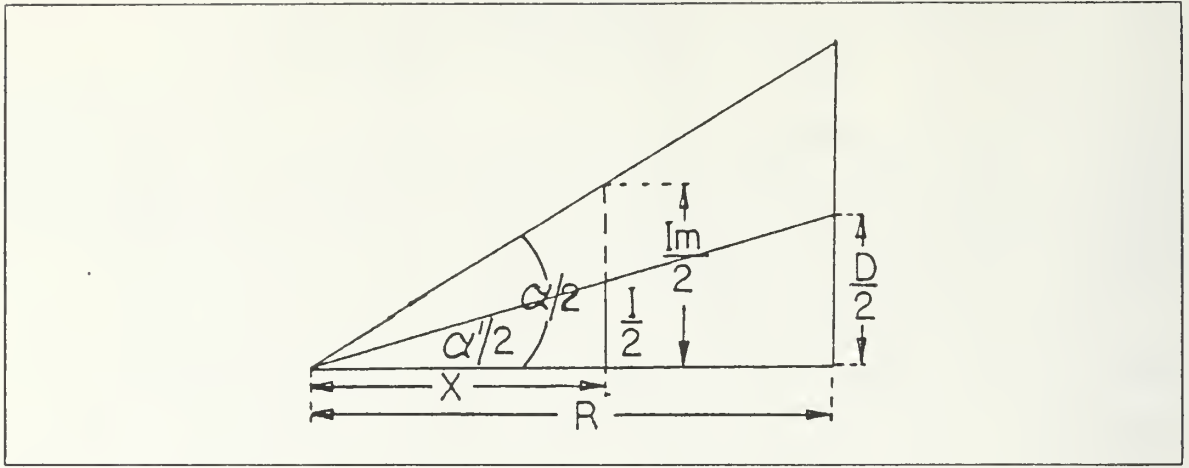


Figure 3.10 Range Calculation.

The distance between the lens and the image plane can be adjusted in order to have a clear image on the film. The camera has a field of view (FOV) angle as shown in Figure 3.9. The object has to be in the field of view of the camera. The determination of the distance is shown in Figure 3.10

For simplicity of calculation the inside of the camera is flipped to the same side as the object as shown in Figure 3.10. when the angle of the FOV is  $\alpha$ ,  $I/2$  is the half of the full image size,  $D/2$  is the half of the dimension of the object,  $X$  is the distance from the lens to the image, and  $R$  is the distance from the lens to the object.

Assume that the distance  $I/2$ , distance  $Im/2$ , and angle  $\alpha/2$  are known. Then, the distance  $R$  can be determined by

$$X = \frac{Im}{2} \tan \frac{\alpha}{2} \quad (3.9)$$

$$\tan \frac{\alpha'}{2} = \frac{I}{2X} \quad (3.10)$$

$$R = \frac{D}{2} \frac{1}{\tan \frac{\alpha}{2}} \quad (3.11)$$

Assume that the angle resolution of the pixel of the field of view of the camera is equal to  $0.2E-3$  radian per pixel. The number of pixel of the frame is equal to 256. The size of an image is  $I$  pixels. The dimension of the object  $D$  in feet is known. The field of view in angle is

$$\alpha = (0.2E-3)256 \quad (3.12)$$

$$X = \frac{256}{2} \frac{d}{\tan \frac{\alpha}{2}} \quad (3.13)$$

where  $d$  is in unit of pixel.

$$\tan \frac{\alpha'}{2} = \frac{I}{2} \frac{d}{X} \quad (3.14)$$

$$\tan \frac{\alpha'}{2} = \frac{I}{256} \tan \frac{\alpha}{2} \quad (3.15)$$

$$R = \frac{D}{2} \frac{1}{\tan \frac{\alpha'}{2}} \quad (3.16)$$

$$R = \frac{128D}{I \tan 0.0256} \quad (3.17)$$

When the length  $D$  of the object is known, from equation 3.17 we can estimate the distance from lens to the object and is shown in Table II

TABLE II  
Range Estimation

CLASS	I(pixel)	D (ft)	R' (kft)	R(kft)	(R-R')100
	MEASURE	KNOWN	CALCULATE	RADAR DIS	R
DD1	96	418	21.766	77	71.71
DD2	80	418	26.119	85	69.27
AOR1	107	659	30.787	78	60.53
AOR2	96	659	34.315	85	59.63
LST1	176	522.3	14.834	51	70.91
LST2	134	522.3	19.484	57	65.82
CGN1	147	565	19.213	45	57.31
CGN2	119	565	23.734	55	56.85
DDG1	126	437	17.337	47	63.11
DDG2	90	437	24.247	64	62.08

DD1,DD2 = Destroyer at range 79000 and 83000 feet.

AOR1,AOR2 = Replenishment oiler at range 78000 and 88000 feet.

LST1,LST2 = Tank landing ship at range 51000 and 62000 feet.

CGN1,CGN2 = Guided missile cruiser at range 45000 and 64000 feet.

DDG1,DDG2 = Guided missile destroyer at range 41000 and 64000 feet.

The error distance between the estimated distance and calculated distance in percentage is  $((R - R')/R) 100$ .

## 2. Remark

Calculated distance error in R may come either from the pixel measurement in an image or from the angular resolution estimation. The distance that is stored in the image label has an error of about 1 to 2 kilo-feet. If the angle

of the field of view is accurate, we can estimate the number of pixel in an object image. Then we can classify the object by comparing the number of pixel of the original image with that of the test image. Some known system parameters can help to determine the range of the target ship. The problem is that existing errors in the system parameter causes in the larger errors in the estimated range. Consequently, they are not very useful in classifying the ships.



#### IV. B-SPLINE COEFFICIENT METHOD

Using B-spline coefficient is another method to describe a ship profile. This method uses the B-spline coefficients to determine the beginning, peak, and area of lumps which contain significant information about the type of the ships. The comparisons of the knot position (in parametric value) from the midships to the peak or beginning of the lump, can be very helpful. Different ships will have different lump positions and sizes.

##### A. BACKGROUND

A spline function is a piecewise polynomial used to interpolate points. This kind of curve is smooth and the discontinuities in its  $k$ -th derivative is as small as possible. In this case, Cubic spline was used, where the first and second derivatives for any set of interpolating points are continuous, while the third derivative may be discontinuous. The reason for using Cubic spline is to keep the third derivative discontinuity as small as possible and the curve as smooth as possible. The B-spline calculation procedure is also very stable. In our case the order of spline used is 3, while the 1-st and 2-nd derivative are continuous. The choice of 4-th order spline function is due to the fact that it is generally sufficient for most ship profile curves. Discontinuity, in a sense, may be stated as the jumps of the third derivative, which is the means to control smoothness of the connecting pieces.

## B. B-SPLINE APPROXIMATION WITH FREE KNOT

As mention earlier, the B-spline function is used in the spline approximation with free knot. The free knots is sometimes called uneven or irregular knots; that is no fixed number of knots are used. Furthermore, the spline position need not be on the original curve so as to minimize the number of knot position while preserving most of the details of the original curve.

First, minimize the value of the lack of the smoothness  $\eta(\bar{c})$  defined as [Ref. 3]

$$\eta(\bar{c}) = \sum_{j=1}^n \left( \sum_{i=-k}^n a_{ij} c_i \right)^2 \quad (4.1)$$

where  $C_i$  is a coefficient of spline at the knot position.  $a_{ij}$  is defined as follows

$$a_{ij} = M_{i,k+1}^{(k)}(t_j+0) - M_{i,k+1}^{(k)}(t_j-0) \quad (4.2)$$

where  $M_{i,k+1}$  is the normalize B-spline function and defined as

$$M_{i,k+1}(x) = (t_{i+k+1} - t_i) \Delta_t^{k+1}(t_i, \dots, t_{i+k+1}) g_k(t; x) \quad (4.3)$$

and  $G_k(t; x)$ ,  $t$  are defined as follows

$$g_k(t; x) \begin{cases} = (t-x)_+^k = (t-x)^k & \text{if } t \geq x \\ = 0 & \text{if } t < x \end{cases} \quad (4.4)$$

$\Delta_l^l(Z_i, Z_{i+1}, \dots, Z_{i+l})f(t)$  stands for the  $l$ -th divided difference of the function  $f(t)$  on the point  $Z_i, \dots, Z_{i+l}$  where  $t$  is the position values of knots in term of  $Z$  parameter.  $Z$  parameter is defined as follows

$$Z(I) = Z(I-1) + \left[ (X(I) - X(I-1))^2 + (Y(I) - Y(I-1))^2 \right]^{1/2} \quad (4.5)$$

$$Z(0) = 0 \quad (4.6)$$

where  $I$  is the number of the sampling point and  $I=1, 2, 3, \dots, m$ .

Second, the smoothing is subjected to a constraint

$$\delta(c) \leq s \quad (4.7)$$

where  $S$  is the smoothing factor,  $\delta(\bar{c})$  is the weighted sum of the square residuals defined as

$$\delta(c) = \sum_{j=1}^m w_j \left[ Y_j - \sum_{i=-k}^n c_i M_{i,k+1}(X_j) \right]^2 \quad (4.8)$$

$X_j, Y_j$  are the values of  $X$  and  $Y$  at  $Z$  parameter of the sampling points;  $w_j$  is a weighting factor for all sampling points [Ref. 3] defined as

$$w_j = (\delta Y_j)^{-2} \quad (4.9)$$

where  $w_j$  is an estimate of the standard deviation of  $Y_j$ . Then, the value of  $S$  is in the range  $m + \sqrt{2m}$ . If nothing is known about the statistical standard deviation of  $Y_j$ , each

W can be equal to one and the S can be determined by the trial and error method.

#### 1. Interpolation and Approximation Using B-spline Function

There is a difference between the interpolation and the approximation. In interpolation, the number of knot positions required are equal to the number of sampling points and the value of S in eq(4.7) is small. In this case, the computation time required will increase tremendously. Whereas, in approximation, it is not of great concern that the approximated value at a position be the same value of the sampling point, and most of the information still remain in the original curve. Thus, decreasing the value of S will result in the large number of the knot positions and the final curve obtained will be similar to the original curve.

The justification for using approximation rather than interpolation approach is that though the resulting curve may not be the best fitting curve, it is smooth and close enough to the original. The approximation spline function has less number of knots than the number of samples, which reduces the total processing time. In approximation approach, when the appropriate choice of S value is made, it will, in turn, generate sufficient number of knot positions required to provide a close approximation to the original curve. The ratio of sampling point to spline coefficient is 10 to 1 as shown in Figure 4.1. In Figure 4.1, a guided missile cruiser ship at a distance of 45000 feet, with 290 sampling points and the associated B-spline knots are shown. The original samples are plotted in solid curve. The knot position are show by small circles in Figure 4.1. After B-spline approximation, the number of the knots are reduced to 33. Hence, this technique is essentially a kind of data compression technique.

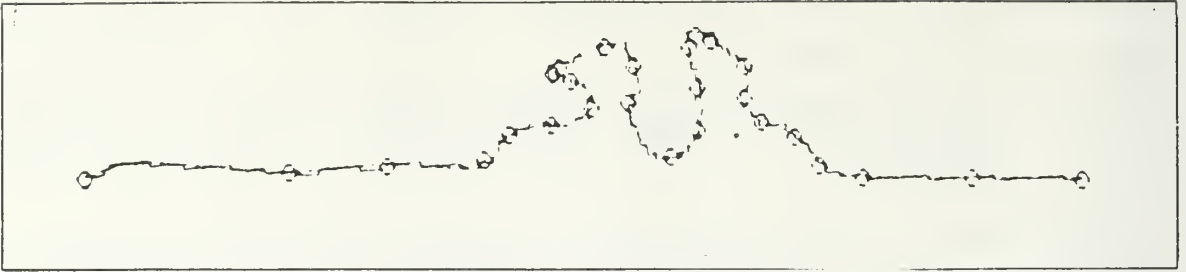


Figure 4.1 Plot of the Original Data and the Approximate with Free Knot.

#### C. TO DETERMINE THE KNOT POSITION AND THE B-SPLINE COEFFICIENTS

The approximate smoothing factor in eq(4.7) is to be calculated before using the subroutine "PARAM" [Appendix C]. In the subroutine, there is a check on S factor every time it is run. If the S input values do not satisfy or meet the criteria, the program will return with a error code IER.

There is a parameter NEST which is set to a constant value. This value determine the dimension of the knot positions which relates to the array T(NEST), Cx(1..NEST) and Cy(1..NEST). The NEST is an overestimate of the dimension of the arrays set by the user. The limitation on the value of NEST for subroutine "PARAM" are as follows [Appendix C]

1.  $2k+2 \leq \text{NEST} \leq M+k+1$

2. Typically, value of NEST  $M/2$

where M is the number of the total sampling points and  $k=3$  is the highest order in the B-spline function.

The subroutine "PARAM" produces several outputs as, N the total number of knot positions, T(N) the array of the value of knot positions in Z parameter, Cx(N) and Cy(N), B-spline coefficient of X functions and Y functions for knot positions defined in array T(N).



## 1. Limitation on B-spline Coefficient Determination

If the value of NEST is too small, the user will receive error code IER and the number of knot positions will appear to be dense in the first part of the curve. While sparse in the other part. The example of an incorrect selection of NEST is shown in Figure 4.2.

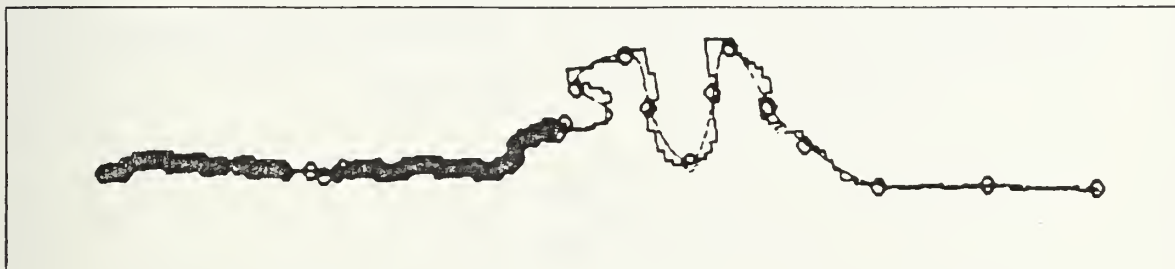


Figure 4.2 Knot Selection if the NEST Parameter is too Small.

Another problem which causes difficulty in programming is that the main program is in PASCAL while the subroutine is in FORTRAN. As for the PARAM program in FORTRAN, the array index value starts from 1, while , for the user PASCAL program it starts from 0. Therefore, in linking the main program to the subroutine, one has to keep in mind of the difference. In addition, the FORTRAN programming logic structure is so complicated that, when an error occurs, it is very difficult to debug and locate the error.

The values of  $C_x$  and  $C_y$  depend upon uneven knot positions, and they contribute controlling effect to the reconstructed curve which would be both smooth and close to the original curve. In running the B-spline approximation program for the first time, the values of  $C_x$  and  $C_y$  of the last 4 knots at the right end are zeros. However, in running it again, increase the value of  $S$  did not yield zero values of  $C_x$  and  $C_y$  at those points. This, nevertheless, has no effect on the reconstruction of the curve.

Plots of the reconstructed profile of three ships of the CGN class using different value of  $S$ , result in different number of knot positions and are shown for comparison in Figure 4.3, 4.4 and Figure 4.5. The original samples are plotted in solid curve, the reconstructed curve is plotted in dashed line in Figure 4.3.

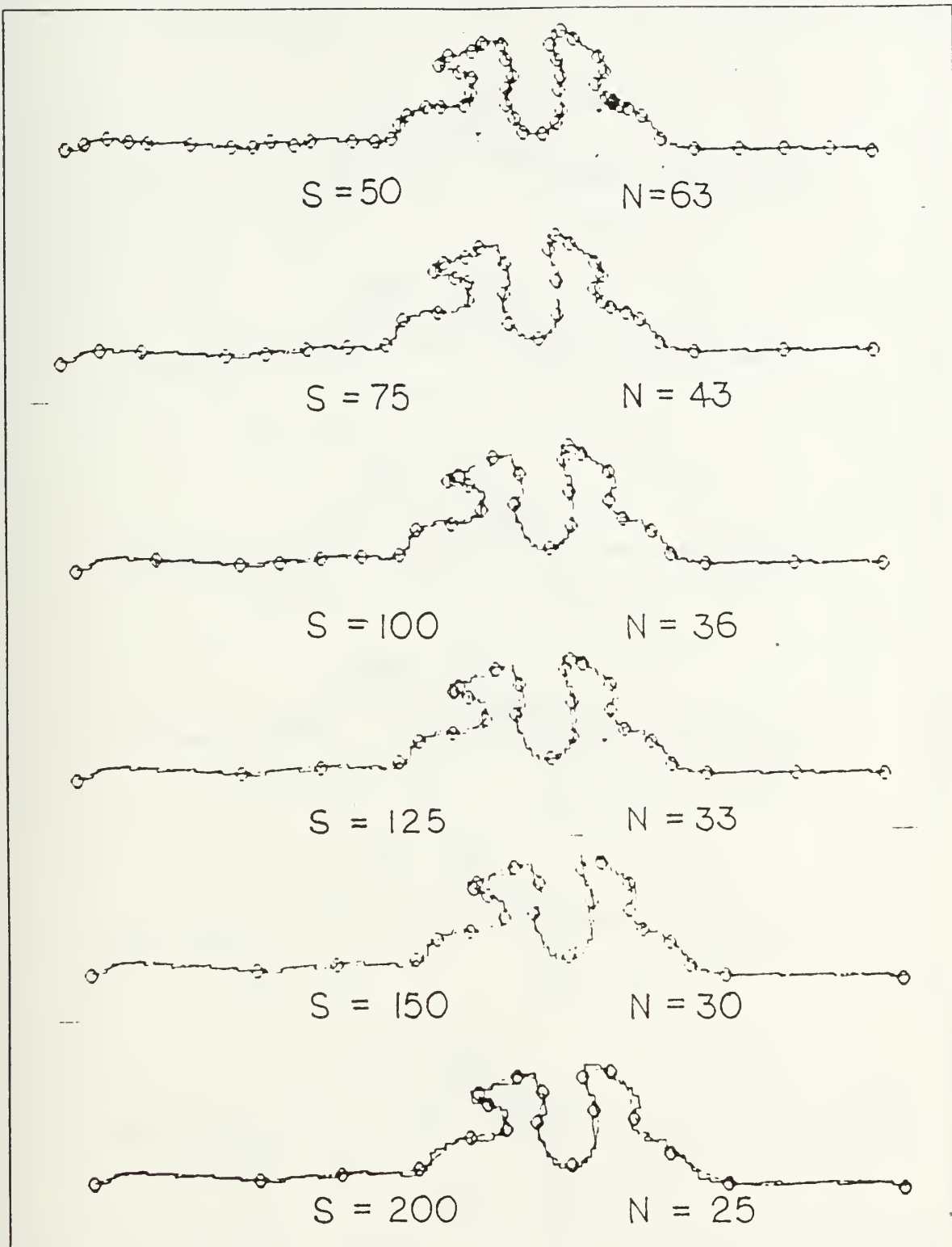


Figure 4.3 A CGN at Range of 45000 feet.

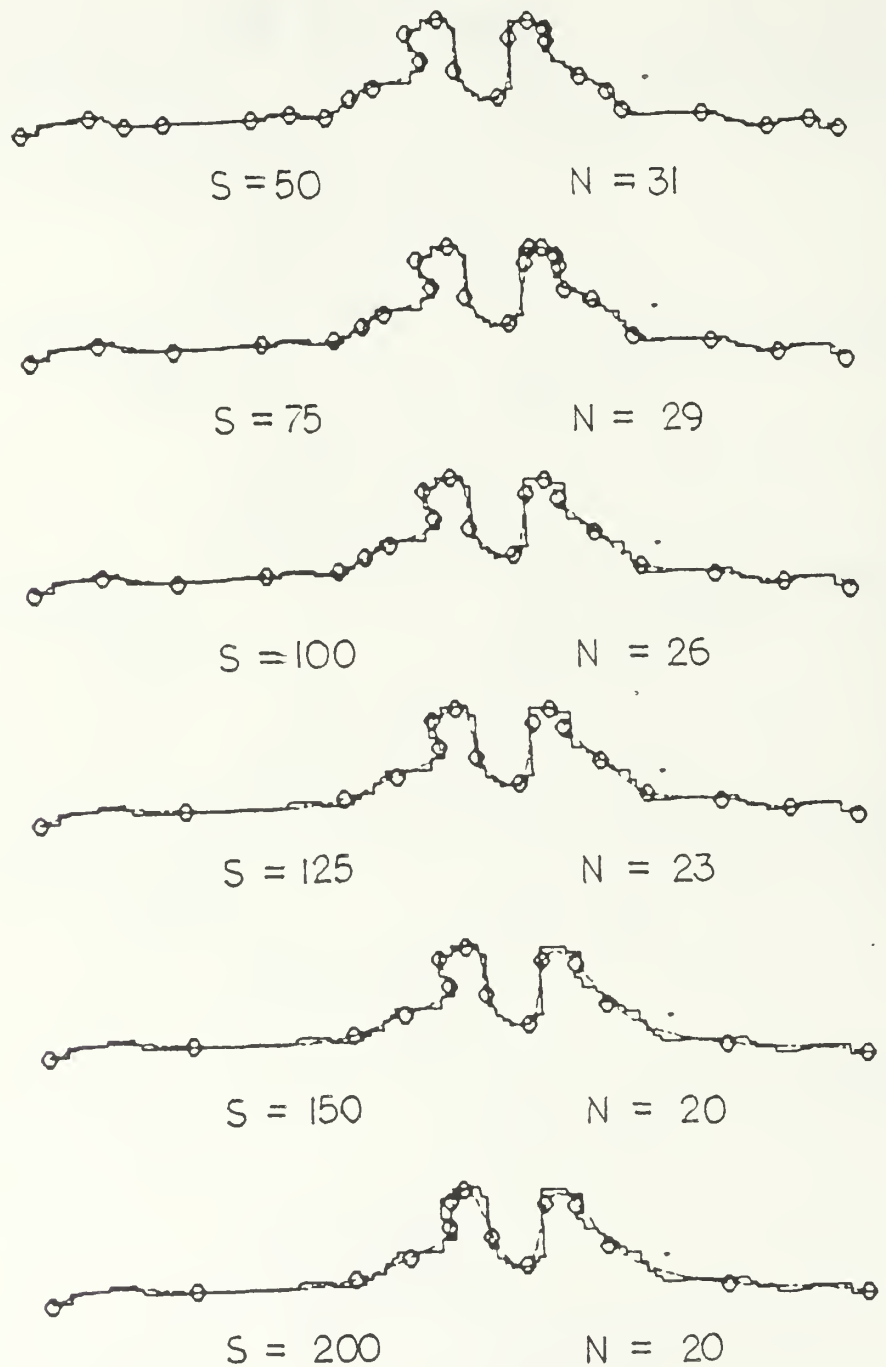


Figure 4.4 A CGN at Range of 55000 feet.

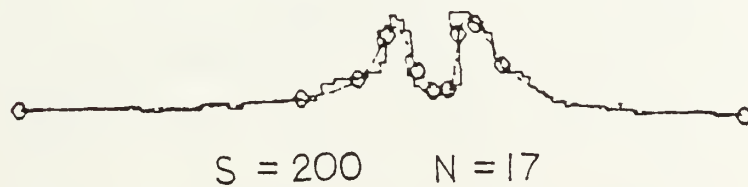
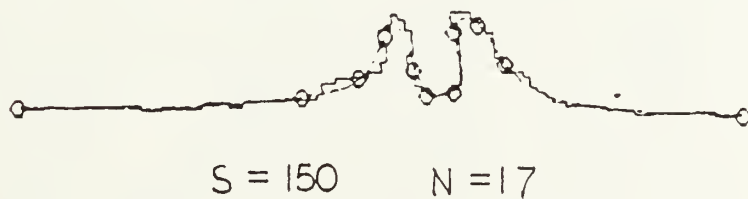
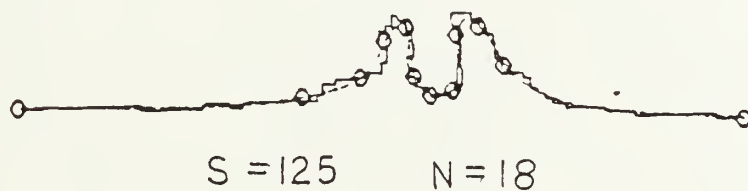
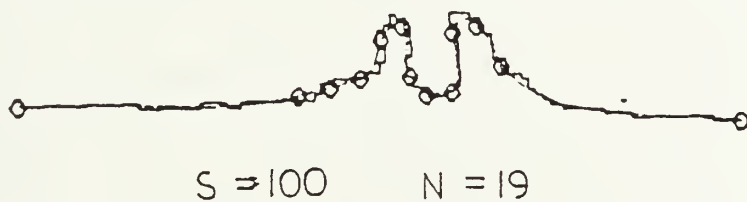
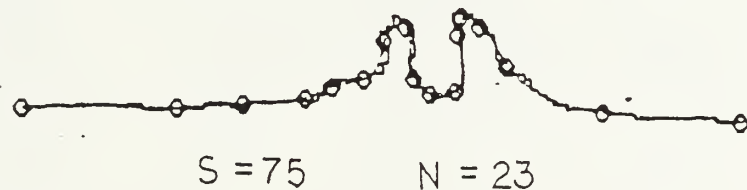
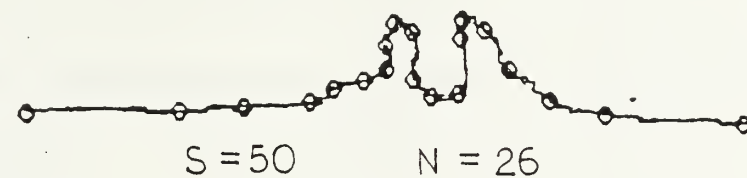


Figure 4.5 A CGN at Range of 64000 feet.



## 2. Selection of Smoothing Factor Value(S)

It is found by trials and errors using the computer program PARAM that in order to retain maximum information of the profile curve, the number of knot positions required should be in the range of 25 to 35. The number of knot positions depends upon the value of  $S$  which must be set accordingly. The appropriate choice of  $S$  to satisfy the condition stated above, is important. For the class of a guided missile cruiser(CGN), three ship images at 3 different ranges were selected. Then, run the appropriate program for various  $S$  factors to see how the number of knot( $N$ ) will vary. The results are shown in Figure 4.6. Plots of  $N$  vs  $S$  in Figure 4.7 through Figure 4.13 show that, in most cases, the value of  $N$  decrease quite rapidly when the value of  $S$  is in the range of 0 to 100, and gradually for  $S$  factor in the range of 100 to 200, thereafter, the value of  $N$  decreases very little. Obviously, the curve seems to decay exponentially. Furthermore, for some classes of ships changes are more pronounced than the others which is probably due to the actual number of knots present in the profile. For guided missile cruiser ship(CGN) with 2 lumps, the number of knot positions required can be 33 as shown in Figure 4.6 We select the factor  $S$  to be about 100.

The selection of the factor  $S$  depends upon the number of the original sampling points. If the factor  $S$  is small, large number of knots are needed. When the factor  $S$  is large, small number of knots are needed. When the number of knot and the  $C_x$  and  $C_y$  coefficients are small, the B-spline coefficients  $C_x$  and  $C_y$  obtained can not be used to reconstruct the curve close to the original profile.

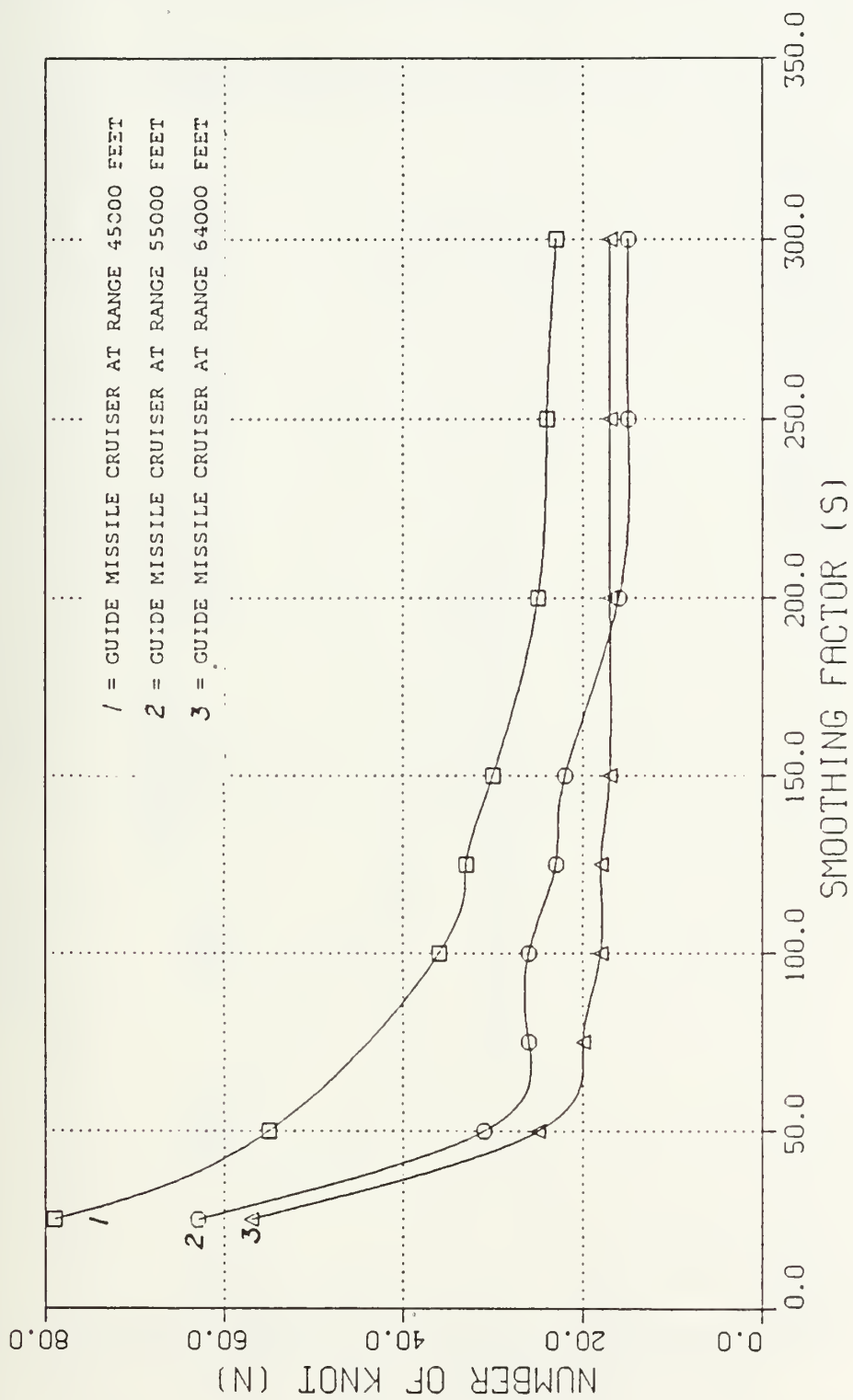


Figure 4.6 Plot N vs S for a CGN.

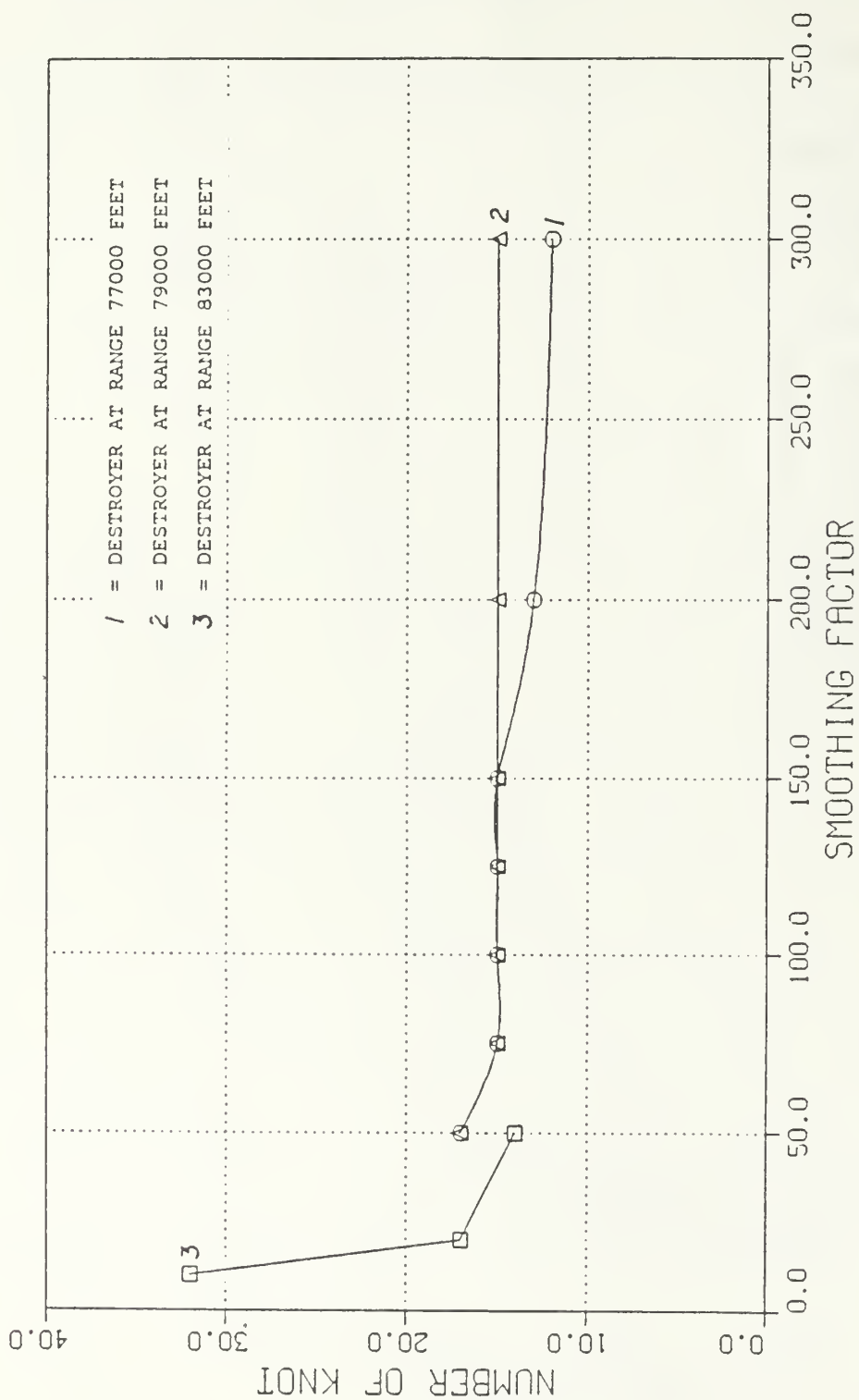


Figure 4.7 Plot N vs S for a DD.

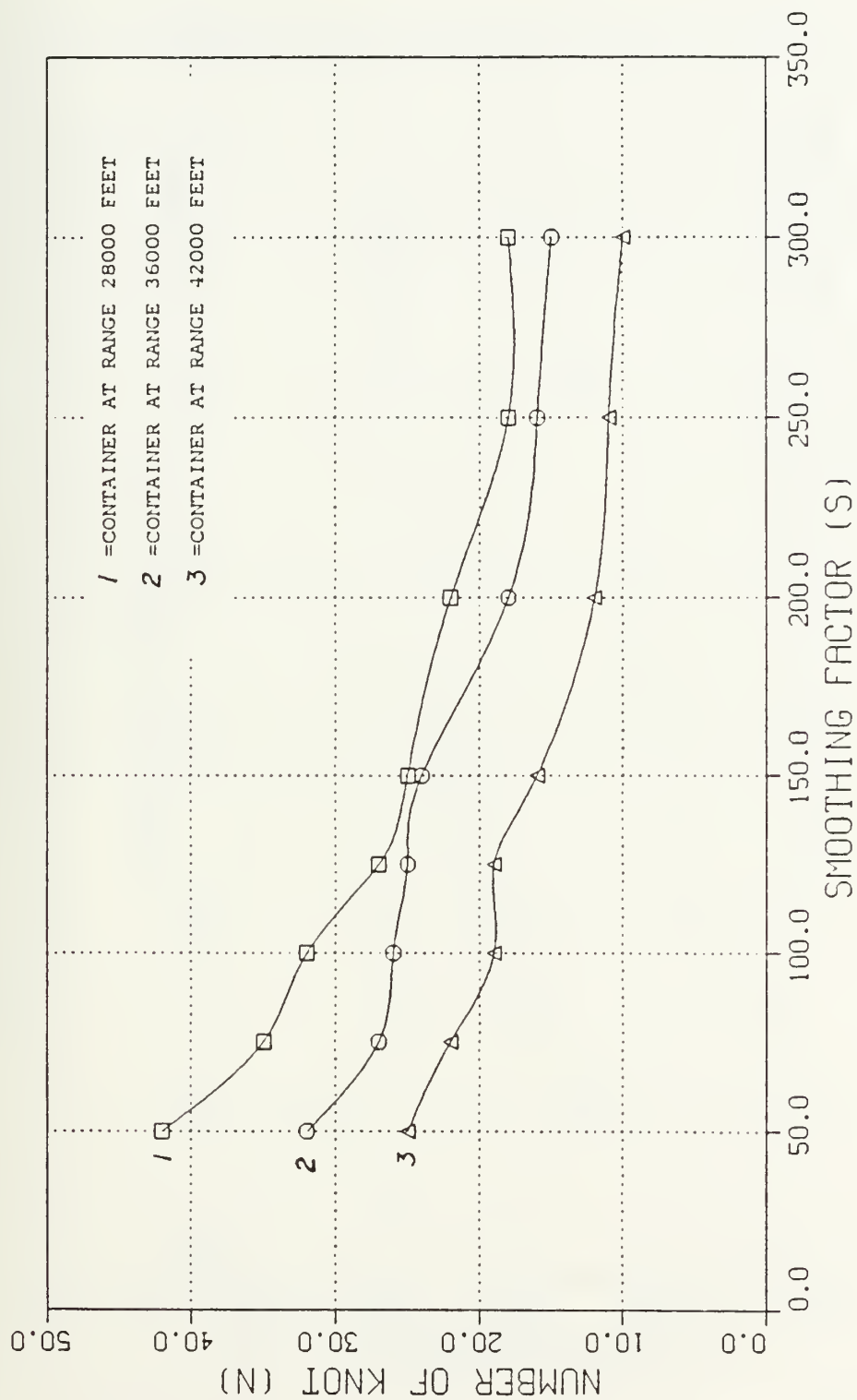


Figure 4.8 Plot N vs S for a Container.

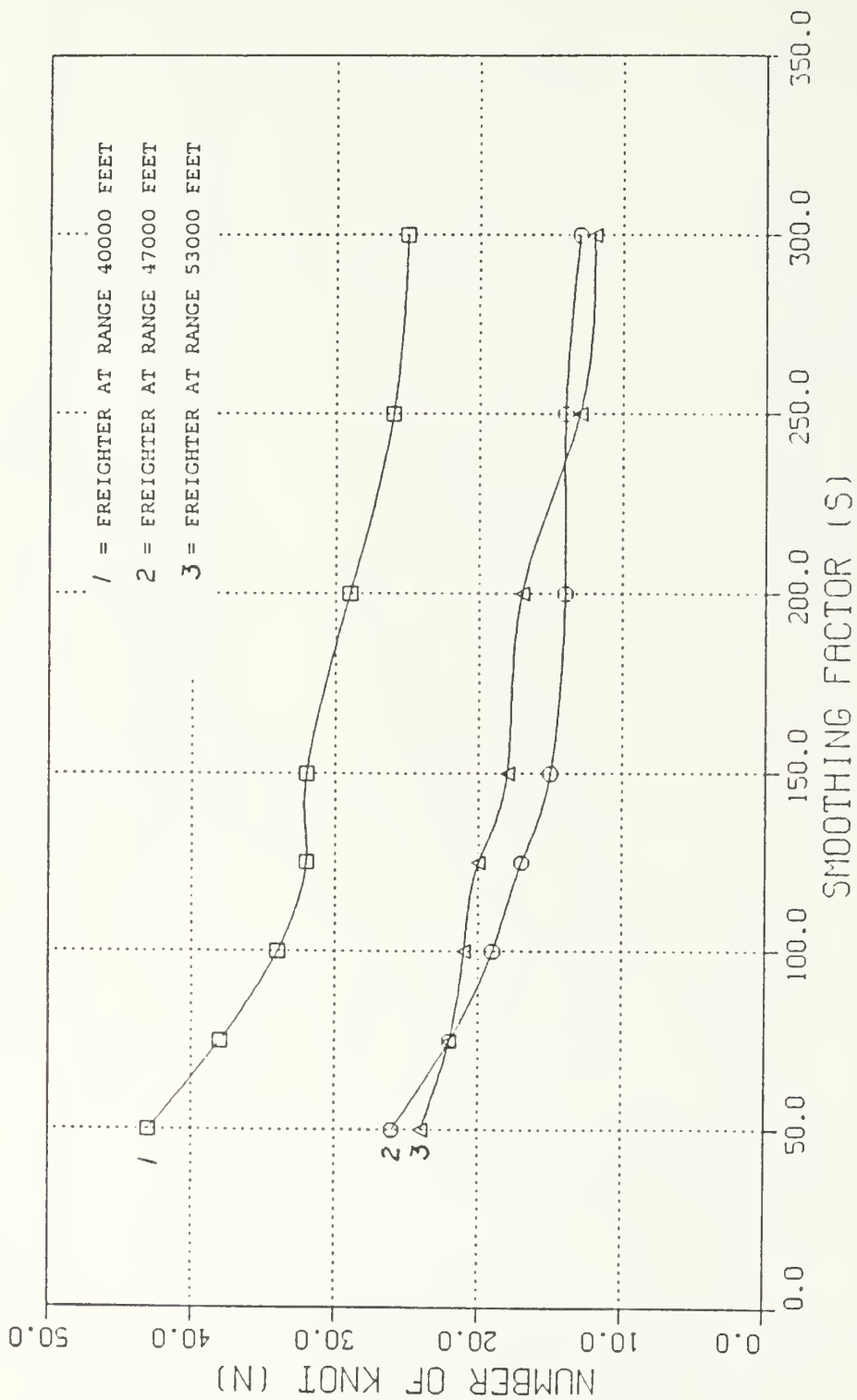


Figure 4.9 Plot N vs S for a Freighter.



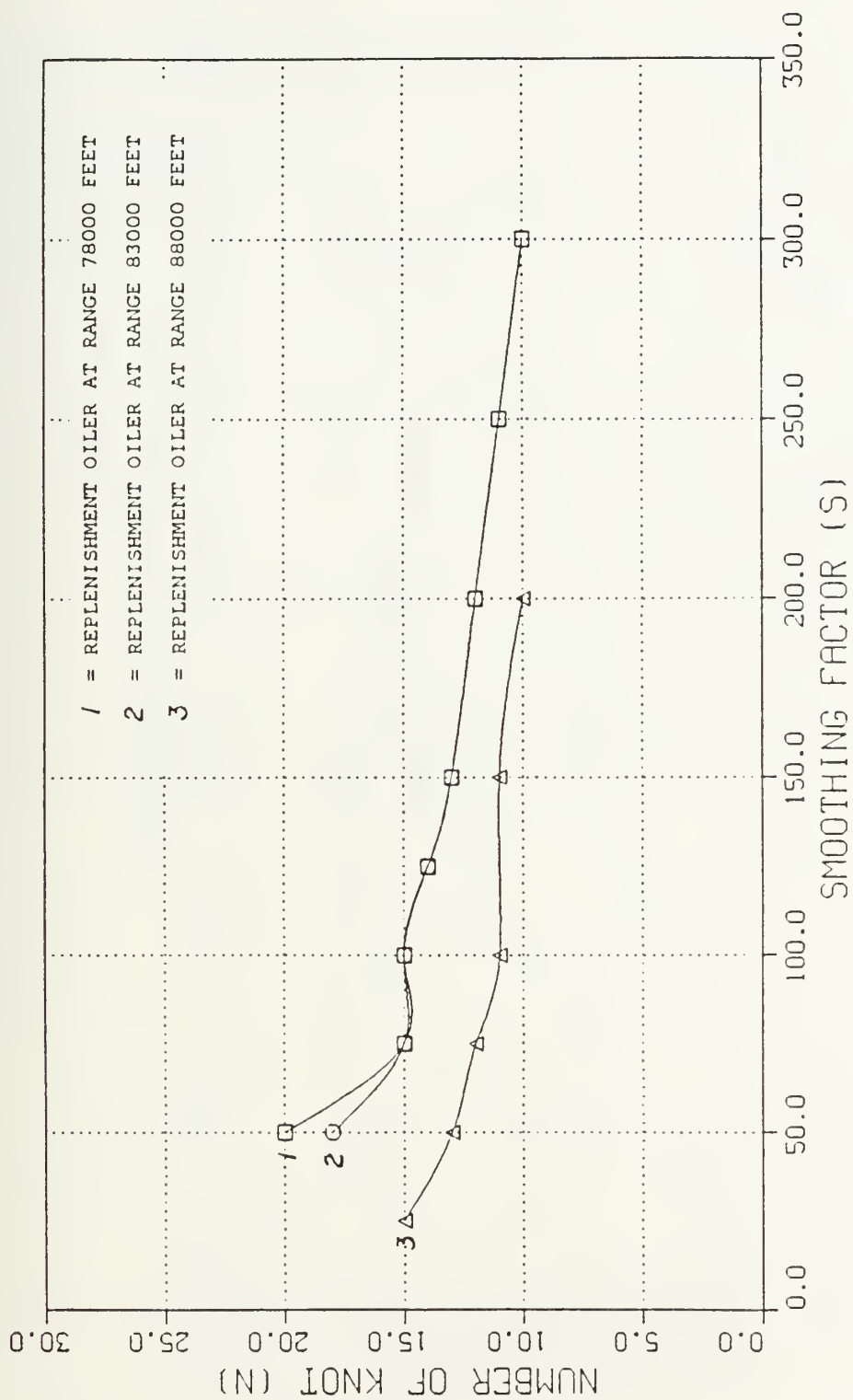


Figure 4.10 Plot N vs S for a AOR.

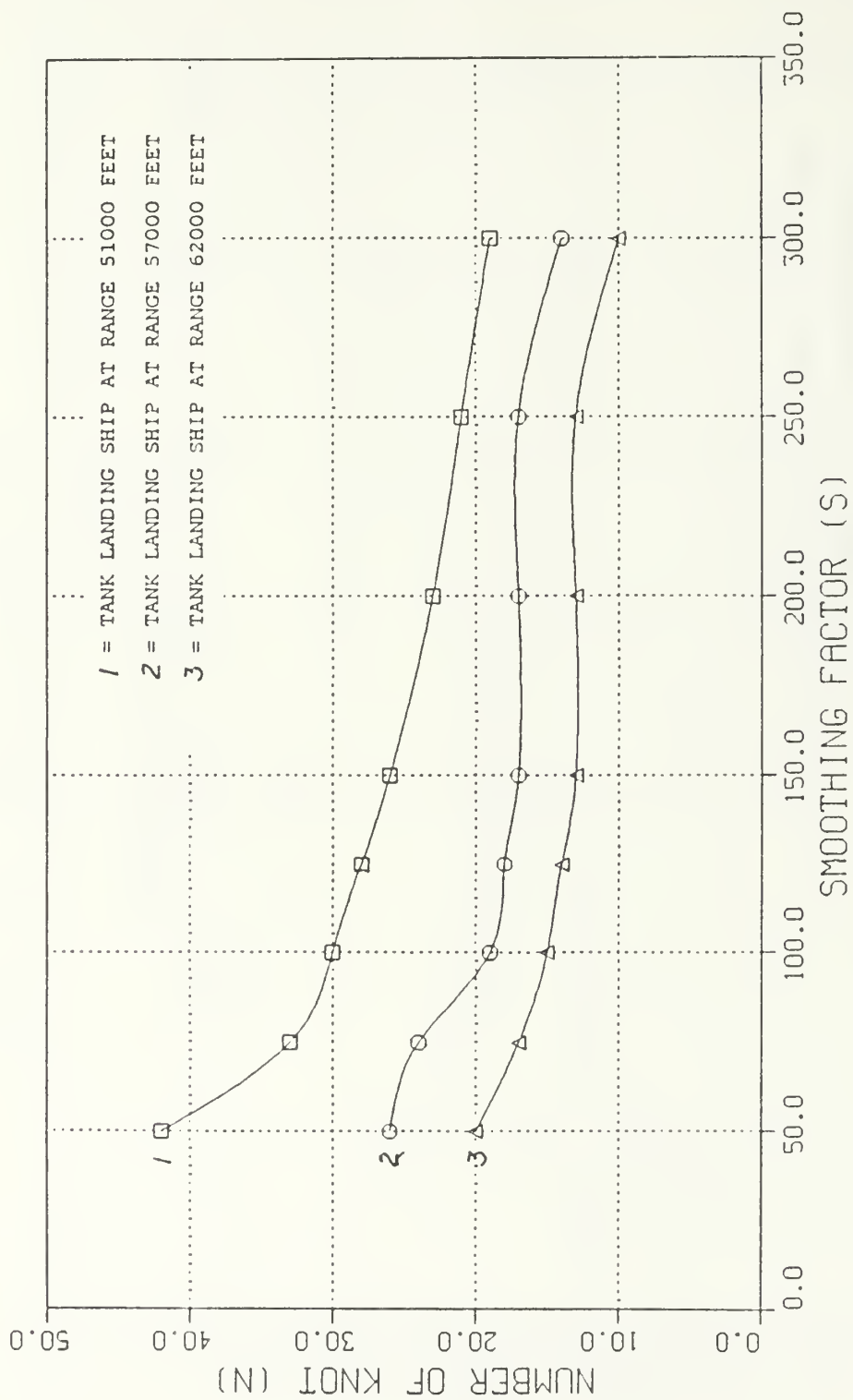


Figure 4.11 Plot N vs S for a LST.

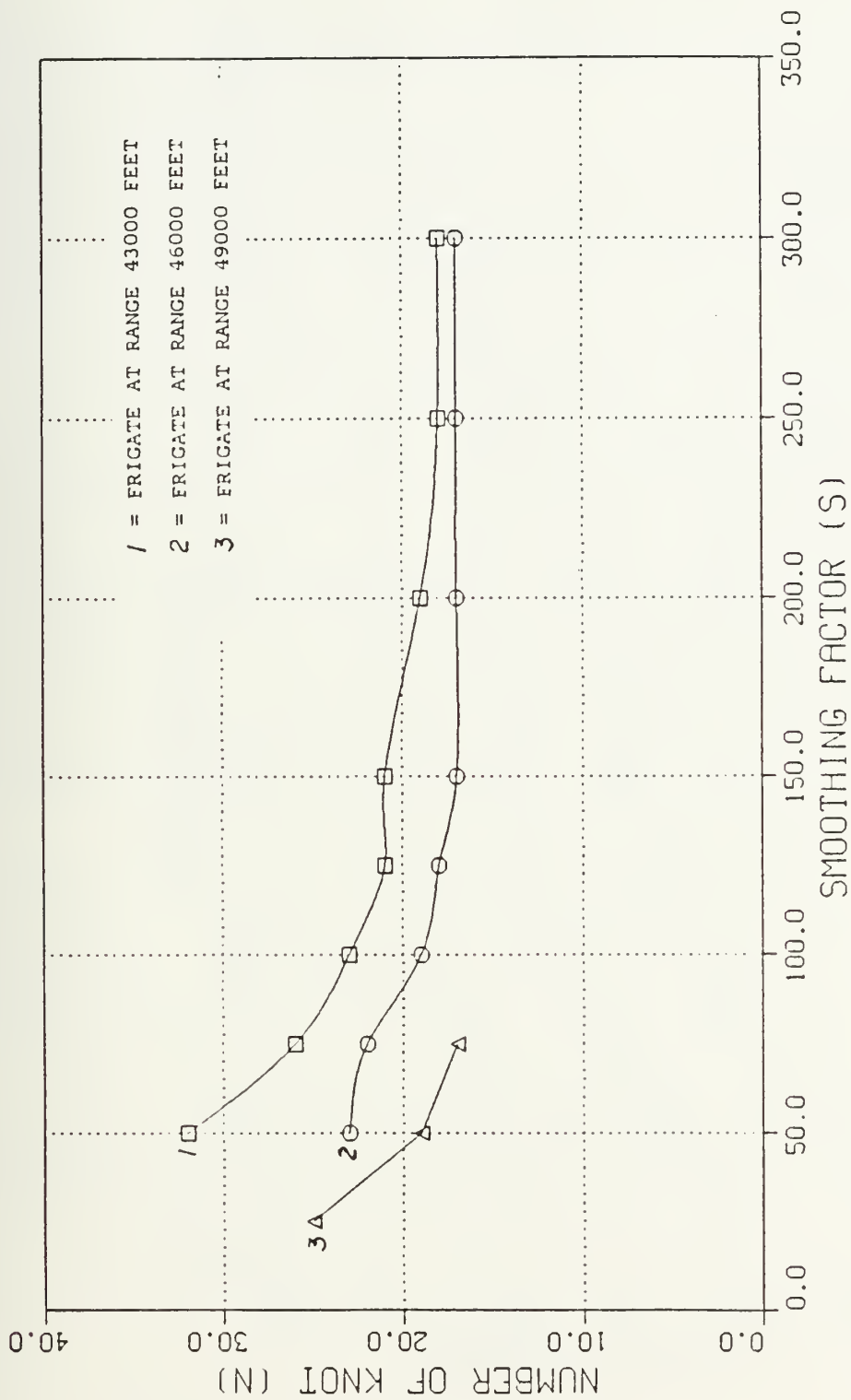


Figure 4.12 Plot N vs S for a FF.

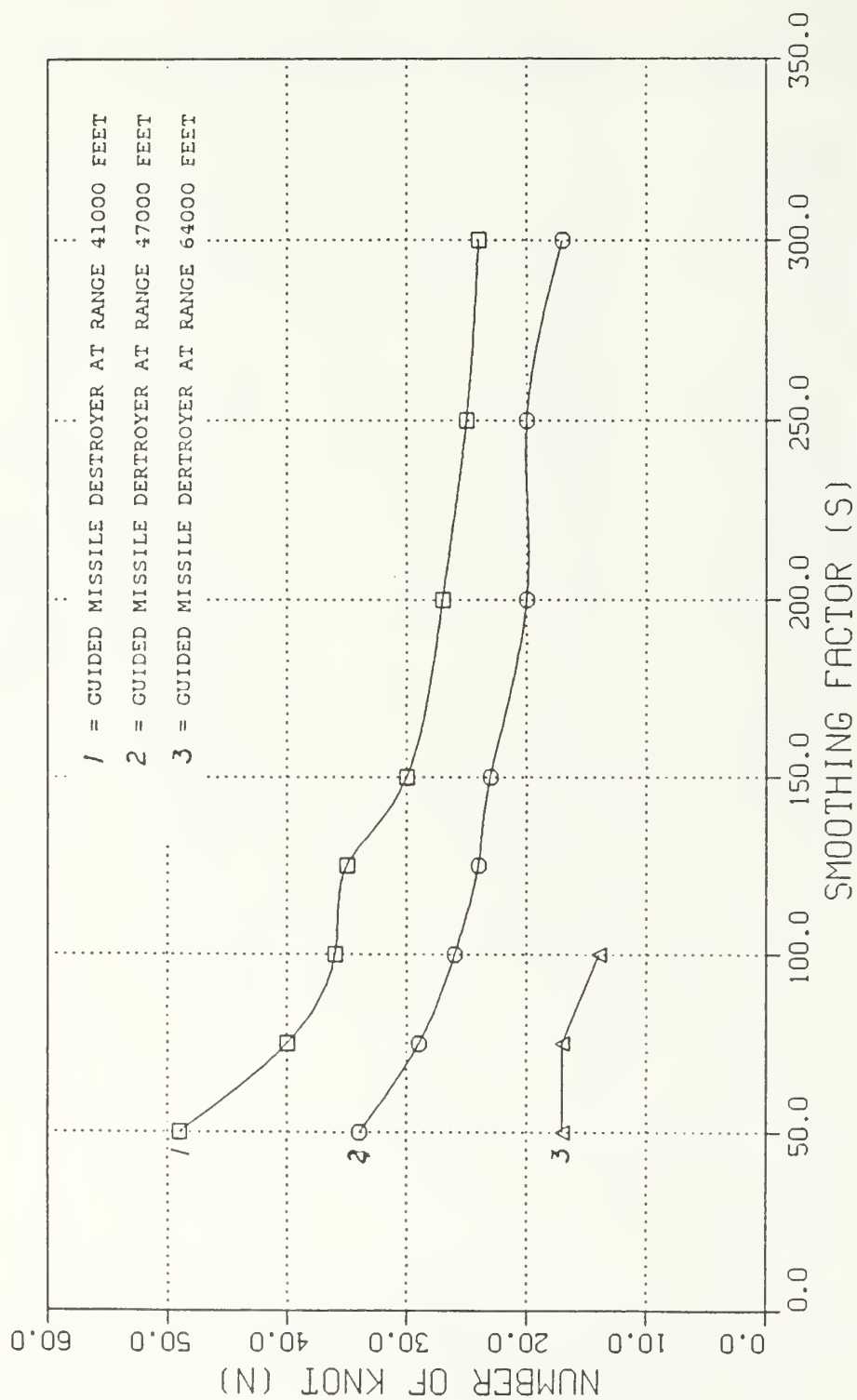


Figure 4.13 Plot N vs S for a DDG.

### 3. The Output Cx,Cy and Knots Profiles

From the previous study the factor S is selected for each of the class of ship as follows

1. Destroyer(DD), S = 20.0
2. Container, S = 100.0
3. Freighter, S = 100.0
4. Replenishment oiler, S = 20.0
5. Tank landing ship(LST), S = 25.0
6. Frigate(FF), S = 100.0
7. Guided missile cruiser(CGN), S = 125.0
8. Guided missile destroyer(DDG), S = 125.0

The plot of the B-spline coefficients, Cx and Cy, and X, Y at the positions of the sampling points vs the Z parameter are shown in Figure 4.14 through Figure 4.21. Observation and comparisons of the curves show that the values of Cx and Cy exhibit changes similar to that of X and Y except that the variation of values leads that of the X and Y. This is due to the fact that Cx and Cy have to act as the controlling factor for the reconstructed B-spline curve to get the result close to the original curve.

Examination of the plots of the results of X and Y show that when X is increasing monotonically, Y is almost constant; but when X is almost constant, Y is increases or decreases. This behavior relating to profile reconstruction may be explained as follows. For a ship profile when X is increasing and Y not increase too much, this may be interpreted as an almost leveled profile. When X is almost constant, and Y may be increasing or decreasing, it may be interpreted as the beginning or the ending of the lump.



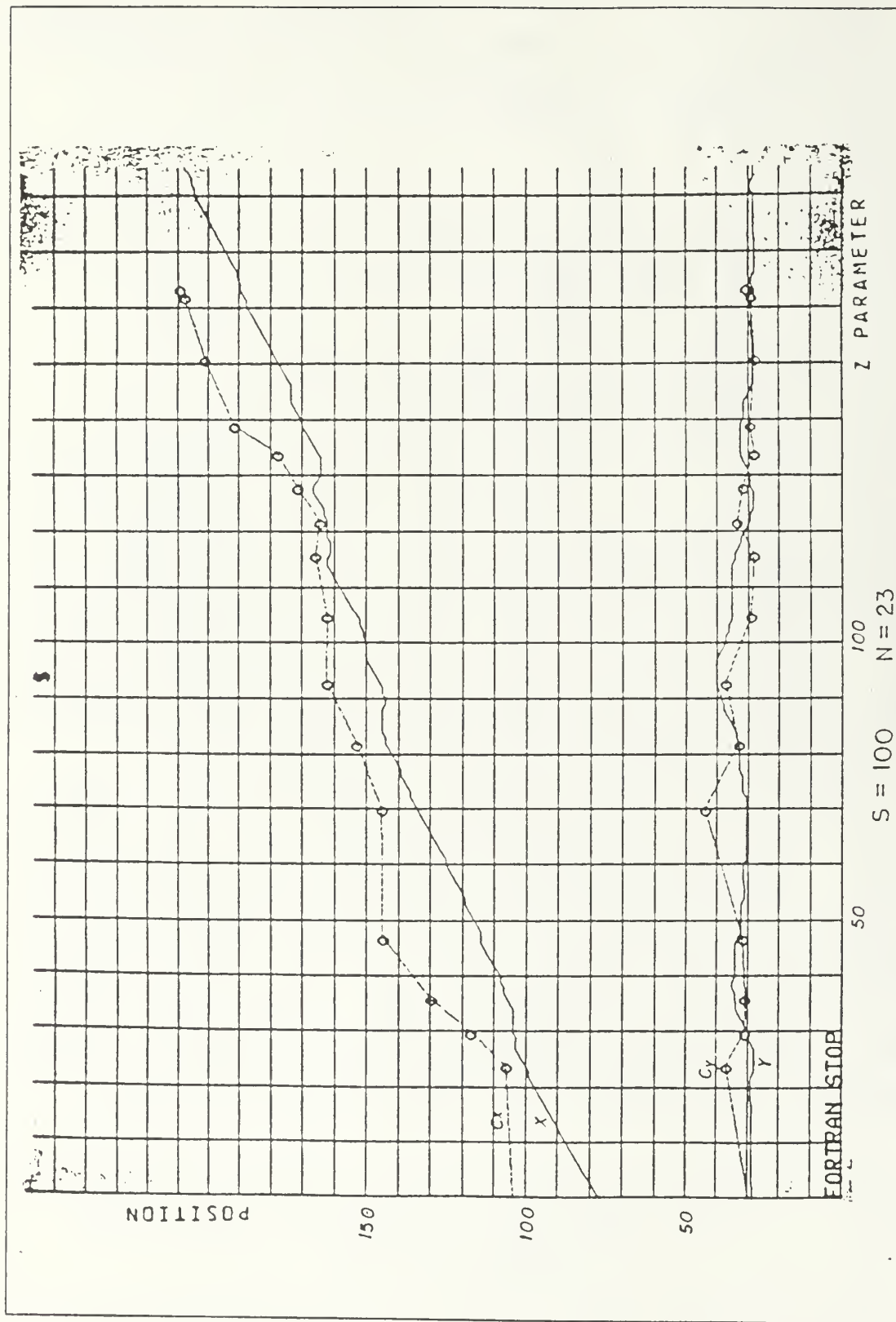


Figure 4.14 Plot X,Y,Cx, and Cy vs Z for a FF at a Range of 43 K-ft.

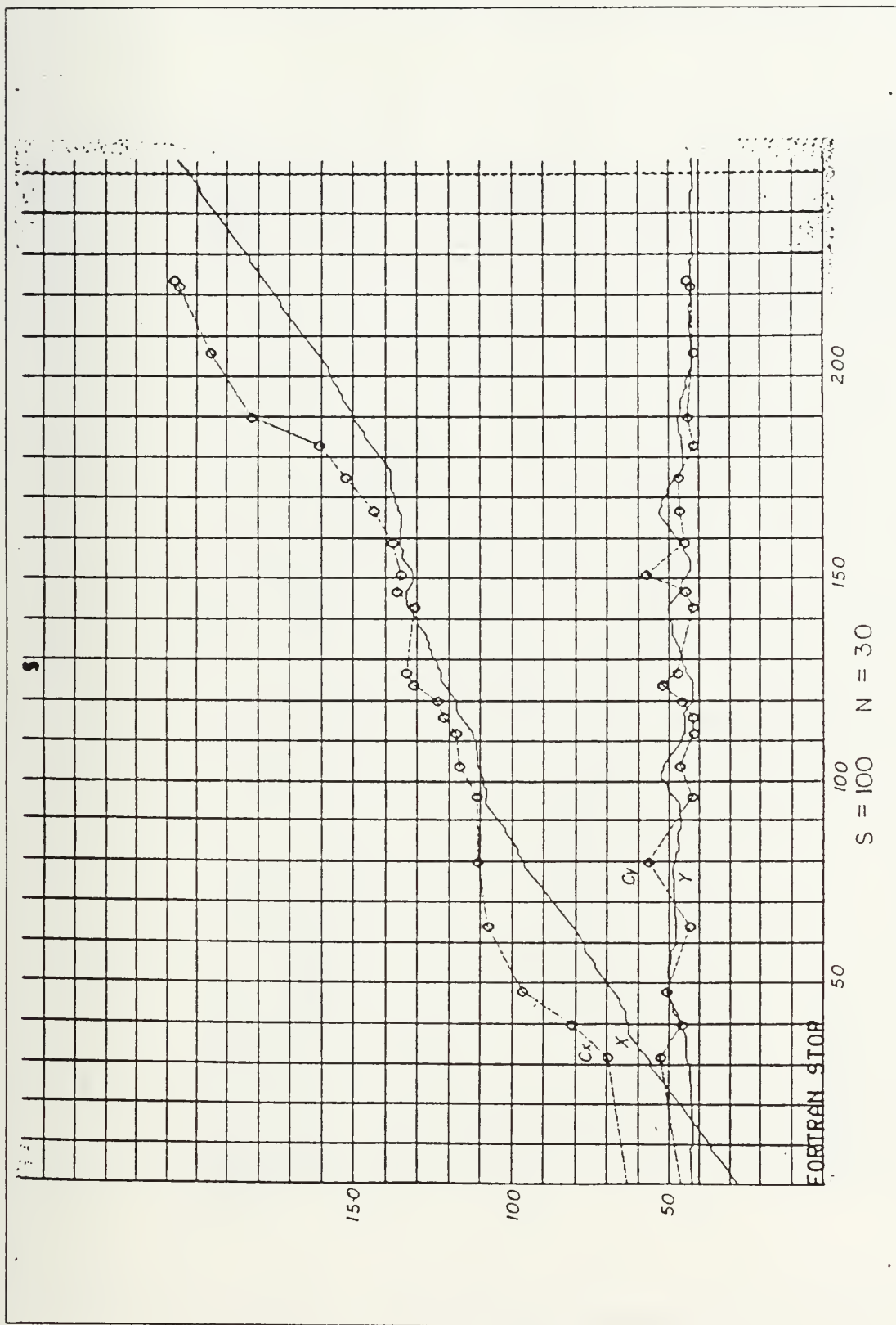


Figure 4.15 Plot X,Y,Cx, and Cy vs Z for a LST at a Range of 51 K-ft.

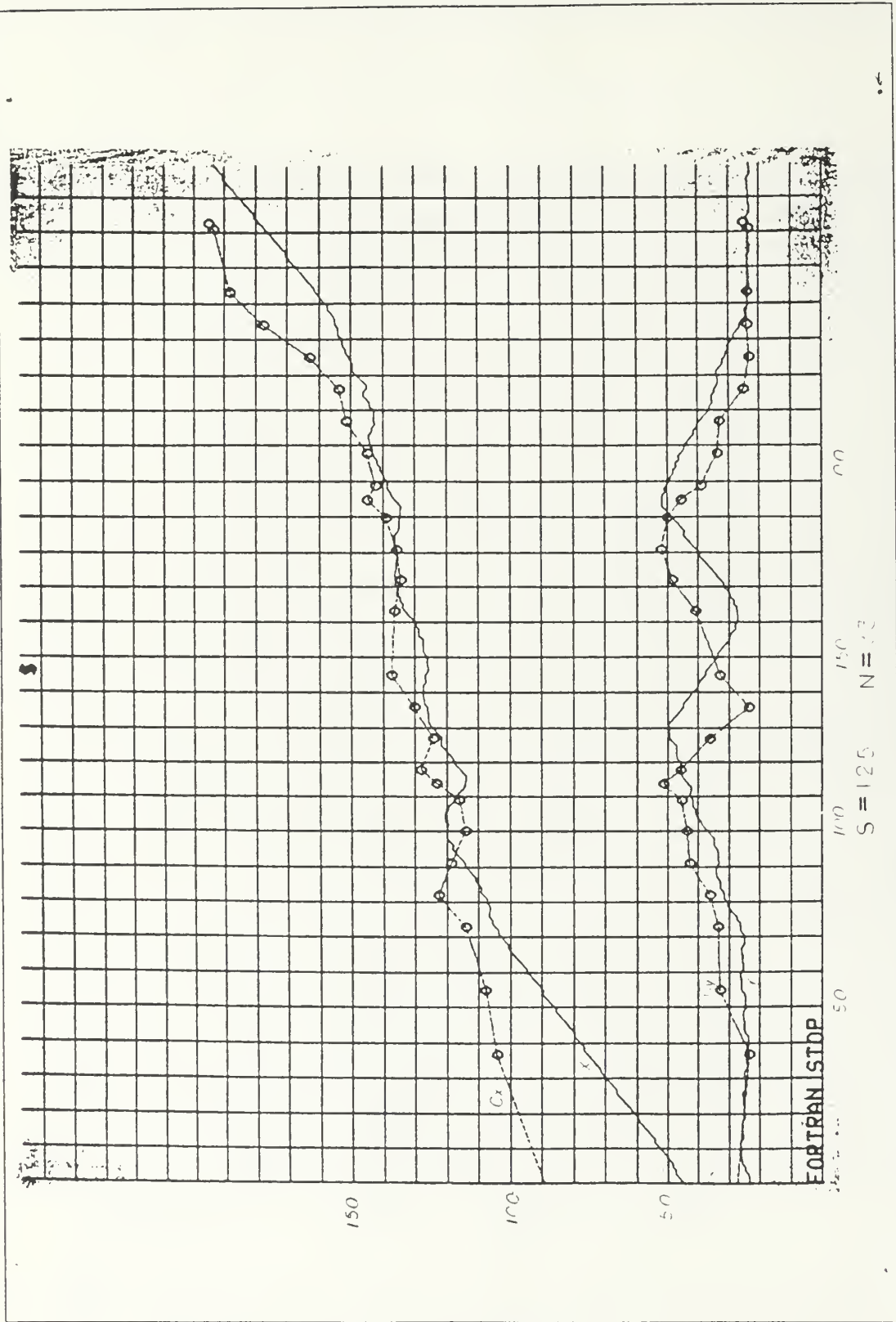


Figure 4.16 Plot X,Y,Cx, and Cy vs Z for a CGN at a Range of 45 K-ft.

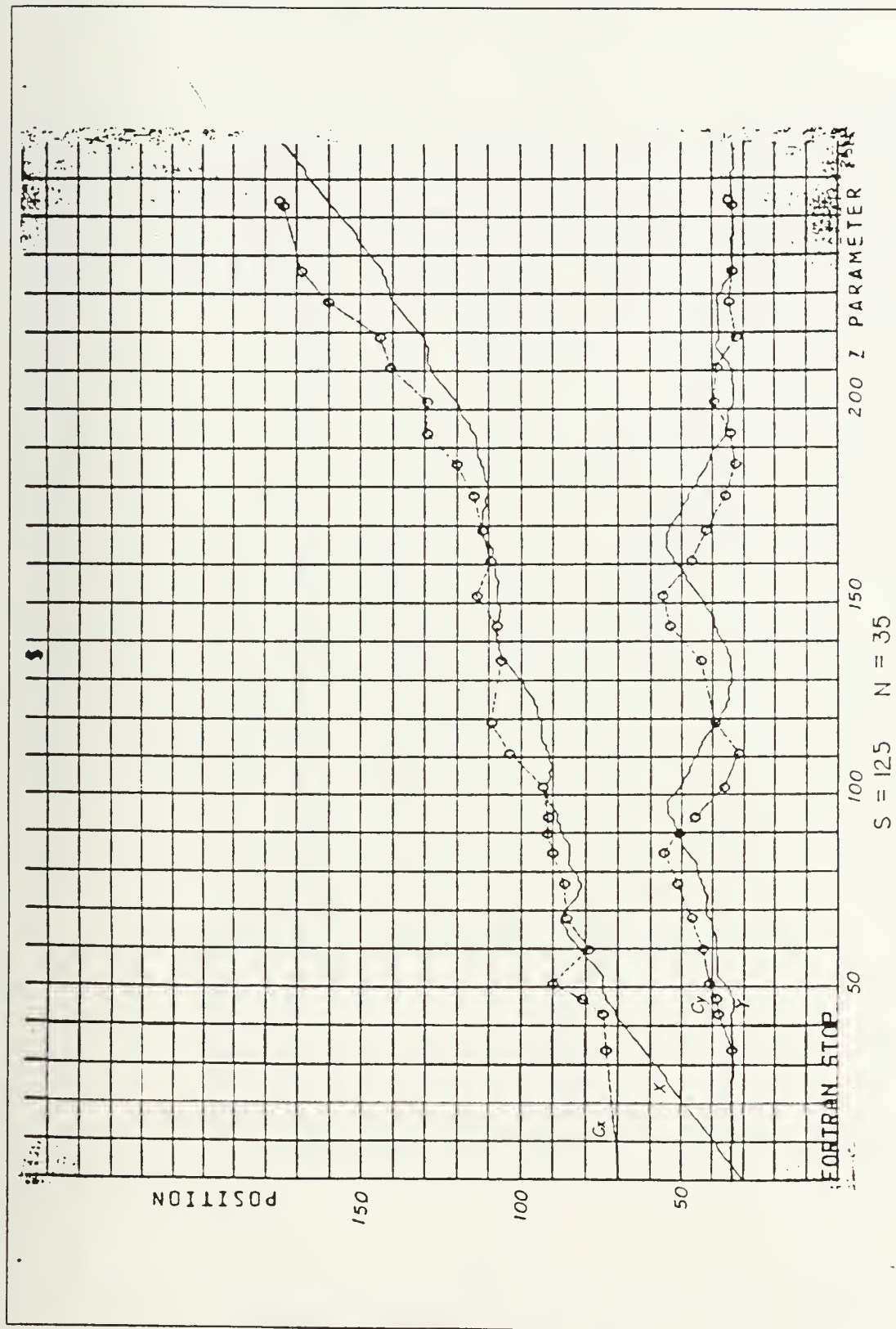


Figure 4.17 Plot X,Y,Cx, and Cy vs Z for a DDG at a Range of 41 K-ft.

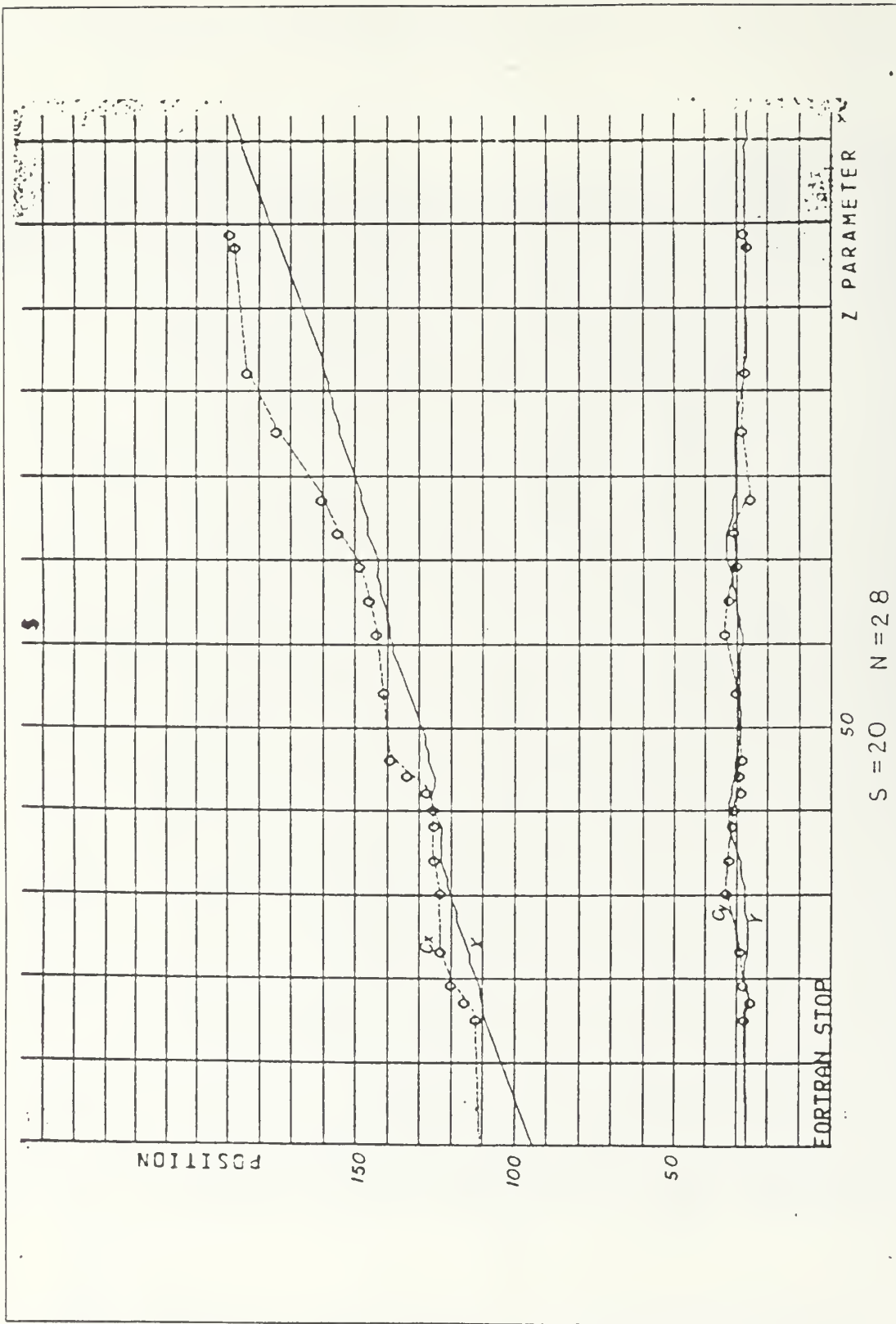


Figure 4.18 Plot X,Y,Cx, and Cy vs Z for a DD at a Range of 77 K-ft.

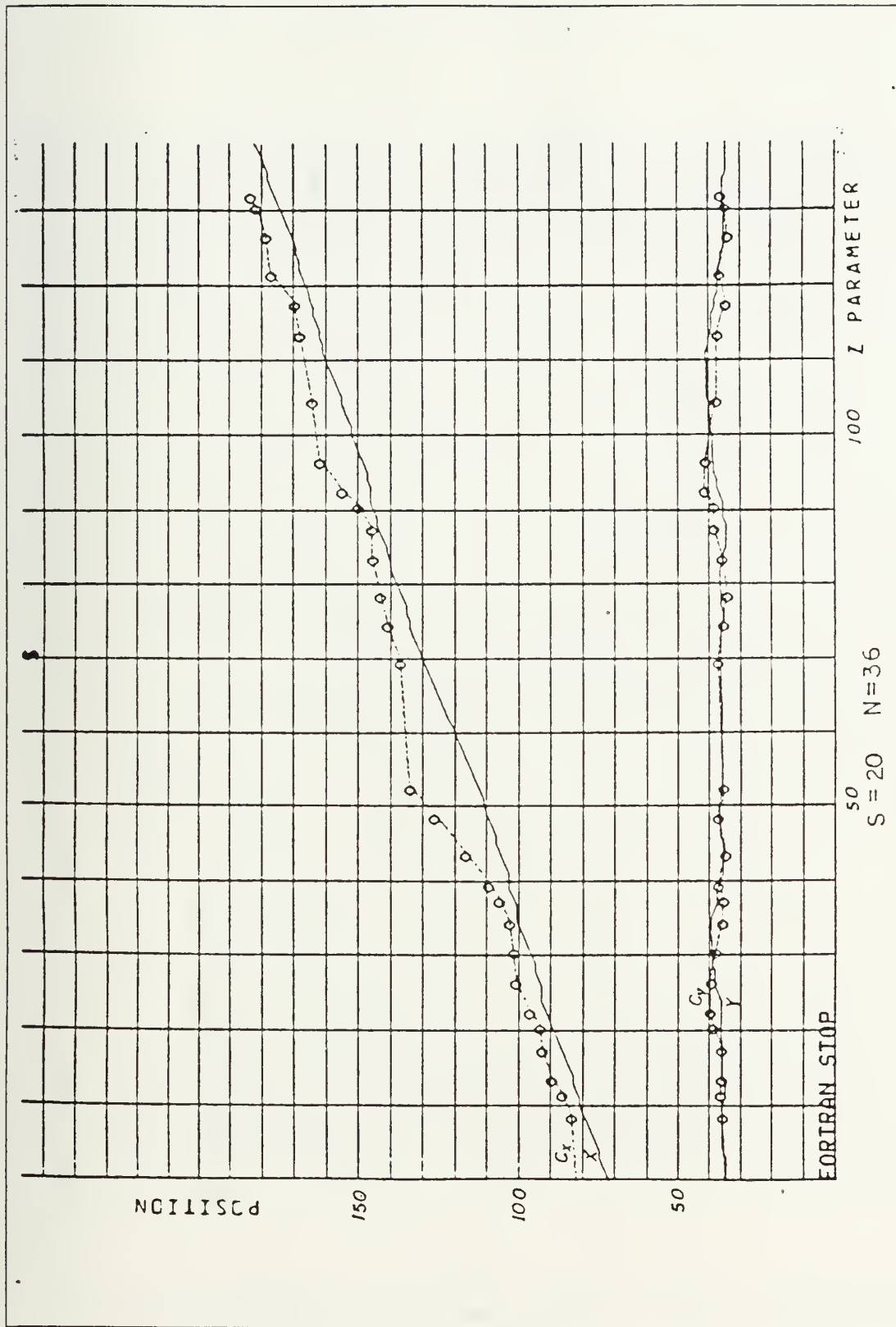


Figure 4.19 Plot X,Y,Cx, and Cy vs Z for a AOR at a Range of 78 K-ft.



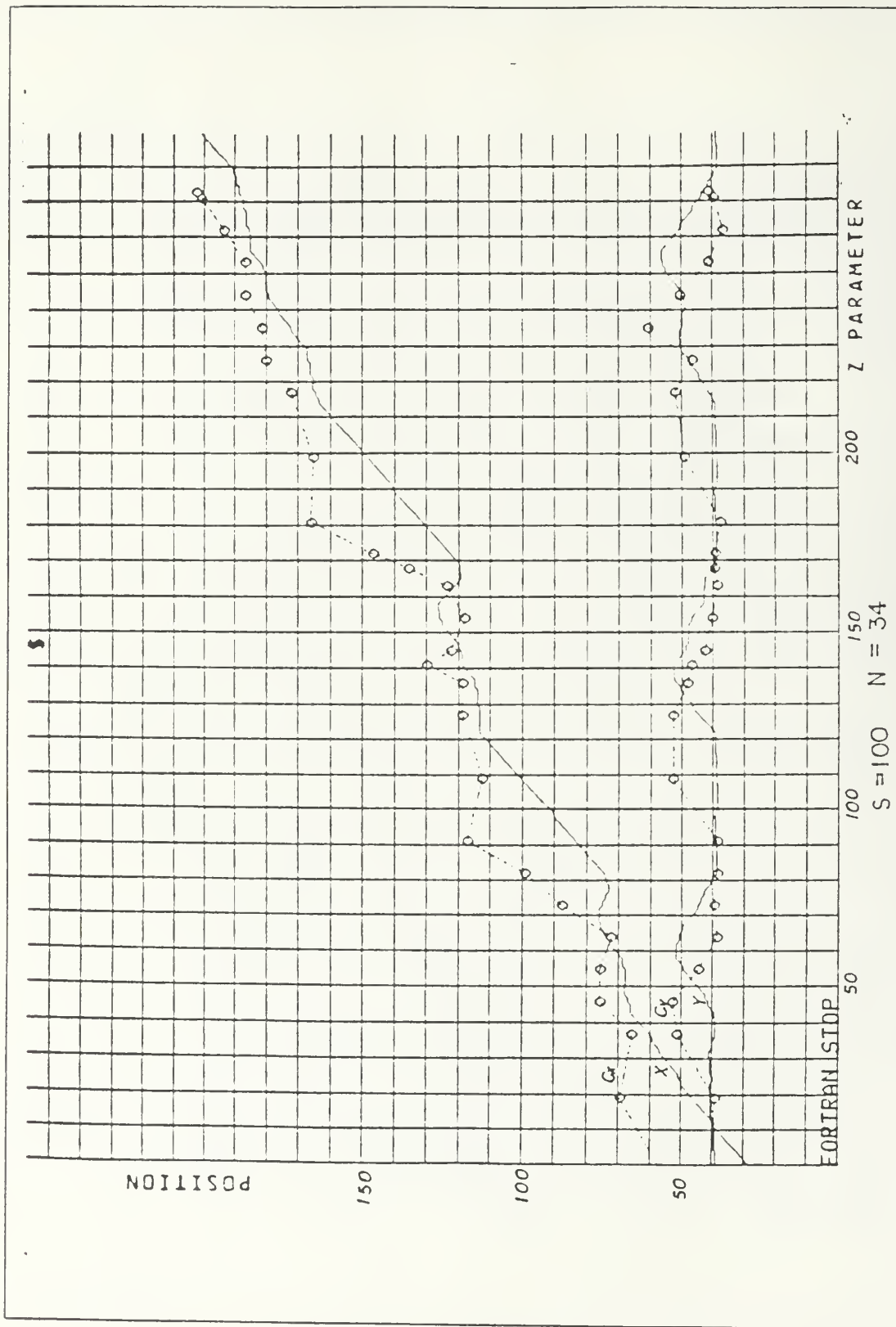


Figure 4.20 Plot X,Y,Cx, and Cy vs Z for a Freightier at a Range of 40 K-ft.

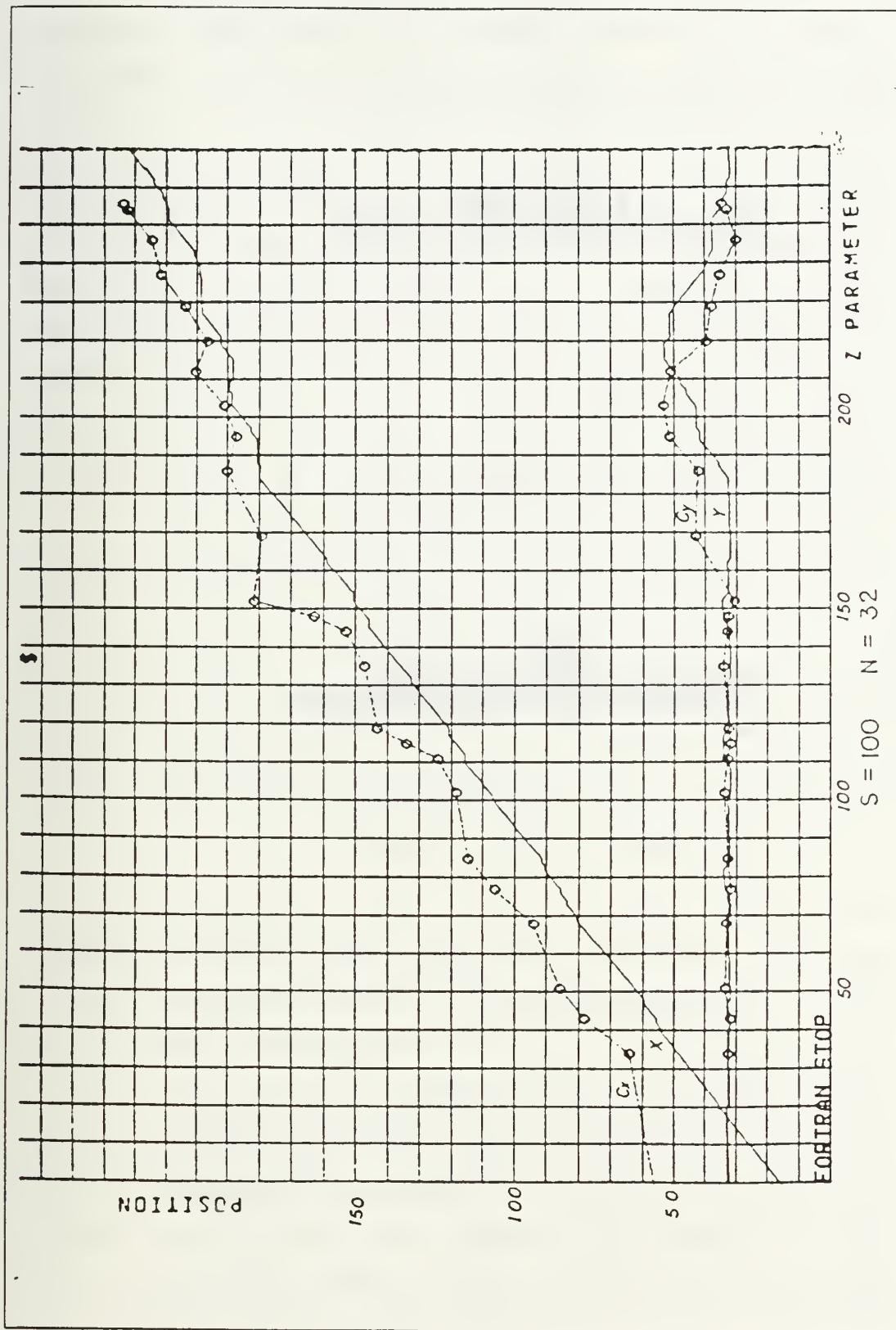


Figure 4.21 Plot X,Y,Cx, and CY vs Z for a Container at a Range of 28 K-ft.

#### D. SHIP DESCRIPTION

The shape of the ships depend upon the shapes and positions of the lumps. Characteristics of the lumps for different type of ships are as follows:

1. Frigate - The beginning of the lump is at  $1/6$  of the ship length to the left side of the midships and the peak of the lump is at the mast which is located at the midships. The termination of the lump is approximate  $1/3$  of the ship length from the midships to right side. In addition, the average height of the lump is a little higher than the level between the bow and the stern, and its size is  $1/2$  of the ship length as shown in Figure 4.22.

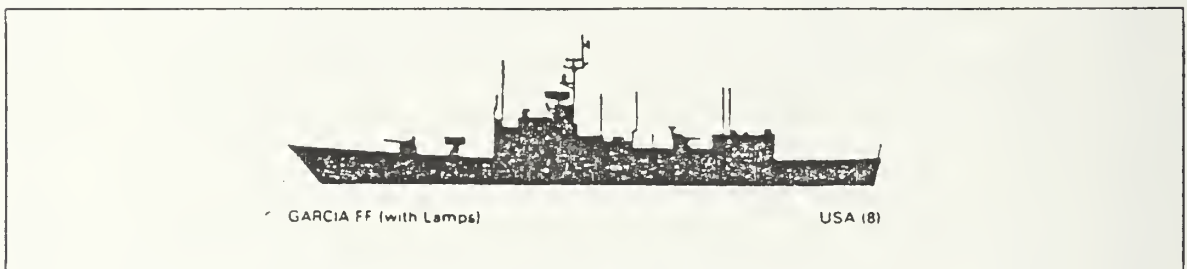


Figure 4.22 Frigate.

2. Container - The beginning of the lump is at  $1/3$  of the ship length to the right side of the midships while the lump is high and terminate at the stern. The lump appears to be in a rectangular shape with small crane.
3. Tank landing ship(LST) - The beginning of the lump is at  $1/4$  of the ship length from the midships to the left side; the height of the lump is higher than the level between the bow and the stern by a small margin, while its highest point is located at approximate  $1/6$  of the ship length from the midships

to the left side with small difference from the average high as shown in Figure 4.23.

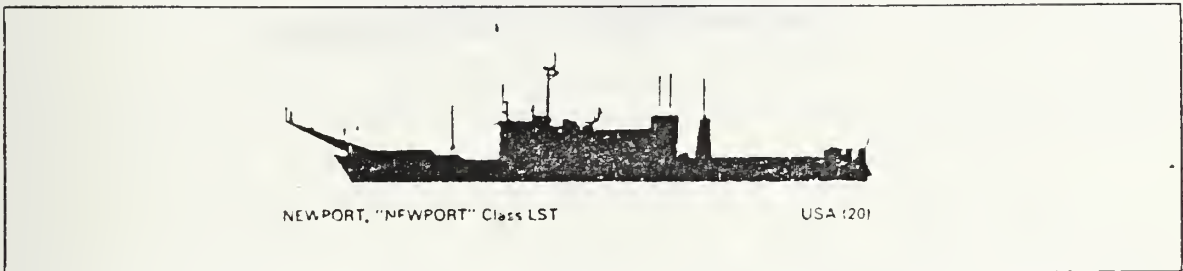


Figure 4.23 Tank Landing Ship(LST).

4. Guided missile destroyer - the beginning of the lump starts at approximately  $1/4$  of the ship length from the midships to the left side with its highest point at the mast. After this the slope will decline in a very rapid fashion following a noticable deck distance, then the beginning of the second lump occurs due to the redome presence. Therefore, the lump will be narrow with great height, and will terminate at approximately  $1/6$  of the ship length from the midships to the right side. Furthermore, there is a little lump near the stern, this distinguish destroyer of the same size as shown in Figure 4.24.
5. Destroyer-lump characteristic will be similar to that of guided missile destroyer except for the small lump as in Figure 4.25.
6. Guided missile cruiser - The beginning of the first lump is at  $1/12$  of the ship length from the midships to the left side with highest point at the mast. The lump size is large both in length and height and its height decreases to the point which is a little above the level between the bow and the stern. Therefore, the second lump begins with almost the same size as



Figure 4.24 Guided Missile Destroyer(DDG).

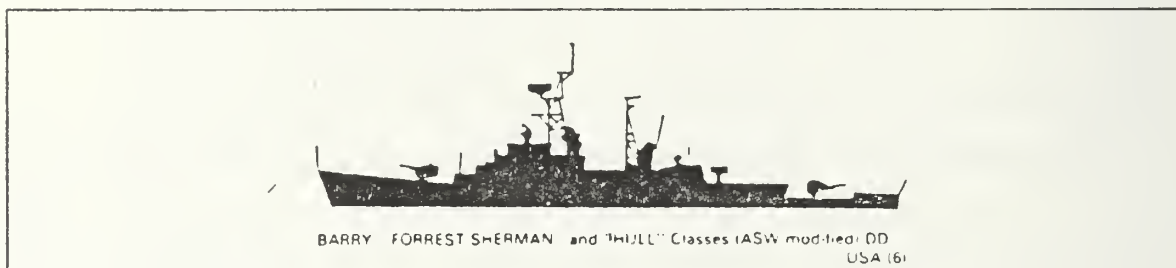


Figure 4.25 Destroyer.

the first one. The second lump ends at approximately 1/4 of the ship length from the midships to the right side. Furthermore, the distance between the peak of both lumps is less than that of the Guided missile destroyer shown in Figure 4.26.

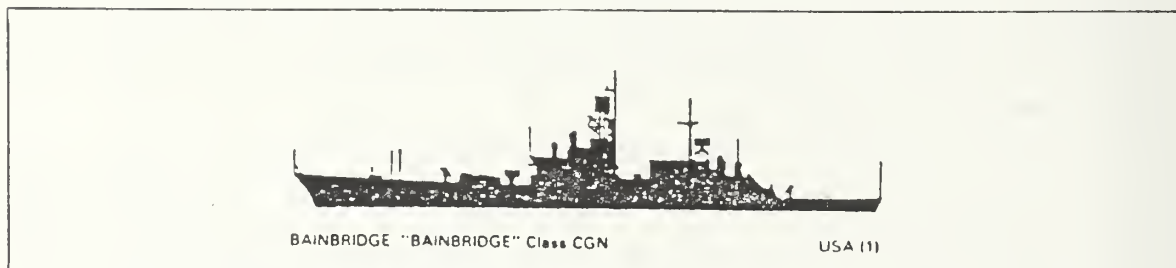


Figure 4.26 Guided Missile Cruiser(CGN).



7. Replenishment oiler (AOR) - There are 2 little lumps with the first one starting at  $4/5$  of the ship length from the midships to the left side exhibiting rapid decrease in height to the level between the bow and the stern. The second lump starts at  $1/4$  of the ship length from the midships to the right side with slow decline to the point near the stern.
8. Freighter-with 3 noticeable lumps, the first two represent lifting crane with the first high lump approximately  $1/6$  of the ship length from the midships to the left side but narrow; meanwhile, the second lump is located at approximately the midships with the same size as the first one. The beginning of the third is at approximately  $1/3$  of the ship length from the midships to the right side with large size. It is higher than the first two and with gradual decline to the stern.

From the lump characteristic of different type of ships, we could distinguish the type of a ship based on the rotated superstructure.

#### E. SHIP CLASSIFICATION

An important aspect in categorizing types of ships is to recognize the lumps above the main deck. Therefore, it is useful to plot the positions of  $X$  and  $Y$  vs the  $Z$  parameter where the  $Z$  parameter can be determined from eq(4.5) where  $Z(I)$  is an array (1..M) as shown in Figure 4.27 where  $X$  is plotted in solid line and  $Y$  is plotted in dash line. Then, plot the B-spline coefficients  $C_x$  and  $C_y$  vs  $T(N)$ ; the knots position in  $Z$  where  $N$  is the total number of knots as shown in Figure 4.28. The  $C_x$  is plotted in solid line. The  $C_y$  is plotted in dash line. The Values of  $C_y$  will be close to the curve of  $Y$  while the values of  $C_x$  for the same knot posi-



tions as  $C_y$  is exhibiting monotonic increasing trend. When the difference in  $C_x$  is decreasing or zero, the value of  $C_y$  will have a pronounced change where the beginning or the ending points of a lump can be detected. The value of knot position( $T$ ) at those points related to the sizes of the  $C_x, C_y$  can be determined. Finally, with the values of  $C_x$  and  $C_y$  at those points known, the area of the lumps can also be determined.

Hence, from the ship's characteristics and information derived from the above procedure, classification of ships can be made by considering

First, the number of lumps detected, 1, 2, or 3.

1. The number of lump=1: Frigate, Tank landing ship
2. The number of lump=2: Destroyer, Guided missile cruiser, Guide missile Destroyer, and Replenishment oiler(AOR) .
3. The number of lump=3: Freighter and Container

Second, the position of a lump relative the midships is measured. This quantity is scaled by the total length of the ship. This scaled quantity will be invariant with respect to the different ship sizes at different ranges

Third, the area of the lump is normalized to the ship length squared.

#### 1. Lump Detection

As shown in the plot of  $X$ ,  $Y$ ,  $C_x$ , and  $C_y$  vs  $Z$  parameter in Figure 4.15, when the difference ( $\Delta C_x$ ) between successive value of  $C_y$  varies from increasing to decreasing, and then, to increasing again,  $C_y$  exhibits noticable variation. From observation of  $C_x$  values, it is seen that the difference ( $\Delta C_x$ ) always has variation in the same sequence as stated above. Therefore, only  $C_y$  values are taken into consideration in the program that detects lump.

There are two distinct procedures used in dealing with two different kinds of lumps big or small. Thus, it is necessary to distinguish in the first place whether the lump is big or small. For the big lump, the differences of  $C_y$  is positive for all initial four knots. It continues to the peak and decreases toward the ending of the lump. For the small lump, the difference of  $C_y$  is positive for the first two knots and stay constant or positive for the third knot, but the difference of  $C_y$  for the forth knot is negative, thereafter, the program proceeds to establish the following values for each lump detected:

First, the knot positions ( $Z$  value) at

1. the beginning of the lump
2. the ending of the lump

Second, the Knot number for

1. the beginning of the lump
2. the ending of the lump

Third, Number of lumps detected.

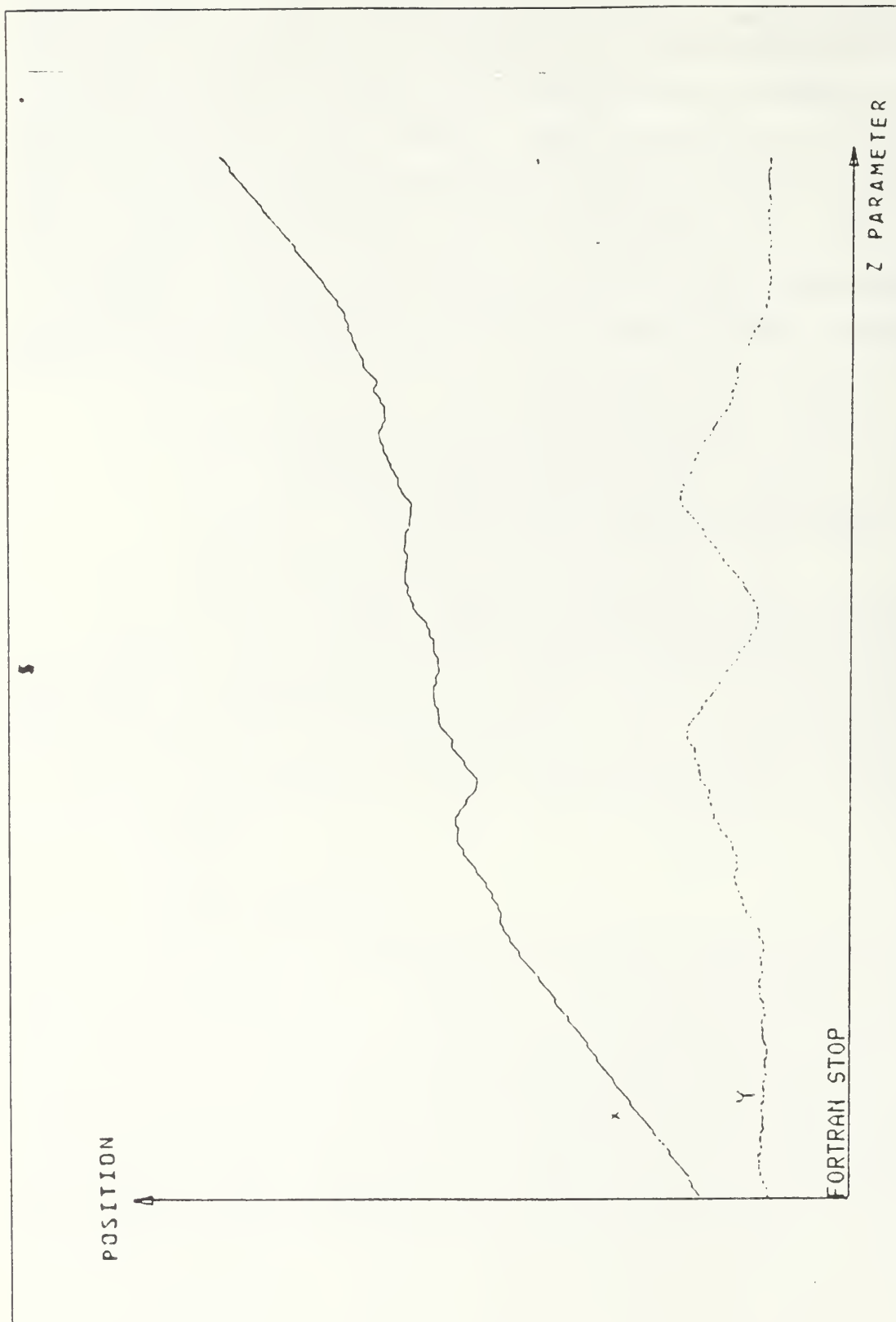


Figure 4.27 Plot X,Y vs Z for a CGN at 45kft.

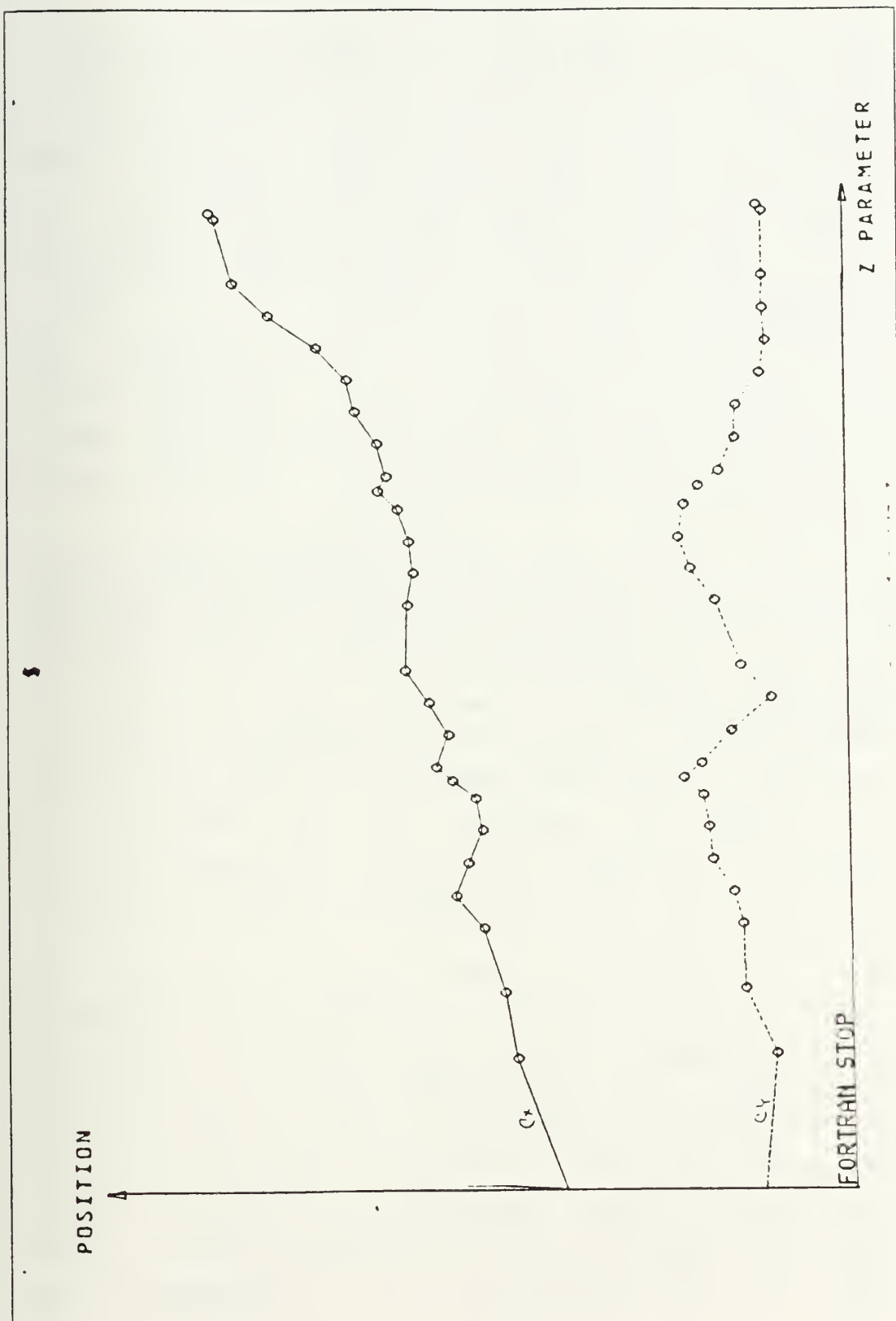


Figure 4.28 Plot  $C_x, C_y$  vs  $Z$  for a CGN at 45kft.

a. Procedure to Detect the Lump

1. Check a big or small lump by testing the conditions of three varying values of  $Cy$  increments ( $\Delta Cy$ ) in sequence. If they are positive it is a big lump and set the flag-lump to 1, otherwise it is a small lump and set the flag-lump to zero; then go to 2
2. Check the present knot position to see where its  $Z$  value equal to zero or to maximum  $Z$  value, if it is  $Z$  maximum then stop, it not go to 3
3. If the process begins at the first knot position, set the begin and the flag-end to zero; check the status of the flag-lump for 1 (big lump) or 0 (small lump), if it is a big lump, then go to 4. If it is a small lump, then go to 7.
4. Check the status for the beginning or ending of the lump. If the flag-begin is 1, it represents that the beginning of a lump is found, then go to 5. Otherwise it's not found, then go to 6.
5. Find the ending of the big lump by testing the conditions of 3 values of  $Cy$  increments in sequence. The first 2 should be negative and the third should be constant or positive. If the condition are satisfied, store the  $Z$  value of the position of the third knot, and the flag-begin to zero; then go to 10.
6. Find the beginning of the big lump by testing the conditions of 3 different values of  $Cy$  increment ( $\Delta Cy$ ) in sequence. The first  $Cy$  should be negative or constant, the second should be positive, and the third should be positive. If the conditions are satisfied, store the  $Z$  value of the position of the second knot, and set flag-begin to 1; then go to 10.
7. Check the status for the beginning or ending of the lump. If the flag-begin is 1, it represents the

beginning of a lump is found, then go to 8. Otherwise it's not found, then go to 9.

8. Find the ending of the small lump by testing the conditions of 3 values of  $Cy$  increment ( $\Delta Cy$ ) in sequence. The first one should be negative or constant. If the conditions are satisfied, store the  $Z$  value of the position of the second knot, and set the flag-begin to zero; then go to 10.
9. Find the beginning of the small lump by testing the conditions of 3 values of  $Cy$  increment ( $\Delta Cy$ ) in sequence. The first should be constant or negative, the second should be positive, the third should be constant. If the conditions are satisfied, store the position of the second knot, and set the flag-begin to 1; then go to 10.

10. Move to the next knot, then go to 2.

This procedure is shown in Figure 4.29 and the detail of each procedure is shown in Appendix D.



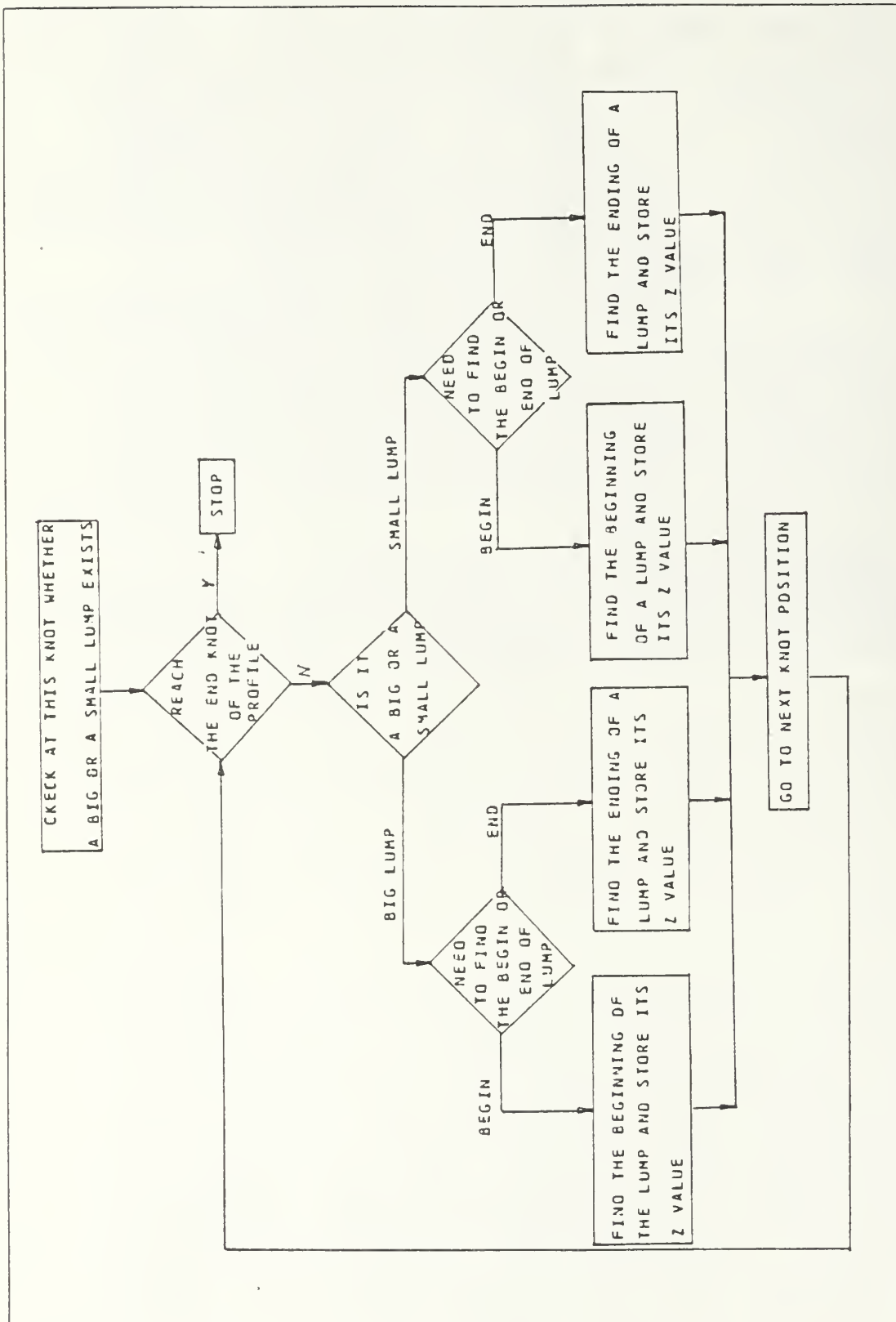


Figure 4.29 Flow Chart for Detecting Lumps.

## 2. To Determine the Area Under a Lump

The area under the lump can be determined by

$$\Delta \text{AREA} = \frac{\Delta C_x \Delta C_y}{2} + C_y \Delta C_y \quad (4.10)$$

Suppose there is a lump as shown in Figure 4.30. The area increment can be calculated by using eq(4.10). Then the area under BC(which is negative) is added to the area under AB.

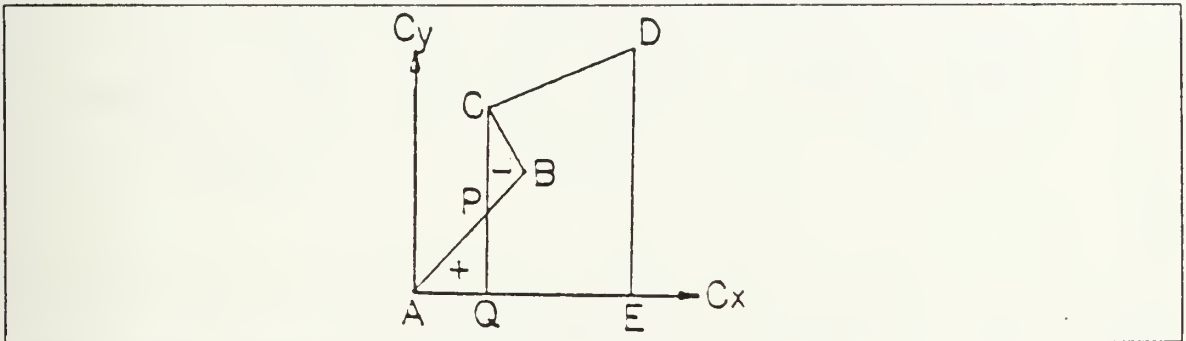


Figure 4.30 First Procedure to Determine the Area.

Next, area under CD is calculated, which is positive and add this to the last resulting area. Obviously, the sum would represent the total area of the lump as shown in Figure 4.31.

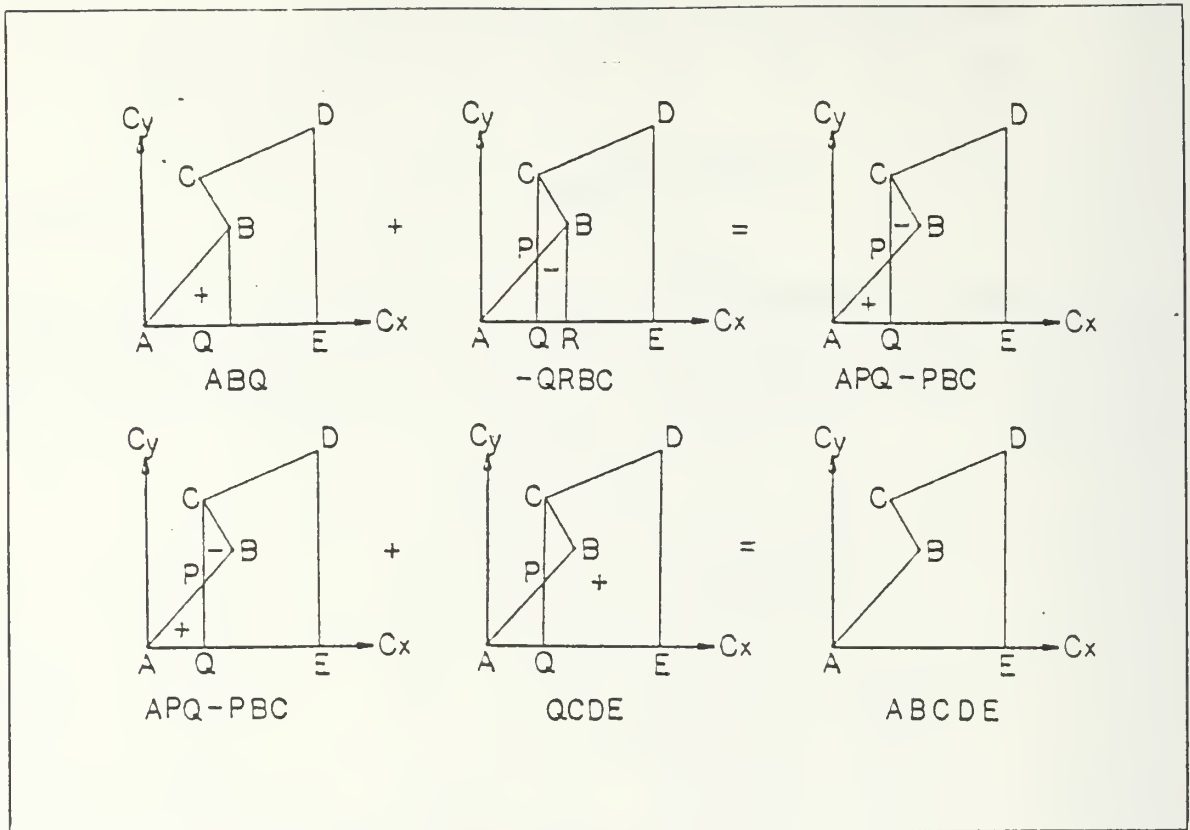


Figure 4.31 Step by Step Procedure to Determine the Area.

#### F. CONSTRUCTION OF THE DECISION TREE

The characteristics of ships used for classification are, the number of lumps, the area of lumps, the knot positions (Z value) and where the beginning of the lump and lump maxima relative to the midships are located.

In view of the above characteristic, it is necessary to find the relationships among them to make classification possible. Therefore, for each of the eight different class of ships, the decision tree is constructed which is based upon relationships observed in the plots of the knot position (Z values) for the beginning of the lumps normalized by the total ship length (Z value) VS. the number of lumps; the area of the lumps normalized by the total ship length

(Z value) squared vs the number of lumps; and the Z value of the lump maxima, as shown in Figure 4.32 through Figure 4.39.

Thus, according to the number of lumps presented in the profile used as the first criteria, ships can be divided into 3 distinct groups. Then, classification, can be made by comparing further characteristic as shown in Table III, and the complete decision tree constructed from Table III is shown in Figure 4.40.

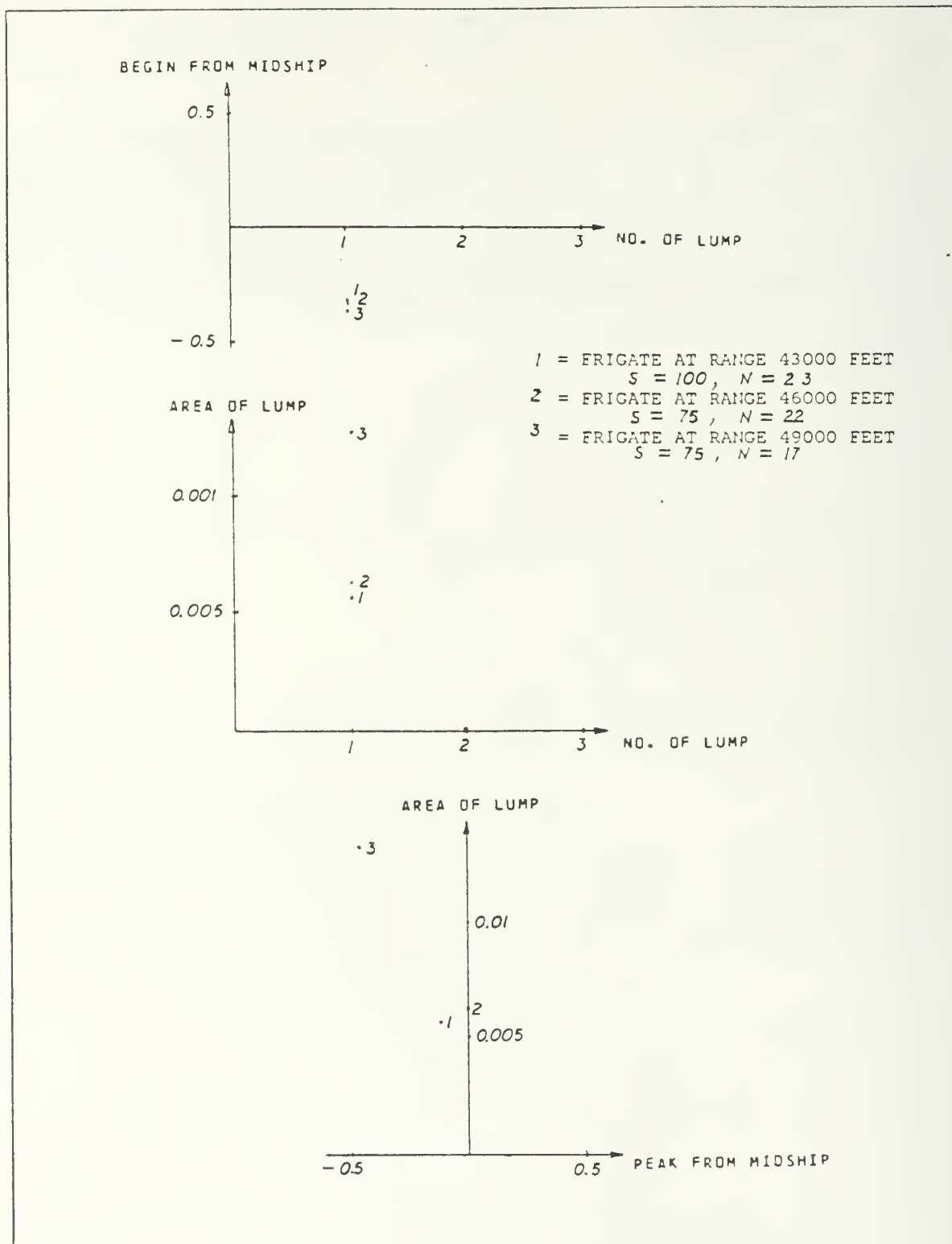
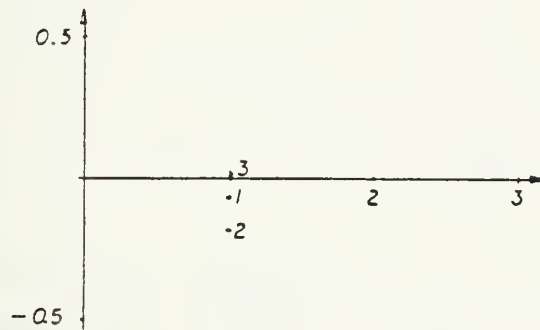


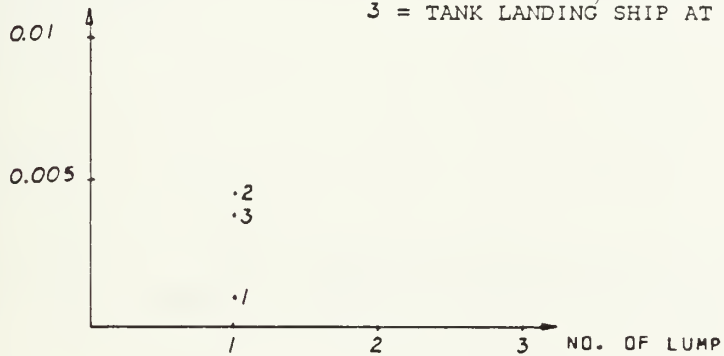
Figure 4.32 Plot the Beginning, Area, and Peak of a FF.

BEGIN FROM MIDSHIP



1 = TANK LANDING SHIP AT RANGE 51000 FEET  
 $S = 100$ ,  $N = 30$   
 2 = TANK LANDING SHIP AT RANGE 57000 FEET  
 $S = 80$ ,  $N = 23$   
 3 = TANK LANDING SHIP AT RANGE 62000 FEET

AREA OF LUMP



AREA OF LUMP

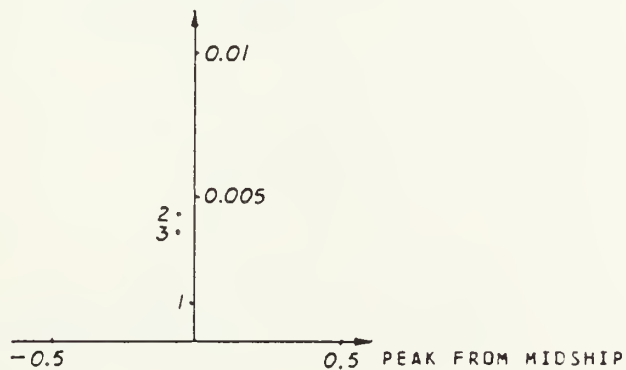
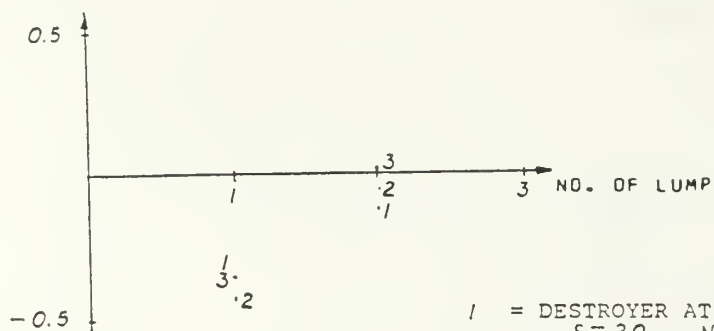


Figure 4.33 Plot the Beginning, Area, and Peak of a LST.

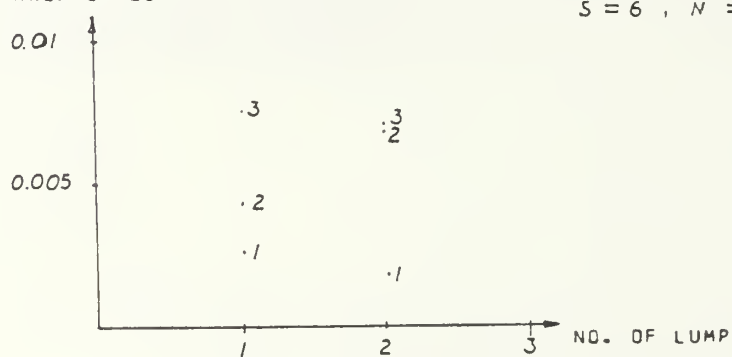


BEGIN FROM MIDSHIP



- 1 = DESTROYER AT RANGE 77000 FEET  
S = 20 , N = 28
- 2 = DESTROYER AT RANGE 79000 FEET  
S = 15 , N = 26
- 3 = DESTROYER AT RANGE 83000 FEET  
S = 6 , N = 44

AREA OF LUMP



AREA OF LUMP

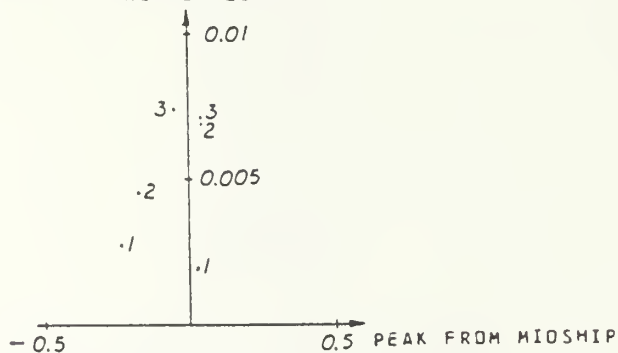
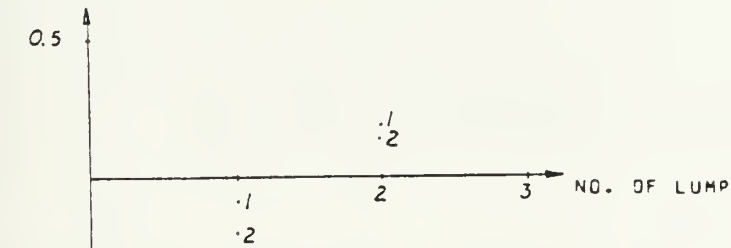


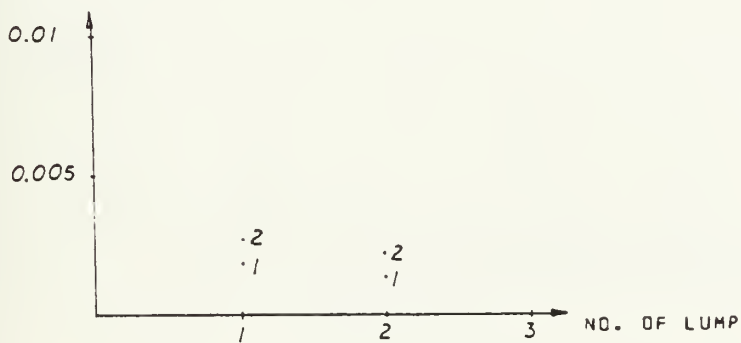
Figure 4.34 Plot the Beginning, Area, and Peak of a DD.

BEGIN FROM MIDSHIP



1 = GUIDED MISSILE DESTROYER AT RANGE 41000 FEET  
 $S = 125$ ,  $N = 35$   
 2 = GUIDED MISSILE DESTROYER AT RANGE 47000 FEET  
 $S = 100$ ,  $N = 27$

AREA OF LUMP



AREA OF LUMP

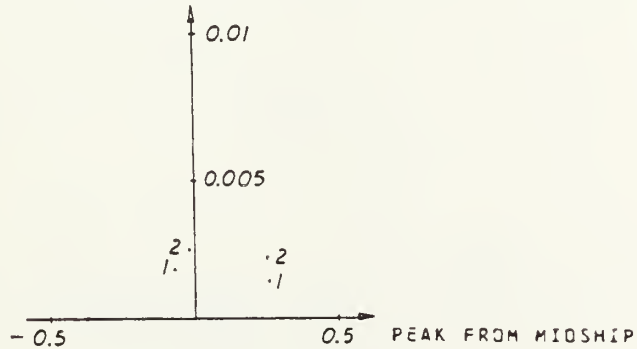
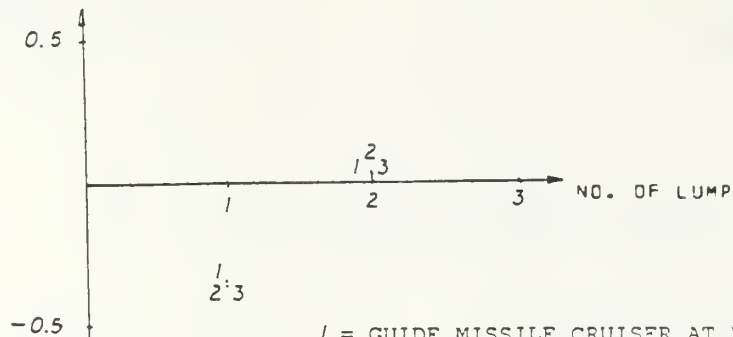


Figure 4.35 Plot the Beginning, Area, and Peak of a DDG.

BEGIN FROM MIDSHIP



1 = GUIDE MISSILE CRUISER AT RANGE 45000 FEET  
 $S = 125$  ,  $N = 33$   
 2 = GUIDE MISSILE CRUISER AT RANGE 55000 FEET  
 $S = 100$  ,  $N = 26$   
 3 = GUIDE MISSILE CRUISER AT RANGE 64000 FEET  
 $S = 50$  ,  $N = 26$

AREA OF LUMP

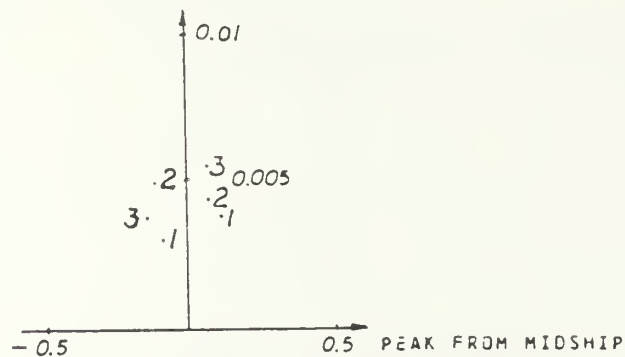
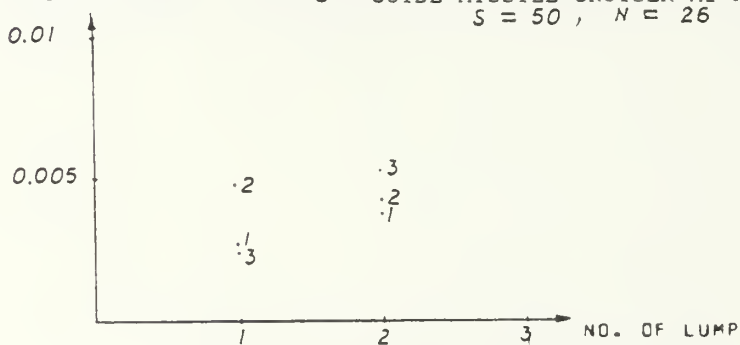
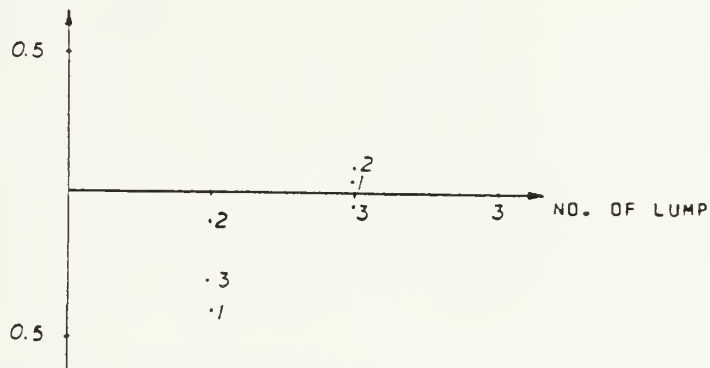
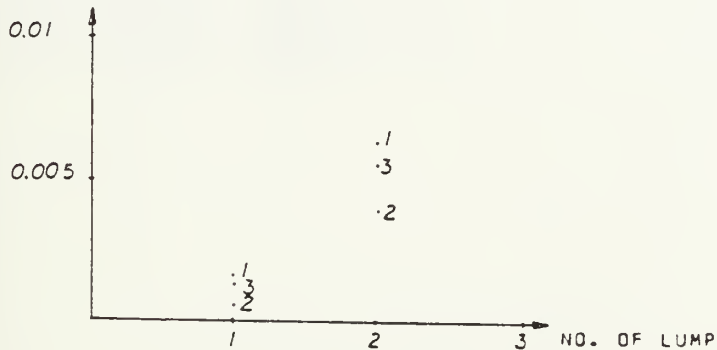


Figure 4.36 Plot the Beginning, Area, and Peak of a CGN.

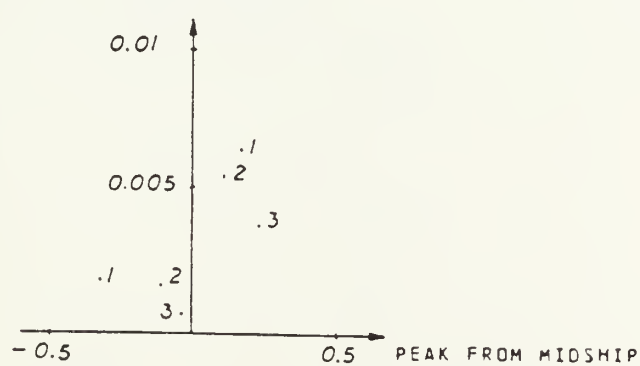
BEGIN FROM MIDSHIP



AREA OF LUMP



AREA OF LUMP



- 1 = REPLENISHMENT OILER AT RANGE 78000 FEET  
 $S = 20$ ,  $N = 36$
- 2 = REPLENISHMENT OILER AT RANGE 83000 FEET  
 $S = 10$ ,  $N = 33$
- 3 = REPLENISHMENT OILER AT RANGE 88000 FEET  
 $S = 10$ ,  $N = 36$

Figure 4.37 Plot the Beginning, Area, and Peak of a AOR.

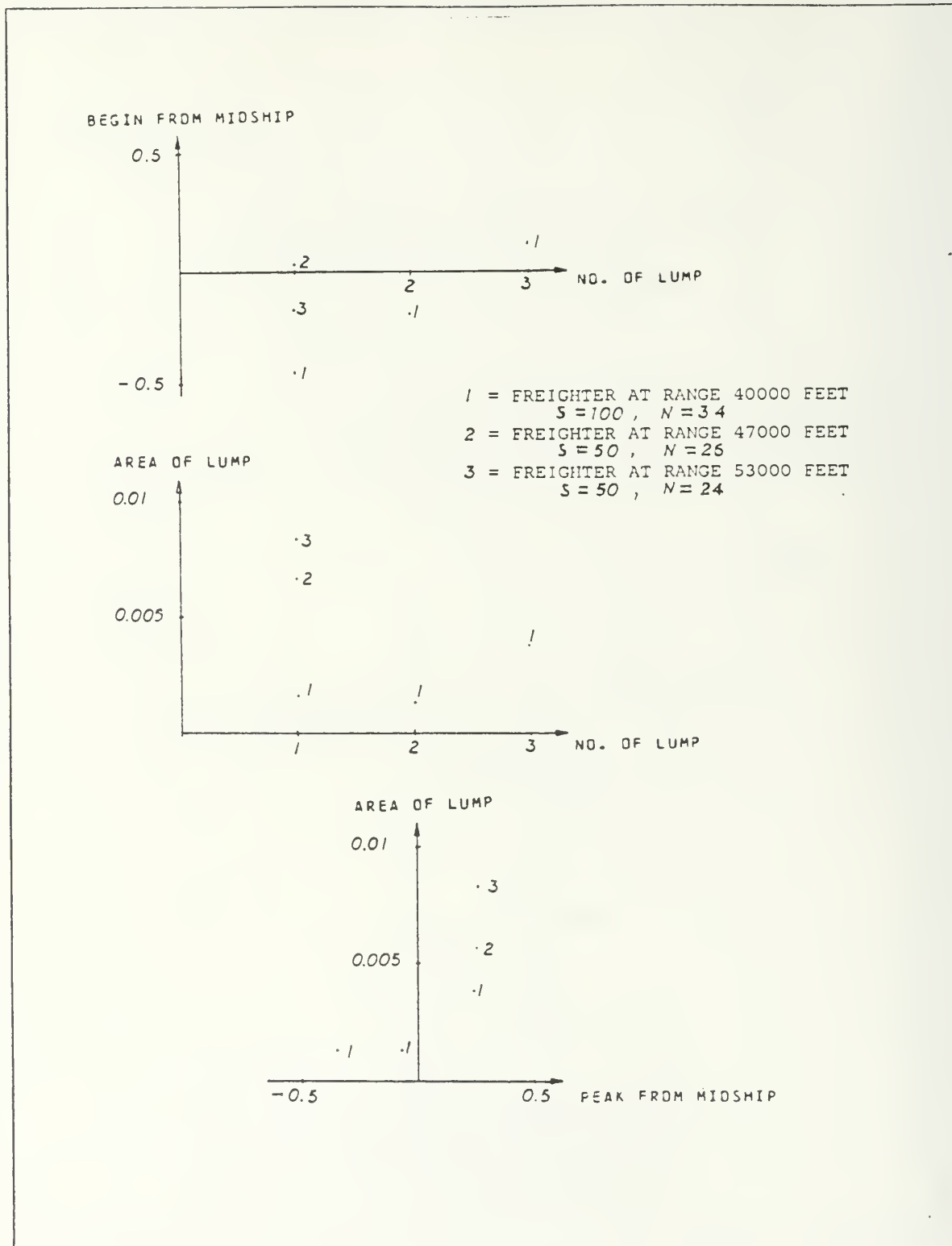
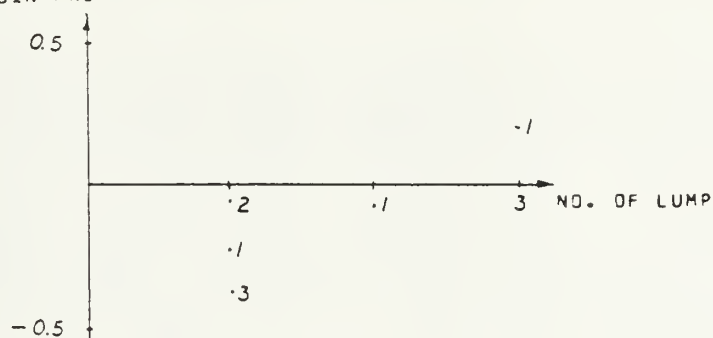
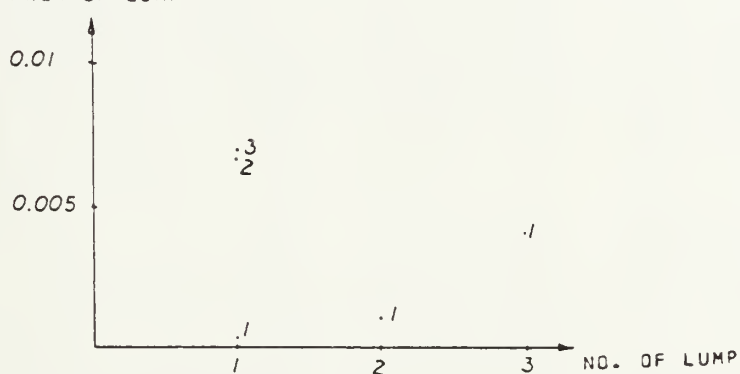


Figure 4.38 Plot the Beginning, Area, and Peak of a Freighter.

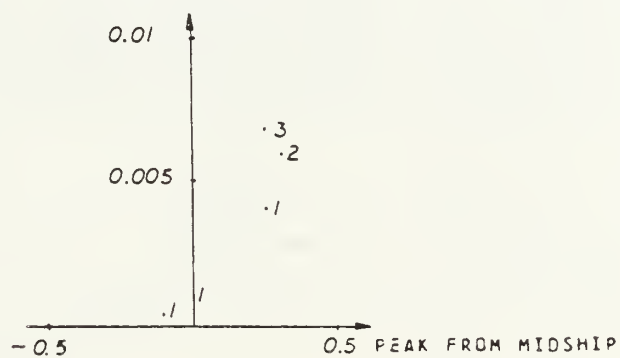
BEGIN FROM MIDSHIP



AREA OF LUMP



AREA OF LUMP



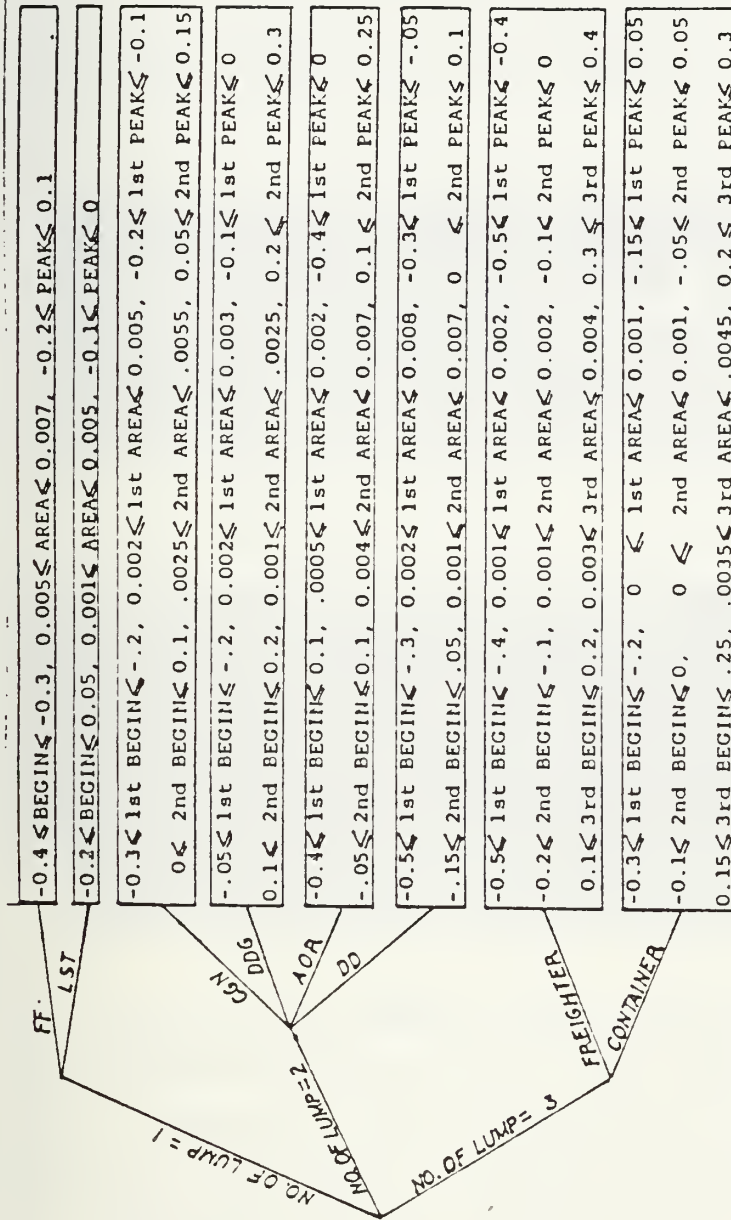
- 1 =CONTAINER AT RANGE 28000 FEET  
S=100, N=32
- 2 =CONTAINER AT RANGE 36000 FEET  
S=50, N=32
- 3 =CONTAINER AT RANGE 42000 FEET  
S=50, N=25

Figure 4.39 Plot the Beginning, Area, and Peak of a Container.



TABLE III  
Comparison of Different Types of Ships

No. OF LUMP	BEGINNING OF LUMP FROM MIDSHIP	PEAK IN THE LUMP	AREA UNDER THE LUMP	RESULT
1	-0.4 BEGIN -0.3	- 0.2 PEAK 0.1	0.005 AREA 0.007	FF
	-0.2 BEGIN 0.05	-0.1 PEAK 0	0.001 AREA 0.005	LST
2	-0.3 1st BEGIN -0.2 0 2nd BEGIN 0.1	-0.2 1st PEAK -0.1 0.05 2nd PEAK 0.15	0.002 1st AREA 0.005 .0025 2nd AREA 0.0055	CGN
	-0.2 1st BEGIN -0.05 0.1 2nd BEGIN 0.2	-0.1 1st PEAK 0 0.2 2nd PEAK 0.3	0.002 1st AREA 0.003 0.001 2nd AREA 0.0025	DDG
	-0.4 1st BEGIN 0.1 -0.5 2nd BEGIN 0.1	-0.4 1st PEAK 0 0.1 2nd PEAK 0.25	.0005 1st AREA 0.002 0.004 2nd AREA 0.007	AOR
	-0.5 1st BEGIN -0.3 -.15 2nd BEGIN 0.05	-0.3 1st PEAK 0.05 0 2nd PEAK 0.1	0.002 1st AREA 0.008 0.001 2nd AREA 0.007	DD
	-0.5 1st BEGIN -0.4 -0.2 2nd BEGIN -0.1 0.1 3rd BEGIN 0.2	-0.5 1st PEAK -0.4 -0.1 2nd PEAK 0 0.3 3rd PEAK 0.4	0.001 1st AREA 0.002 0.001 2nd AREA 0.002 0.003 3rd AREA 0.004	FREIGHTER
	-0.3 1st BEGIN -0.2 -0.1 2nd BEGIN 0 0.15 3rd BEGIN 0.25	-.15 1st PEAK -.05 -.05 2nd PEAK 0.05 0.2 3rd PEAK 0.3	0 1st AREA 0.001 0 2nd AREA 0.001 .0035 3rd AREA 0.0045	CONTAINER



BEGIN = THE BEGINNING OF THE LUMP FROM MIDSHIP NORMALIZE BY THE TOTAL

LENGTH OF THE SHIP

AREA = THE AREA OF THE LUMP NORMALIZE BY THE TOTAL LENGTH OF SHIP

SQUARED

PEAK = THE PEAK HEIGHT OF THE LUMP NORMALIZE BY TOTAL LENGTH OF SHIP

Figure 4.40 The Decision Tree.

## G. SUMMARY

The decision tree which is constructed from Table III does not provide 100-percent correct classification for all types of ship images. Furthermore, the presence of noise in images may cause complications in classifying the ships. For example, with excessive noise in the original image, Sobel operator for edge enhancement in the preprocessing process, still yield result with residual noise present in Figure 4.41. These residual noise is undesirable since it causes failure to extract profiles which retain necessary informations from the original images. The subsequent classification of profile by the Fourier Coefficient method and the B-spline Coefficient method becomes difficult.

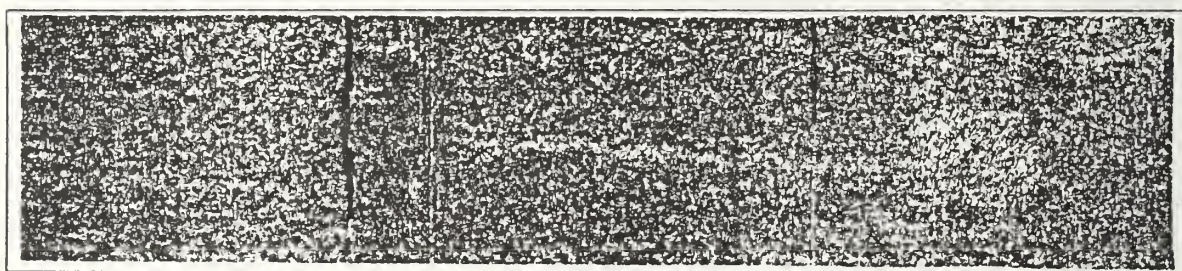


Figure 4.41 Noisy Image.

As discussed before in order to reduce the noise, an appropriate threshold gray value for the preprocessed image is set. This results in a silhouette image. However, if the value is too high the profile becomes broken which prevent successful operation in the closing process discussed in chapter 2, as shown in Figure 4.42. On the other hand decreasing the threshold value results in erroneous profile, as shown in Figure 4.43. In some cases, when the closing process is used, certain vital information is lost. This effect can be seen in comparing the image in Figure 4.44,



and Figure 4.45. Therefore, loss of informations due to the attempt of eliminating noise and the inability in the preprocessing process to cope with the residual noise, yield erroneous profile. Consequently, failure occurs in the succeeding classification steps.

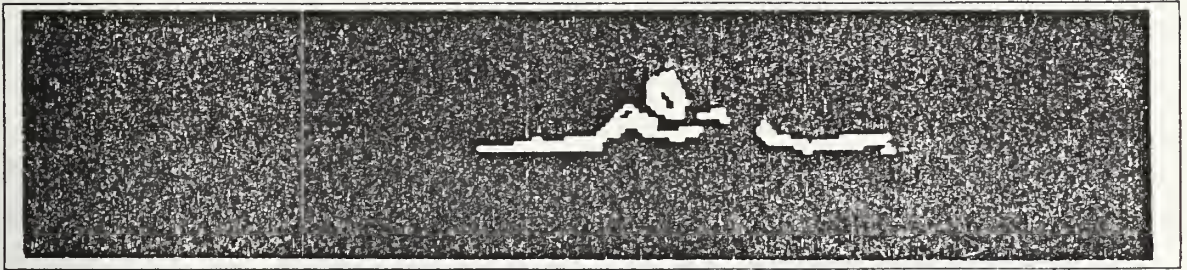


Figure 4.42 High Threshold.

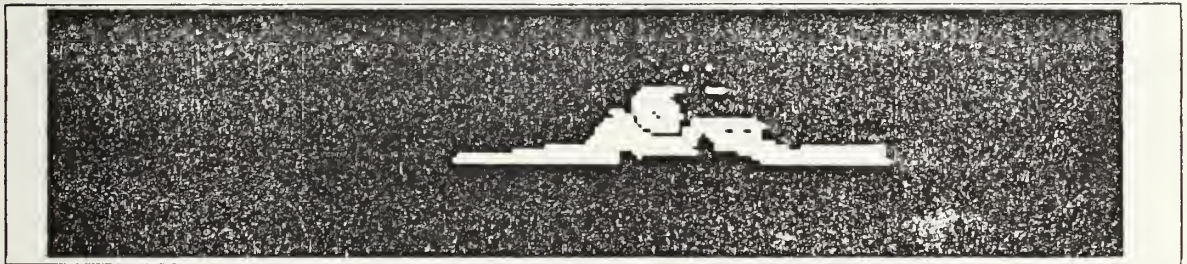


Figure 4.43 Low Threshold.

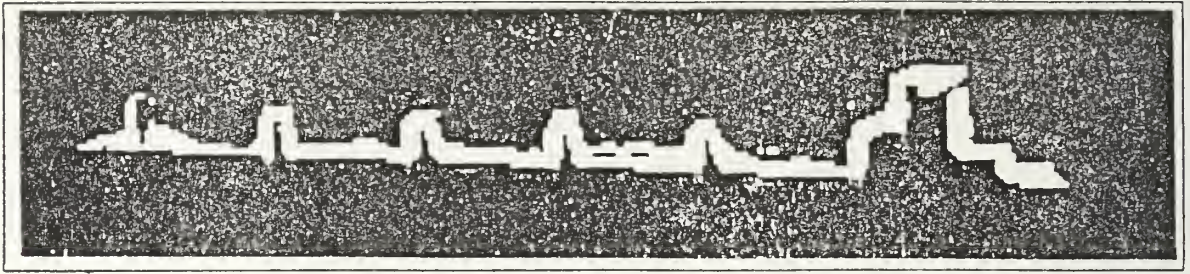


Figure 4.44 Before Closing.

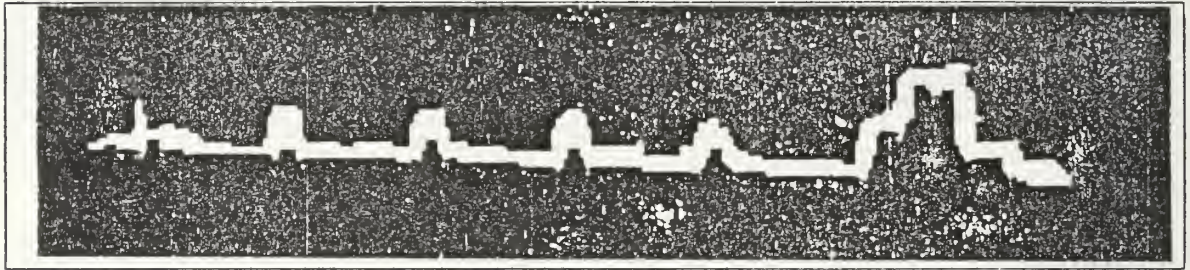


Figure 4.45 After Closing.

## V. CONCLUSION

The shape of the superstructure of a ship is the most important feature used in the classification algorithms. To extract the edge profile, a Sobel operator is employed. The limitation of a Sobel operator is that some small details of the edge is lost. For example, the image of a DD at a range of 79000 feet after applying the Sobel operator shows the superstructure and the radar. But the edge of a small mast disappeared as shown in Figure 5.1. Furthermore, if the threshold value is set too big, the edge image of the superstructure profile becomes broken. The top superstructure profile is obtained by setting the gray value under the slope between the bow and the stern to zero. We need to apply a contour tracking process to refine the superstructure profile. For some images, the connection of the broken profile pieces may be achieved in a Closing operation as discussed in Chapter 2. The disadvantage of the Closing operation is that some small details of the profile may disappear. The superstructure profile of a freighter at a range of 53000 feet is shown in Figure 5.2. After the Closing process the detail of the cranes disappeared as shown in Figure 5.3.

The ship classification can be achieved by either the Fourier Coefficient method or the B-spline Coefficient method. In using these coefficients to classify ships, it is found that only the initial coefficients that lie between the 0th to the 20th, are relevant, while the rest are not. Inspection of the comparison curves of the same class of ships shows that, similarities in patterns exists up to the 20-th point. Beyond that, diversities in shape are so great that inclusion of those additional points for classification



will be of no use. Examples are shown in Figure 5.4 and Figure 5.5.

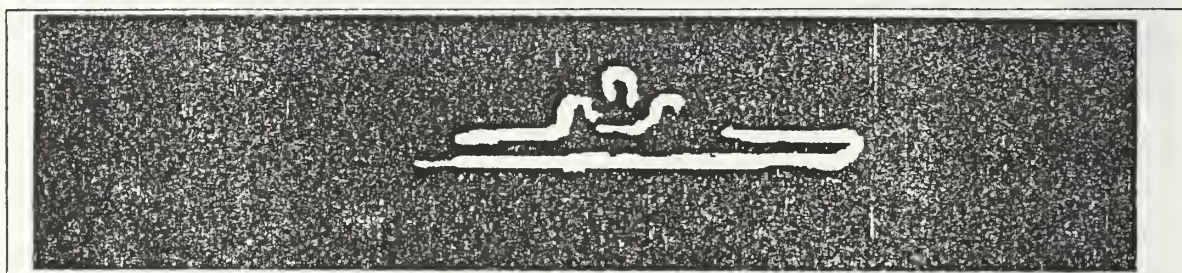


Figure 5.1 Edge Image of a DD at a Range of 79000 feet.

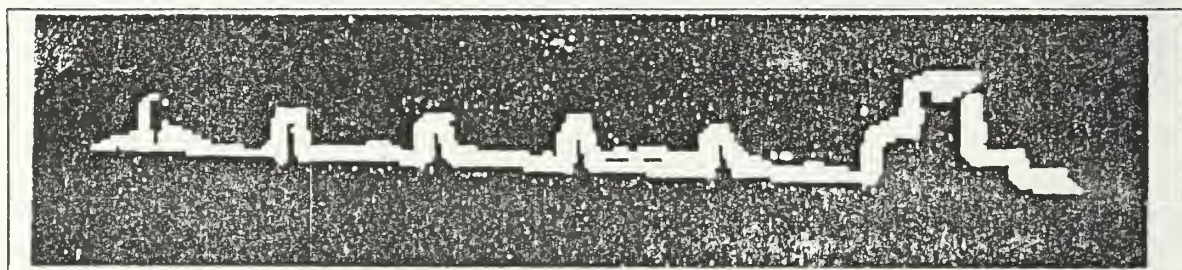


Figure 5.2 Before Closing Process.

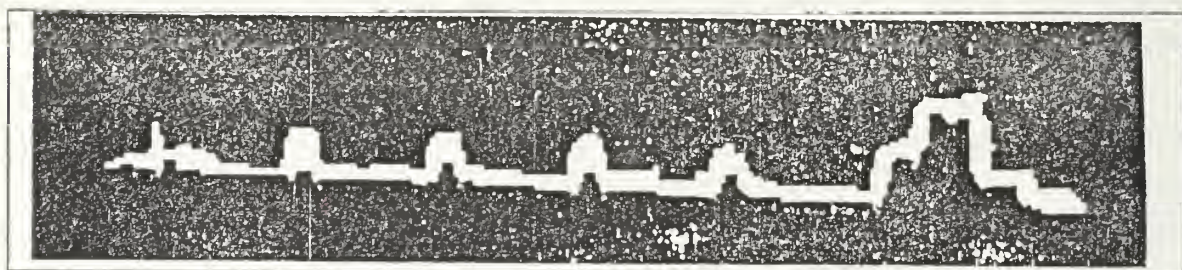


Figure 5.3 After Closing Process.

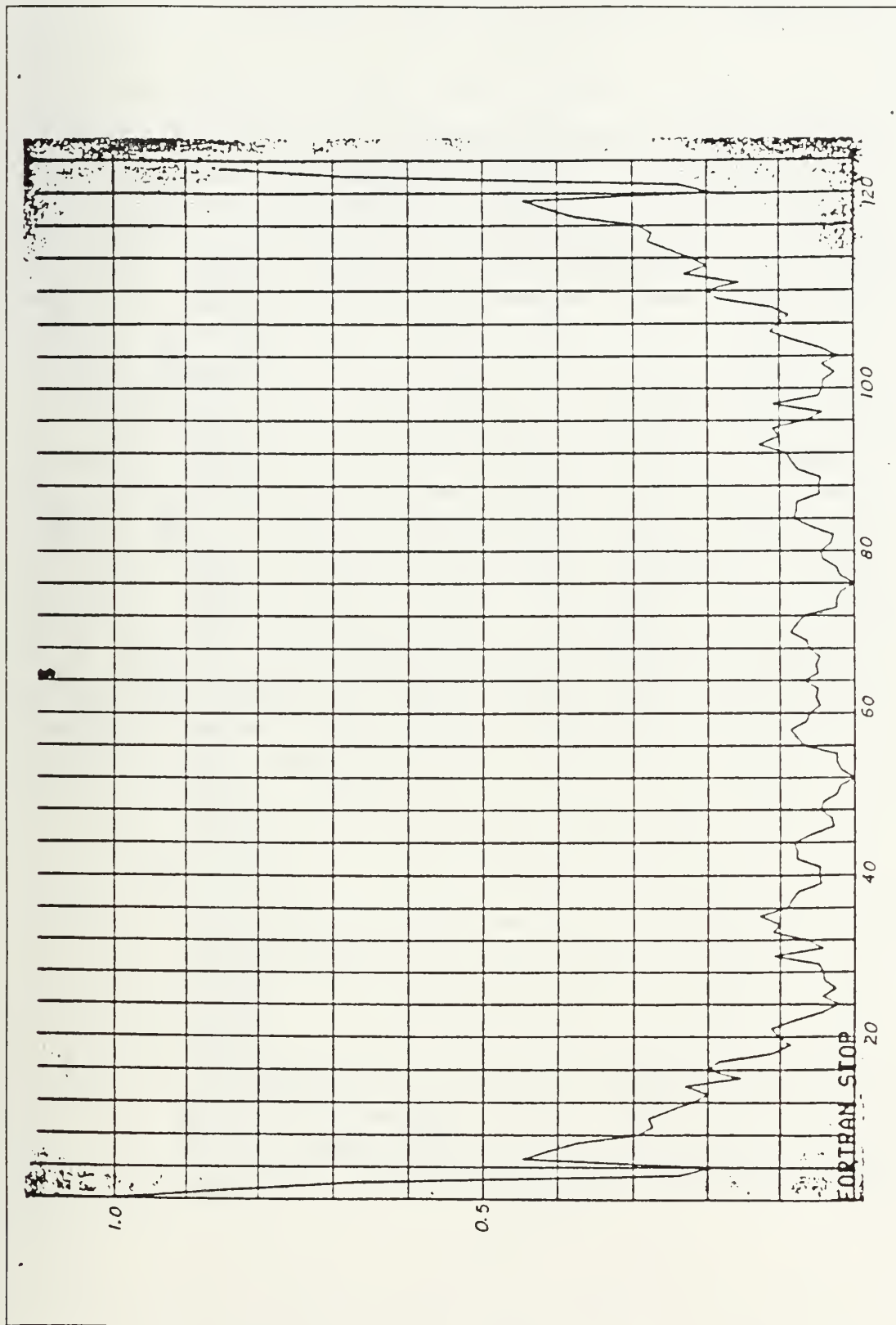


Figure 5.4 Logarithmic Magnitude of a CGN at a Range of 45000 feet.

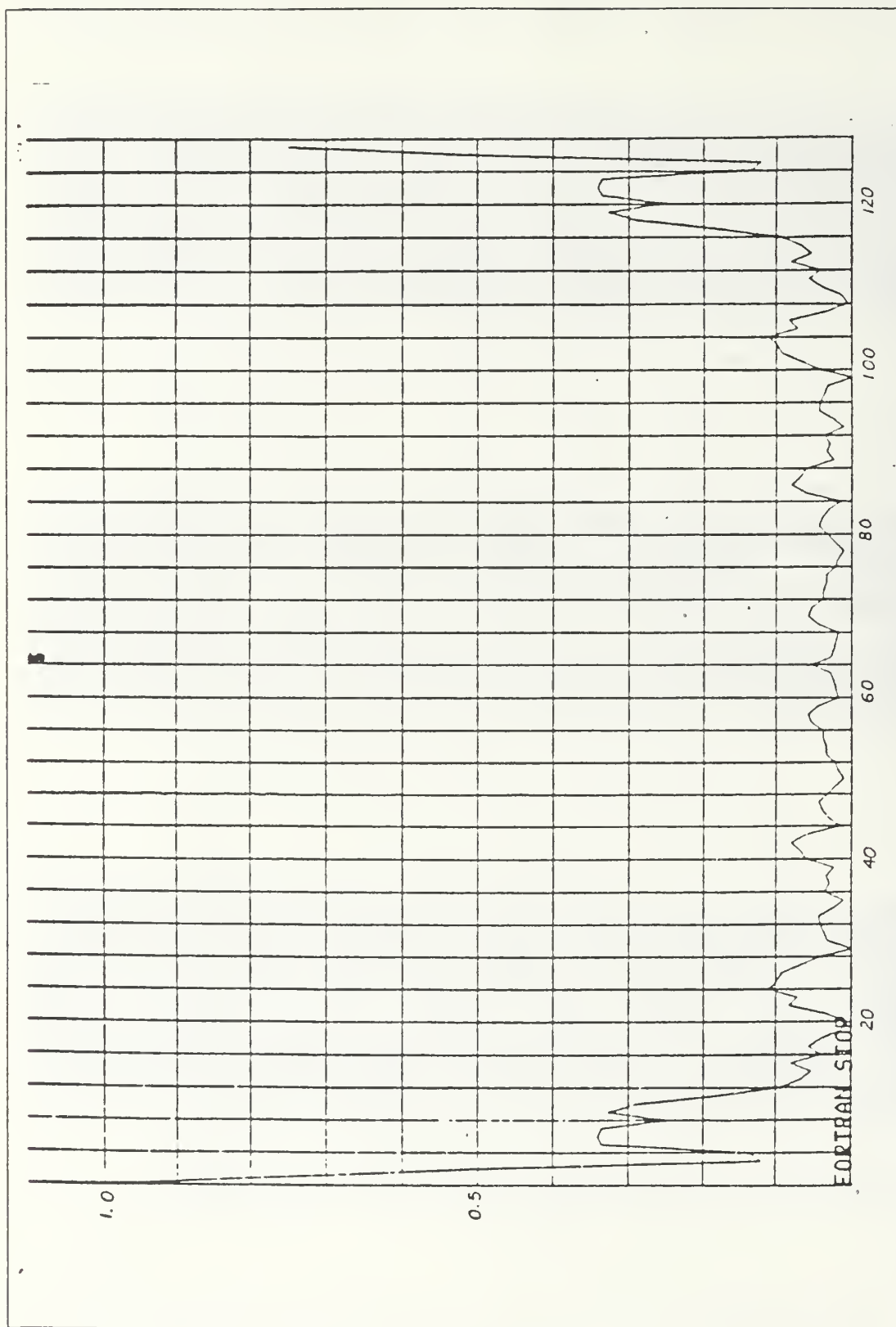


Figure 5.5 Logarithmic Magnitude of a CCN at a Range of 55000 feet.

In the B-spline coefficient method the technique of uneven knot selection has been employed. With the original sampling points on the ship profile as input, a smaller set of approximation samples are collected. These can be used to reconstruct the ship profile with sufficient information retained from the original image. The execution time as approached to that of handling the original data directly has been greatly reduced. The reduction of the number of sample points by almost a factor of 10 is common. For a CGN ship a set of original sampling points of 290, has been reduced to 36.

Comparing the two methods, Fourier Coefficient and B-spline Coefficient, the former method, in some cases is not effective to establish satisfactory classification of ships. This is due to difficulties in matching similarities of the shape of the coefficient curves. The latter, however, surpasses the former in that it is able to classify more ships accurately using computer programs. It is possible to improve the reliability of those two methods by reducing noise in the data collection process and the preprocessing process.

APPENDIX A

THE PROGRAM TO OBTAIN THE SUPERSTRUCTURE



## Source Listing

6-Dec-1984 17:18:40

VAX-

6-Dec-1984 17:18:31

\_DRA

```
program cut(input,output,infile,outfile);
```

```
type
```

```
byte = 0..255;  
imagerow1 = packed array [0..257] of byte;  
imagerow2 = packed array [0..255] of integer;  
imagerow3 = packed array [0..255] of byte;  
row1 = packed array [0..255] of byte;  
row2 = packed array [0..255] of integer;
```

```
var
```

```
sobel : array [0..63] of imagerow2;  
i,j : byte;  
f : array [0..65] of imagerow1;  
outfile : file of imagerow3;  
dx,dy,range,bright,n,m : integer;  
infile : file of row1;  
slope : real;  
image : array [0..63] of row1;  
x,x1,x2,y1,y2 : integer;  
NUM1,NUM2,NUM3,NUM4,max,min,max1 : integer;  
cal : array [0..63] of row1;  
thes : integer;  
NAME : PACKED ARRAY [1..20] OF CHAR;
```

```
BEGIN
```

```
WRITELN('INPUT SHIP FILENAME OUT CUT.DAT');  
READLN(NAME);  
WRITELN('THRESHOLD');  
READ(THES);  
WRITELN('NUM TO CUT FROM LEFT');  
READ(NUM1);  
WRITELN('NUM2 TO CUT FROM RIGHT');  
READ(NUM2);  
WRITELN('NUM3 TO CUT FROM TOP');  
READ(NUM3);  
WRITELN('NUM4 TO CUT FROM BOTTOM');  
READ(NUM4);  
open (infile,NAME,history :=old,  
      access_method :=sequential,  
      record_length :=256,record_type :=fixed);  
open (outfile,'CUT.dat',history :=new,record_length :=256,  
      record_type :=fixed);  
reset(infile);  
rewrite (outfile);  
i:=0;  
while not eof (infile) do  
begin  
  read (infile,image[i]);  
  for j:=0 to 255 do  
    f[i+1,j+1] := image[i,j];  
  i:=i+1;  
end;
```

```
(*compute sobel*)
```



## Source Listing

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\_DRA

```
FOR J:=0 TO 7 DO
  F[64,J+248]:=F[63,J+248];
for i := 0 to 63 do begin
  f[i+1,0] := f[i+1,1];
  f[i+1,257] := f[i+1,256];
end;
for j:= 0 to 255 do begin
  f[0,j+1] := f[1,j+1];
  f[65,j+1] := f[64,j+1];
end;
f[0,0]:=f[1,1];
f[0,257]:=f[1,256];
f[65,0]:=f[64,1];
f[65,257]:=f[64,256];

(*set initial max, min *)

max:=0;
max1:=0;
min:=255;
for i:= 0 to 63 do begin
  for j:= 0 to 255 do begin
    dx:=f[i,j]+2*f[i+1,j]+f[i+2,j]-f[i,j+2]
      -2*f[i+1,j+2]-f[i+2,j+2];
    dy:=f[i+2,j]+2*f[i+2,j+1]+f[i+2,j+2]-f[i,j]
      -2*f[i,j+1]-f[i,j+2];
    sobel[i,j]:=round((dx**2+dy**2)**0.5);
    if max < sobel[i,j] then
      max := sobel[i,j];
    if min > sobel[i,j] then
      min := sobel[i,j];
  end;
end;
range := max-min;

(* rescale*)

for i:= 0 to 63 do begin
  for j:= 0 to 255 do begin
    cal[i,j] := round(((sobel[i,j]-min)*255)/range);
    if (j<=NUM1) or (j)=255-NUM2) then
      cal[i,j] :=0;
    if (i<=NUM3) or (i)=63-NUM4) then
      cal[i,j] := 0;
  end;
end;

(* cut theshold*)
IF THES<>0 THEN BEGIN
  if cal[i,j] <=thes
  then
    cal[i,j] := 0
  else
    cal[i,j] := 255;
END;
end;
end;

(*find point at raw and aft*)
```

## Source Listing

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\_DRA

```
m := 0;
n := 0;
bright:= 255;
for j := 0 to 150 do begin
  for i:=0 to 63 do begin
    if(cal[i,j]=bright) and (n=0) then
      begin
        y1 := j;
        x1 := i;
        n := n+1;
      end; (*if*)
    if(cal[i,255-j]=bright) and (m=0) then
      begin
        y2 := 255-j;
        x2 := i;
        m := m+1;
      end; (*if*)
    end;(*for*)
  writeln("x1=",x1,"y1=",y1,"x2=",x2,"y2=",y2);
(* cut line*)

slope :=1.0;
if x1=x2 then
begin
  slope := 0;
  for i:=x1+1 to 63 do begin
    for j:=y1-5 to y2+5 do
      cal[i,j] :=0;
    end;(*for*)
  end;(*if*)
  if slope<>0 then
  begin
    slope :=(x2-x1)/(y2-y1);
    if slope < 0 then begin
      slope := abs(slope);
      for j := y1 to y2 do begin
        x := x1+1-round((j-y1)*slope);
        for i :=x to 63 do
          cal[i,j]:=0;
        end;(*for*)
      end;(*if*)
      if slope >0 then begin
        for j:= y1 to y2 do begin
          x := x1+1+round((j-y1)*slope);
          for i :=x to 63 do
            cal[i,j] :=0;
          end;(*for*)
        end;(*if*)
        writeln("step2");
      end;(*if*)
    end;(*if*)
  end;(*if*)

(*put outfile*)

  for i:=0 to 63 do begin
```

Source Listing

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VAX-

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\_OR1

```
    for j :=0 to 255 do
      outfile^[j] := cal[i,j];
    put (outfile);
  end;(*for*)
  close(infile);
end.
```

APPENDIX B

THE PROGRAM OF CONTOUR FOLLOWING

## Source Listing

10-Dec-1984 13:53:33  
6-Dec-1984 17:38:13VAX-11  
\_DRA0:

PROGRAM TRACE(INPUT,OUTPUT,INFILE,OUTFILE):

TYPE

BYTE = 0..255;  
IMAGEROW1 = PACKED ARRAY [0..255] OF BYTE;  
ROW1 = PACKED ARRAY [0..257] OF BYTE;

VAR

R,C,F0,F1,F2,F3,F4,F5,F6,F7,F8 : BYTE;  
F : ARRAY [0..65] OF ROW1;  
A,IMAGE : ARRAY [0..63] OF IMAGEROW1;  
INFILE : FILE OF IMAGEROW1;  
R1,C1,R2,C2,CDIR,I,J,COUNT : INTEGER;  
ROW,CDL : ARRAY [0..512] OF INTEGER;  
OUTFILE : FILE OF IMAGEROW1;  
NAME : PACKED ARRAY [1..20] OF CHAR;

PROCEDURE STORE;

BEGIN

COL[I]:=C-1;  
ROW[I]:=R-1;  
WRITELN("COL=",COL[I],"ROW=",ROW[I]);  
COUNT :=I;  
I:=I+1;  
END;

PROCEDURE CMOVE;

BEGIN

F0:=F[R-1,C]; F1:=F[R-1,C+1]; F2:=F[R,C+1]; F3:=F[R+1,C+1];  
F4:=F[R+1,C]; F5:=F[R+1,C-1]; F6:=F[R,C-1]; F7:=F[R-1,C-1];  
CASE CDIR OF

0: BEGIN

IF (F3=0) AND (F2=255) THEN BEGIN C:=C+1; CDIR:=1; END  
ELSE IF (F2=0) AND (F1=255) THEN BEGIN R:=R-1;  
STORE: C:=C+1; CDIR:=0; END  
ELSE IF (F1=0) AND (F0=255) THEN BEGIN R:=R-1;  
CDIR:=0; END  
ELSE IF (F0=0) AND (F7=255) THEN BEGIN C:=C-1;  
STORE: R:=R-1; CDIR:=3; END  
ELSE IF (F7=0) AND (F6=255) THEN BEGIN C:=C-1;  
CDIR:=3; END  
ELSE IF (F6=0) AND (F5=255) THEN BEGIN R:=R+1;  
STORE: C:=C-1; CDIR:=3; END  
ELSE BEGIN R:=R+1; CDIR:=2; END;

END;

1: BEGIN

IF (F5=0) AND (F4=255) THEN BEGIN R:=R+1;  
CDIR:=2; END  
ELSE IF (F4=0) AND (F3=255) THEN BEGIN C:=C+1;  
STORE: R:=R+1; CDIR:=1; END  
ELSE IF (F3=0) AND (F2=255) THEN BEGIN C:=C+1;  
CDIR:=1; END  
ELSE IF (F2=0) AND (F1=255) THEN BEGIN R:=R-1;  
STORE: C:=C+1; CDIR:=0; END  
ELSE IF (F1=0) AND (F0=255) THEN BEGIN R:=R-1;  
CDIR:=0; END

Source Listing

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6-Dec-1984 17:38:13

VAX-11  
\_DRA0:

```

ELSE IF (F0=0) AND (F7=255) THEN BEGIN C:=C-1;
  STORE; R:=R-1; CDIR:=0; END
ELSE BEGIN C:=C-1; CDIR:=3; END;
END;
2: BEGIN
  IF (F7=0) AND (F6=255) THEN BEGIN C:=C-1;
    CDIR:=3; END
  ELSE IF (F6=0) AND (F5=255) THEN BEGIN R:=R+1;
    STORE; C:=C-1; CDIR:=3; END
  ELSE IF (F5=0) AND (F4=255) THEN BEGIN R:=R+1;
    CDIR:=2; END
  ELSE IF (F4=0) AND (F3=255) THEN BEGIN C:=C+1;
    STORE; R:=R+1; CDIR:=1; END
  ELSE IF (F3=0) AND (F2=255) THEN BEGIN C:=C+1;
    CDIR:=1; END
  ELSE IF (F2=0) AND (F1=255) THEN BEGIN R:=R-1;
    STORE; C:=C+1; CDIR:=1; END
  ELSE BEGIN R:=R-1; CDIR:=0; END;
END;
3: BEGIN
  IF (F1=0) AND (F0=255) THEN BEGIN R:=R-1;
    CDIR:=0; END
  ELSE IF (F0=0) AND (F7=255) THEN BEGIN C:=C-1;
    STORE; R:=R-1; CDIR:=0; END
  ELSE IF (F7=0) AND (F6=255) THEN BEGIN C:=C-1;
    CDIR:=3; END
  ELSE IF (F6=0) AND (F5=255) THEN BEGIN R:=R+1;
    STORE; C:=C-1; CDIR:=2; END
  ELSE IF (F5=0) AND (F4=255) THEN BEGIN R:=R+1;
    CDIR:=2; END
  ELSE IF (F4=0) AND (F3=255) THEN BEGIN C:=C+1;
    STORE; R:=R+1; CDIR:=2; END
  ELSE BEGIN C:=C+1; CDIR:=1; END;
END;
END;
END;

```

PROCEDURE INITIAL;

```

BEGIN
  FOR I :=0 TO 255 DO BEGIN
    COL[I] :=0;
    ROW[I] :=0;
  END;
  FOR J:=0 TO 63 DO BEGIN
    FOR I:=0 TO 255 DO
      A[R,C]:=0;
    END;
  END;
  I:=0;
END;

```

PROCEDURE FIRST;

```

VAR
  N,M :INTEGER;
BEGIN

```



## Source Listing

10-Dec-1984 13:53:33

VAX-11

6-Dec-1984 17:38:13

\_DRA0:

```
N := 0; M:=0;
FOR C :=0 TO 200 DO BEGIN
  FOR R :=0 TO 63 DO BEGIN
    IF (FCR,C)=255) AND (N=0) THEN BEGIN
      C1 := C; R1 :=R; N := N+1;
    END;
    IF (FCR,255-C)=255) AND (M=0) THEN BEGIN
      C2 := 255-C; R2 := R; M :=M+1;
    END;
  END;
END;
WRITELN("COL1=",C1,"ROW1=",R1,"COL2=",C2,"ROW2=",R2);
END;

(*****
(* MAIN PROGRAM *)
BEGIN
  WRITELN("INPUT CUT OR CONLINE FILENAME");
  READLN(NAME);
  OPEN (INFILE,NAME,HISTORY := OLD,
    ACCESS_METHOD :=SEQUENTIAL,
    RECORD_LENGTH :=256,RECORD_TYPE :=FIXED);
  OPEN (OUTFILE,"TR.0AT",HISTORY :=NEW,RECORD_LENGTH :=256,
    RECORD_TYPE :=FIXED);
  RESET(INFILE);
  REWRITE (OUTFILE);
  R :=0;
  WHILE NOT EOF (INFILE) DO
  BEGIN
    READ (INFILE,IMAGE[R]);
    FOR C := 0 TO 255 DO
      FCR+1,C+1] := IMAGE[R,C];
    R := R+1;
  END;
  FIRST;
  R :=R1; C :=C1; COIR :=1;
  INITIAL;
  WHILE (C<>C2) DO BEGIN
    STORE;
    CMQVE;
  END;
  FOR I:=0 TO COUNT DO
    A[ROW[I],COL[I]] :=255;
  FOR R:=0 TO 63 DO BEGIN
    FOR C:=0 TO 255 DO
      OUTFILE^[C] := A[R,C];
    PUT(OUTFILE);
  END;
  WRITELN("NUMBER=",COUNT);
  CLOSE(INFILE);
END.
```

APPENDIX C  
THE SUBROUTINE "PARAM"

```

      SUBROUTINE PARAM(X,Y,Z,W,M,ZB,ZE,K,S,N,T,CX,CY,FP,IOPT,IPAR,IER) 0930
C   GIVEN THE SET OF DATA POINTS (X(I),Y(I)) WITH CORRESPONDING Z- 0940
C   VALUES Z(I), I=1,2,...,M AND GIVEN ALSO THE SET OF POSITIVE 0950
C   NUMBERS W(I), I=1,2,...,M, SUBROUTINE PARAM FINDS A SMOOTH APPROXIMAT- 0960
C   ING CURVE WITH PARAMETER REPRESENTATION  $X = SX(Z)$ ,  $Y = SY(Z)$ . 0970
C    $SX(Z)$  AND  $SY(Z)$  ARE TWO SPLINE FUNCTIONS OF DEGREE K WITH THE NUMBER 0980
C   AND THE POSITION OF THE KNOTS  $T(J)$ ,  $J=1,2,...,N$  AUTOMATICALLY 0990
C   CHOSEN BY THE ROUTINE. THE SMOOTHNESS OF  $SX(Z)$  AND  $SY(Z)$  IS 1000
C   ACHIEVED BY MINIMALIZING THE  $SUM(DX(R)**2 + OY(R)**2)$  WHERE  $DX(R)$  1010
C   AND  $OY(R)$  STAND FOR THE DISCONTINUITY JUMP OF THE KTH DERIVATIVE 1020
C   OF  $SX(Z)$  AND  $SY(Z)$  AT THE KNOT  $T(R)$ ,  $R=K+2,...,N-K-1$ . 1030
C   THE AMOUNT OF SMOOTHNESS IS DETERMINED BY THE CONDITION THAT  $F(P) =$  1040
C    $SUM(W(I)*((X(I)-SX(Z(I)))**2 + (Y(I)-SY(Z(I)))**2))$  BE  $\leq S$ , WITH 1050
C   S A GIVEN NON-NEGATIVE CONSTANT. 1060
C   THE SPLINE FUNCTIONS  $SX(Z)$  AND  $SY(Z)$  ARE GIVEN IN THEIR B-SPLINE 1070
C   REPRESENTATION (B-SPLINE COEFFICIENTS  $CX(J)$ , RESP.  $CY(J)$ ,  $J=1,...,N-K-1$ ) 1080
C   AND CAN BE EVALUATED BY MEANS OF FUNCTION DERIV. 1090
C   CALLING SEQUENCE: 1100
C       CALL PARAM(X,Y,Z,W,M,ZB,ZE,K,S,N,T,CX,CY,FP,IOPT,IPAR,IER) 1110
C 1120
C   INPUT PARAMETERS: 1130
C       X   : ARRAY, LENGTH M, CONTAINING THE ABSCISSAE OF THE DATA POINTS 1140
C       Y   : ARRAY, LENGTH M, CONTAINING THE ORDINATES OF THE DATA POINTS 1150
C       W   : ARRAY, MINIMUM LENGTH M, CONTAINING THE WEIGHTS W(I). 1160
C       M   : INTEGER VALUE, CONTAINING THE NUMBER OF DATA POINTS. 1170
C       K   : INTEGER VALUE, CONTAINING THE DEGREE OF  $SX(Z)$  AND  $SY(Z)$ . 1180
C       S   : REAL VALUE, CONTAINING THE SMOOTHING FACTOR. 1190
C       IOPT : INTEGER VALUE WHICH TAKES THE VALUE 0 OR 1. 1200
C           IOPT=0: THE ROUTINE WILL RESTART ALL COMPUTATIONS. 1210
C           IOPT=1: THE ROUTINE WILL START WITH THE KNOTS FOUND AT THE 1220
C                   LAST CALL OF THE ROUTINE. IF IOPT=1 THE OUTPUT 1230
C                   PARAMETERS T AND N ARE INPUT PARAMETERS AS WELL. 1240
C                   IF IOPT=1 THE USER MUST PROVIDE WITH A COMMON BLOCK 1250
C                   COMMON/OPT1/NRODATA(NEST),FP0,FPO0,NPLUS 1260
C       IPAR : INTEGER FLAG. 1270
C           IPAR = 0: FOR EACH DATA POINT (X(I),Y(I)) THE PROGRAM AUTOMATICALLY 1280
C                   CHOOSES A CORRESPONDING VALUE OF THE PARAMETER Z, I.E. 1290
C                    $Z(1)=0; Z(I)=Z(I-1)+SQRT((X(I)-X(I-1))**2 + (Y(I)-Y(I-1))**2)$  1300
C                   THE BOUNDARIES FOR THE PARAMETER Z ARE CHOSEN AS FOLLOWS 1310
C                    $ZB = Z(1); ZE = Z(M)$ . 1320
C           IPAR = 1: THE USER HIMSELF PROVIDES WITH THE VALUES OF THE 1330
C                   PARAMETER Z AND WITH THE BOUNDARIES ZB AND ZE. 1340
C       Z   : ARRAY, LENGTH M, CONTAINING THE VALUES OF THE PARAMETER Z 1350
C           (IPAR = 1) 1360
C       ZB,ZE: REAL VALUES, CONTAINING THE BOUNDARIES OF THE PARAMETER Z 1370
C           (IPAR = 1). 1380
C 1390
C   OUTPUT PARAMETERS: 1400
C       T   : ARRAY, LENGTH NEST (SEE DATA INITIALIZATION STATEMENT), 1410
C           WHICH CONTAINS THE POSITION OF THE KNOTS, I.E. THE POSITION 1420
C           OF THE INTERIOR KNOTS  $T(K+2),...,T(N-K-1)$ , AS WELL AS THE 1430
C           POSITION OF THE KNOTS  $T(1)=T(2)=...=T(K+1)=ZB$  AND  $ZE =$  1440
C            $T(N-K)=...=T(N)$  WHICH ARE NEEDED FOR THE B-SPLINE REPRESENT. 1450
C       CX,CY: ARRAYS, LENGTH NEST, CONTAINING THE B-SPLINE COEFFICIENTS 1460
C           OF  $SX(Z)$ , RESP.  $SY(Z)$ . 1470
C       N   : INTEGER VALUE, CONTAINING THE TOTAL NUMBER OF KNOTS. 1480
C       FP  : REAL VALUE, WHICH CONTAINS THE  $SUM(WI*(XI-SX(ZI))**2)$  1490
C           +  $SUM(WI*(YI-SY(ZI))**2)$ ,  $I=1,2,...,M$ . 1500
C       IER : ERROR CODE 1510
C           IER=0: NORMAL RETURN. 1520
C           IER=-1: NORMAL RETURN,  $SX(Z)$  AND  $SY(Z)$  ARE INTERPOLATING SPLINES 1530

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C      IER=-2:NORMAL RETURN, SX(Z) AND SY(Z) ARE POLYNOMIALS OF DEGREE K      1540
C      IER>0 :ABNORMAL TERMINATION.                                          1550
C      IER=1:THE REQUIRED STORAGE SPACE EXCEEDS THE AVAILABLE                 1560
C      STORAGE SPACE,SPECIFIED BY THE PARAMETER NEST.                       1570
C      PROBABLY CAUSES:NEST OR S TOO SMALL.                                  1580
C      IER=2:A THEORETICALLY IMPOSSIBLE RESULT WAS FOUND DURING              1590
C      THE ITERATION PROCESS.                                                1600
C      PROBABLY CAUSES:TOL TOO SMALL                                         1610
C      IER=3:THE MAXIMAL NUMBER OF ITERATIONS MAXIT HAS BEEN                 1620
C      REACHED.                                                                1630
C      PROBABLY CAUSES:MAXIT OR TOL TOO SMALL.                               1640
C      IER=10:SOME OF THE INPUT DATA ARE INVALID(SEE RESTRICTIONS).         1650
C                                                                              1660
C RESTRICTIONS:                                                              1670
C      1) M > K > 0                                                            1680
C      2) ZB <= Z(R) < Z(R+1) <= ZE, R=1,2,...M-1. (IPAR = 1)             1690
C      3) W(R) > 0, R=1,2,...M.                                              1700
C      4) S >= 0.                                                            1710
C      5) NEST >= 2*K+2.                                                     1720
C                                                                              1730
C OTHER SUBROUTINES REQUIRED:                                                 1740
C      BSPLIN,COSSIN,ROTATE,BACK,NKNOT,DISCO AND RATION.                    1750
C      DIMENSION X(M),Y(M),W(M),Z(M),T(200),CX(200),CY(200),
C      < FPINT(200),RX(200),RY(200),DIAG(200),DPRIME(200),
C      < G(200,6),B(200,7),Q(400,6),H(7),NROATA(200),A(200,5)
C COMMON/OPT1/NROATA(NEST),FPO,FPOLD,NPLUS                                  1790
C      NROATA: INTEGER ARRAY,LENGTH NEST,WHICH GIVES THE NUMBER OF          1800
C      DATA POINTS INSIDE EACH KNOT INTERVAL.                              1810
C      FPO : REAL VALUE, WHICH CONTAINS THE SUM(WI*(XI-SX(ZI))**2)+          1820
C      SUM(WI*(YI-SY(ZI))**2) WITH SX(Z) AND SY(Z) LEAST-SQUARES           1830
C      POLYNOMIALS OF DEGREE K.                                             1840
C      FPOLD : REAL VALUE,WHICH CONTAINS THE SUM(WI*(XI-SX(ZI))**2)+         1850
C      SUM(WI*(YI-SY(ZI))**2) WITH SX(Z) AND SY(Z) LEAST-SQUARES           1860
C      SPLINE FUNCTIONS CORRESPONDING TO THE LAST FOUND SET OF             1870
C      KNOTS BUT ONE.                                                       1880
C      NPLUS : INTEGER VALUE,GIVING THE NUMBER OF KNOTS OF THE LAST         1890
C      SET MINUS THE NUMBER OF THE LAST SET BUT ONE.                       1900
C      COMMON/OPT1/NROATA,FPO,FPOLD,NPLUS                                    1910
C DATA INITIALIZATION STATEMENT TO SPECIFY                                 1920
C      TOL : THE REQUESTED RELATIVE ACCURACY FOR THE ROOT OF F(P) = S.      1930
C      MAXIT: THE MAXIMAL NUMBER OF ITERATIONS ALLOWEO.                     1940
C      NEST : AN OVER-ESTIMATE OF THE NUMBER OF KNOTS N. THIS PARAMETER     1950
C      MUST BE SET BY THE USER TO INDICATE THE STORAGE SPACE               1960
C      AVAILABLE TO THE SUBROUTINE. THE DIMENSION SPECIFICATIONS           1970
C      OF THE ARRAYS T,CX,CY,NROATA,FPINT,RX,RY,DIAG,DPRIME(N),            1980
C      A(N,K),G(N,K+1),B(N,K+2),Q(M,K+1) AND H(K+2) DEPEND                1990
C      ON N,M AND K. SINCE N IS UNKNOWN AT THE TIME THE                    2000
C      USER SETS UP THE DIMENSION INFORMATION AN OVER-ESTIMATE            2010
C      OF THESE ARRAYS WILL GENERALLY BE MADE. THE FOLLOWING                2020
C      REMARKS ARE INTENDED TO HELP THE USER                               2030
C      1) 2*K+2 <= N <= M+K+1                                              2040
C      2) THE SMALLER THE VALUE OF S, THE GREATER N WILL BE.               2050
C      3) NORMALLY N = M/2 IS AN OVER-ESTIMATE.                             2060
C      DATA TOL/0.001/,MAXIT/20/,NEST/200/
C BEFORE STARTING COMPUTATIONS A DATA CHECK IS MADE. IF THE INPUT          2080
C DATA ARE INVALID CONTROLE IS IMMEDIATELY REPASSED TO THE DRIVER          2090
C PROGRAM (IER=10).                                                         2100
C      IER = 0                                                                2110
C      K1 = K+1                                                                2120
C      K2 = K1+1                                                              2130
C      NMIN = 2*K1                                                            2140

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IF(K.LE.0) IER = 10	2150
IF(M.LT.K1 .OR. NEST.LT.NMIN) IER = 10	2160
IF(S.LT.0.) IER = 10	2170
IF(IER.NE.0) GO TO 440	2180
C CHECK WHETHER THE Z-VALUES ARE PROVIDED WITH BY THE USER.	2190
IF(IPAR.NE.0) GO TO 6	2200
C FIND FOR EACH DATA POINT A CORRESPONDING VALUE OF THE PARAMETER Z	2210
C AND FIX THE BOUNDARIES ZB AND ZE.	2220
Z(1) = 0.	2230
DO 4 I=2,M	2240
Z(I) = Z(I-1)+SQRT((X(I)-X(I-1))**2+(Y(I)-Y(I-1))**2)	2250
4 CONTINUE	2260
ZB = Z(1)	2270
ZE = Z(M)	2280
6 IF(ZB.GT.Z(1) .OR. ZE.LT.Z(M) .OR. W(1).LE.0.) IER = 10	2290
DO 10 I=2,M	2300
IF(Z(I-1).GE.Z(I) .OR. W(I).LE.0.) IER = 10	2310
10 CONTINUE	2320
IF(IER.NE.0) GO TO 440	2330
C CALCULATION OF ACC, THE ABSOLUTE TOLERANCE FOR THE ROOT OF F(P)=S.	2340
ACC = TOL*S	2350
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC	2360
C PART 1: DETERMINATION OF THE NUMBER OF KNOTS AND THEIR POSITION	2370
C *****	2380
C GIVEN A SET OF KNOTS WE COMPUTE THE LEAST-SQUARES SPLINES SXINF(Z)	2390
C AND SYINF(Z).IF THE SUM F(P=INF)<=S WE ACCEPT THE CHOICE OF KNOTS.	2400
C OTHERWISE WE HAVE TO INCREASE THEIR NUMBER.	2410
C THE INITIAL CHOICE OF KNOTS DEPENDS ON THE VALUE OF S AND IOPT.	2420
C IF S=0 WE HAVE SPLINE INTERPOLATION; IN THAT CASE THE NUMBER OF	2430
C KNOTS EQUALS NMAX = M+K+1.	2440
C IF S > 0 AND	2450
C IOPT=0 WE FIRST COMPUTE THE LEAST-SQUARES POLYNOMIALS OF	2460
C DEGREE K: N = NMIN = 2*K+2	2470
C IOPT=1 WE START WITH THE SET OF KNOTS FOUND AT THE LAST	2480
C CALL OF THE ROUTINE, EXCEPT FOR THE CASE THAT S > FPO; THEN	2490
C WE COMPUTE DIRECTLY THE LEAST-SQUARES POLYNOMIALS OF DEGREE K.	2500
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC	2510
C DETERMINE NMAX, THE NUMBER OF KNOTS FOR SPLINE INTERPOLATION.	2520
NMAX = M+K1	2530
IF(S.GT.0.) GO TO 45	2540
C IF S=0, SX(Z) AND SY(Z) ARE INTERPOLATING SPLINES.	2550
N = NMAX	2560
C TEST WHETHER THE REQUIRED STORAGE SPACE EXCEEDS THE AVAILABLE ONE.	2570
IF(N.GT.NEST) GO TO 420	2580
C FIND THE POSITION OF THE INTERIOR KNOTS IN CASE OF INTERPOLATION.	2590
MK1 = M-K1	2600
IF(MK1.EQ.0) GO TO 30	2610
K3 = K/2	2620
I = K2	2630
J = K3+2	2640
IF(K3#2.EQ.K) GO TO 30	2650
DO 20 L=1,MK1	2660
T(I) = Z(J)	2670
I = I+1	2680
J = J+1	2690
20 CONTINUE	2700
GO TO 60	2710
30 DO 40 L=1,MK1	2720
T(I) = (Z(J)-Z(J-1))/0.5	2730
I = I+1	2740
J = J+1	2750

40	CONTINUE	2760
	GO TO 60	2770
C	IF S>0 OUR INITIAL CHOICE OF KNOTS DEPENDS ON THE VALUE OF IOPT.	2790
C	IF IOPT=0 OR IOPT=1 AND S>=FPO, WE START COMPUTING THE LEAST-SQUARES	2790
C	POLYNOMIALS OF DEGREE K WHICH ARE SPLINES WITHOUT INTERIOR KNOTS.	2800
C	IF IOPT=1 AND FPO>S WE START COMPUTING THE LEAST-SQUARES SPLINES	2810
C	ACCORDING TO THE SET OF KNOTS FOUND AT THE LAST CALL OF THE ROUTINE.	2820
45	IF(IOPT.LE.0) GO TO 50	2830
	IF(FPO.GT.S) GO TO 60	2840
50	N = NMIN	2850
	NRODATA(1) = M-2	2860
C	MAIN LOOP FOR THE DIFFERENT SETS OF KNOTS. M IS A SAVE UPPER BOUND	2870
C	FOR THE NUMBER OF TRIALS.	2880
60	DO 200 ITER = 1,M	2890
	IF(N.EQ.NMIN) IER = -2	2900
C	FIND NRINT, THE NUMBER OF KNOT INTERVALS.	2910
	NRINT = N-NMIN+1	2920
C	FIND THE POSITION OF THE ADDITIONAL KNOTS WHICH ARE NEEDED FOR	2930
C	THE B-SPLINE REPRESENTATION OF SX(Z) AND SY(Z).	2940
	NK1 = N-K1	2950
	I = N	2960
	DO 70 J=1,K1	2970
	T(J) = ZB	2980
	T(I) = ZE	2990
	I = I-1	3000
70	CONTINUE	3010
C	COMPUTE THE B-SPLINE COEFFICIENTS OF THE LEAST-SQUARES SPLINES SXINF(Z)	3020
C	AND SYINF(Z). THE OBSERVATION MATRIX A IS BUILT UP ROW BY ROW AND	3030
C	REDUCED TO UPPER TRIANGULAR FORM BY GIVENS TRANSFORMATIONS	3040
C	WITHOUT SQUARE ROOTS. AT THE SAME TIME FP=F(P=INF) IS COMPUTED	3050
	FP = 0.	3060
C	INITIALIZE THE OBSERVATION MATRIX A.	3070
	DO 80 I=1,NK1	3090
	DIAG(I) = 0.	3090
	RX(I) = 0.	3100
	RY(I) = 0.	3110
	DO 80 J=1,K	3120
	A(I,J) = 0.	3130
80	CONTINUE	3140
	L = K1	3150
	DO 130 IT=1,M	3160
C	FETCH THE CURRENT DATA POINT X(IT),Y(IT),Z(IT).	3170
	XI = X(IT)	3180
	YI = Y(IT)	3190
	ZI = Z(IT)	3200
	WI = W(IT)	3210
C	SEARCH FOR KNOT INTERVAL T(L) <= ZI <= T(L+1).	3220
	IF(ZI.GE.T(L+1) .AND. L.NE.NK1) L = L+1	3230
C	EVALUATE THE (K+1) NON-ZERO B-SPLINES AT ZI AND STORE THEM IN Q.	3240
	CALL BSPLIN(T,N,K,ZI,L,H)	3250
	DO 90 I=1,K1	3260
	IF(H(I).LT.0.1E-07) H(I) = 0.	3270
	Q(IT,I) = H(I)	3280
90	CONTINUE	3290
C	ROTATE THE NEW ROW OF THE OBSERVATION MATRIX INTO TRIANGLE BY	3300
C	GIVENS TRANSFORMATIONS WITHOUT SQUARE ROOTS.	3310
	J = L-K1	3320
	DO 110 I=1,K1	3330
	IF(WI.EQ.0.) GO TO 130	3340
	J = J+1	3350
	PIV = H(I)	3360



IF(PIV.EQ.0.) GO TO 110	3370
C CALCULATE THE PARAMETERS OF THE GIVENS TRANSFORMATION.	3380
CALL COSSIN(PIV,WI,DIAG(J),COS,SIN)	3390
C TRANSFORMATIONS TO RIGHT HAND SIDES.	3400
CALL ROTATE(PIV,COS,SIN,XI,RX(J))	3410
CALL ROTATE(PIV,COS,SIN,YI,RY(J))	3420
IF(1.EQ.K1) GO TO 120	3430
I2 = 0	3440
I3 = I+1	3450
DO 100 I1 = I3,K1	3460
I2 = I2+1	3470
C TRANSFORMATIONS TO LEFT HAND SIDE.	3480
CALL ROTATE(PIV,COS,SIN,H(I1),A(J,I2))	3490
100 CONTINUE	3500
110 CONTINUE	3510
C ADD CONTRIBUTION OF THIS ROW TO THE SUM OF SQUARES OF RESIDUAL	3520
C RIGHT HAND SIDES.	3530
120 FP = FP+WI*(XI**2+YI**2)	3540
130 CONTINUE	3550
IF(IER.EQ.-2) FPO = FP	3560
C BACKWARD SUBSTITUTION TO OBTAIN THE B-SPLINE COEFFICIENTS.	3570
CALL BACK(A,RX,NK1,K,CX)	3580
CALL BACK(A,RY,NK1,K,CY)	3590
C TEST WHETHER THE APPROXIMATION X=SXINF(Z),Y=SYINF(Z) IS AN	3600
C ACCEPTABLE SOLUTION.	3610
FPMS = FP-S	3620
IF(ABS(FPMS).LT.ACC) GO TO 440	3630
C IF F(P=INF) < S ACCEPT THE CHOICE OF KNOTS.	3640
IF(FPMS.LT.0.) GO TO 250	3650
C IF N=NMAX,SXINF(Z) AND SYINF(Z) ARE INTERPOLATING SPLINES.	3660
IF(N.EQ.NMAX) GO TO 430	3670
C INCREASE THE NUMBER OF KNOTS.	3680
C IF N=NEST WE CANNOT INCREASE THE NUMBER OF KNOTS BECAUSE OF	3690
C THE STORAGE CAPACITY LIMITATION.	3700
IF(N.EQ.NEST) GO TO 420	3710
C DETERMINE THE NUMBER OF KNOTS NPLUS WE ARE GOING TO ADD.	3720
IF(IER.EQ.0) GO TO 140	3730
NPLUS = 1	3740
IER = 0	3750
GO TO 150	3760
140 NPL1 = NPLUS*2	3770
IF(FPOLD-FP.GT.ACC) NPL1 = FLOAT(NPLUS)*FPMS/(FPOLD-FP)	3780
NPLUS = MIN0(NPLUS*2,MAX0(NPL1,NPLUS/2,1))	3790
150 FPOLD = FP	3800
C COMPUTE THE SUM(WI*((XI-SXINF(ZI))**2+(YI-SYINF(ZI))**2)) FOR	3810
C EACH KNOT INTERVAL T(J+K) <= ZI <= T(J+K+1) AND STORE IT IN	3820
C FPINT(J),J=1,2,...NRINT.	3830
FPART = 0.	3840
I = 1	3850
L = K2	3860
NEW = 0	3870
DO 180 IT=1,M	3880
IF(Z(IT).LT.T(L).OR.L.GT.NK1) GO TO 160	3890
NEW = 1	3900
L = L+1	3910
160 TERM1 = 0.	3920
TERM2 = 0.	3930
LO = L-K2	3940
DO 170 J=1,K1	3950
LO = LO+1	3960
TERM1 = TERM1+CX(LO)*Q(IT,J)	3970

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      TERM2 = TERM2+CY(L0)*C(IT,J)
170  CONTINUE
      TERM = W(IT)*((TERM1-X(IT))**2+(TERM2-Y(IT))**2)
      FPART = FPART+TERM
      IF(NEW.EQ.0) GO TO 180
      STORE = TERM*0.5
      FPINT(I) = FPART-STORE
      I = I+1
      FPART = STORE
      NEW = 0
180  CONTINUE
      FPINT(NRINT) = FPART
      DO 190 L=1,NPLUS
C   ADD A NEW KNOT.
      CALL NKNJT(Z,M,T,N,FPINT,NRODATA,NRINT)
C   TEST WHETHER WE CANNOT FURTHER INCREASE THE NUMBER OF KNOTS.
      IF(N.EQ.NMAX .OR. N.EQ.NEST) GO TO 200
190  CONTINUE
C   RESTART THE COMPUTATIONS WITH THE NEW SET OF KNOTS.
200  CONTINUE
C   PART 2: DETERMINATION OF THE SMOOTHING SPLINES SXP(Z) AND SYP(Z)
C   THE LEAST-SQUARES KTH DEGREE POLYNOMIALS, IS A SOLUTION OF OUR PROBLEM
250  IF(IER.EQ.-2) GO TO 440
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C   PART 2: DETERMINATION OF THE SMOOTHING SPLINES SXP(Z) AND SYP(Z).
C   *****
C   WE HAVE DETERMINED THE NUMBER OF KNOTS AND THEIR POSITION.
C   WE NOW COMPUTE THE B-SPLINE COEFFICIENTS OF THE SMOOTHING SPLINES
C   SXP(Z) AND SYP(Z). THE OBSERVATION MATRIX A IS EXTENDED BY THE ROWS
C   OF MATRIX B EXPRESSING THAT THE KTH DERIVATIVE DISCONTINUITIES OF
C   SXP(Z) AND SYP(Z) AT THE INTERIOR KNOTS T(K+2),...T(N-K-1) MUST BE
C   ZERO. THE CORRESPONDING WEIGHTS OF THESE ADDITIONAL ROWS ARE SET
C   TO 1/SQRT(P). ITERATIVELY WE THEN HAVE TO DETERMINE THE VALUE OF P
C   SUCH THAT F(P)=SUM(WI*((XI-SXP(ZI))**2+(YI-SYP(ZI))**2) BE = S. WE
C   ALREADY KNOW THAT THE LEAST-SQUARES POLYNOMIALS CORRESPOND TO P=0,
C   AND THAT THE LEAST-SQUARES SPLINES CORRESPOND TO P=INFINITY. THE
C   ITERATION PROCESS WHICH IS PROPOSED HERE, MAKES USE OF RATIONAL
C   INTERPOLATION. SINCE F(P) IS A CONVEX AND STRICTLY DECREASING
C   FUNCTION OF P, IT CAN BE APPROXIMATED BY A RATIONAL FUNCTION
C   R(P) = (U*P+V)/(P+W). THREE VALUES OF P(P1,P2,P3) WITH CORRESPOND-
C   ING VALUES OF F(P) (F1=F(P1)-S,F2=F(P2)-S,F3=F(P3)-S) ARE USED
C   TO CALCULATE THE NEW VALUE OF P SUCH THAT R(P)=S. CONVERGENCE IS
C   GUARANTEED BY TAKING F1>0 AND F3<0.
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C   EVALUATE THE DISCONTINUITY JUMP OF THE KTH DERIVATIVE OF THE
C   B-SPLINES AT THE KNOTS T(L),L=K+2,...N-K-1 AND STORE IN B.
      CALL DISCO(T,N,K2,B)
C   INITIAL VALUE FOR P.
      P1 = 0.
      F1 = FP0-S
      P3 = -1.
      F3 = FPMS
      P = -F1/F3
      ICHECK = 0
      NB = N-NMIN
C   ITERATION PROCESS TO FIND THE ROOT OF F(P) = S.
      DO 350 ITER=1,MAXIT
C   THE ROWS OF MATRIX B WITH WEIGHT 1/SQRT(P) ARE ROTATED INTO
C   THE TRIANGULARISED OBSERVATION MATRIX A WHICH IS STORED IN G.
      PINV = 1.0/P
      DO 260 I=1,NK1

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OPRIME(I) = OIAG(I)	4590
CX(I) = RX(I)	4600
CY(I) = RY(I)	4610
G(I,K1) = 0.	4620
DO 260 J=1,K	4630
G(I,J) = A(I,J)	4640
260 CONTINUE	4650
DO 300 IT=1,N8	4660
C THE ROW OF MATRIX B IS ROTATED INTO TRIANGLE BY GIVENS TRANSFORMATIONS.	4670
DO 270 I=1,K2	4680
H(I) = B(IT,I)	4690
270 CONTINUE	4700
XI = 0.	4710
YI = 0.	4720
WI = PINV	4730
DO 290 J=IT,NK1	4740
IF(WI.EQ.0.) GO TO 300	4750
PIV = H(1)	4760
C CALCULATE THE PARAMETERS OF THE GIVENS TRANSFORMATION.	4770
CALL COSSIN(PIV,WI,OPRIME(J),COS,SIN)	4780
C TRANSFORMATIONS TO RIGHT HAND SIDES.	4790
CALL ROTATE(PIV,COS,SIN,XI,CX(J))	4800
CALL ROTATE(PIV,COS,SIN,YI,CY(J))	4810
IF(J.EQ.NK1) GO TO 300	4820
I2 = K1	4830
IF(J.GT.N8) I2 = NK1-J	4840
DO 280 I=1,I2	4850
C TRANSFORMATIONS TO LEFT HAND SIDE.	4860
CALL ROTATE(PIV,COS,SIN,H(I+1),G(J,I))	4870
H(I) = H(I+1)	4880
280 CONTINUE	4890
H(I2+1) = 0.	4900
290 CONTINUE	4910
300 CONTINUE	4920
C BACKWARD SUBSTITUTION TO OBTAIN THE B-SPLINE COEFFICIENTS.	4930
CALL BACK(G,CX,NK1,K,CX)	4940
CALL BACK(G,CY,NK1,K,CY)	4950
C COMPUTATION OF F(P).	4960
FP = 0.	4970
L = K2	4980
DO 330 IT=1,M	4990
IF(Z(IT).LT.T(L) .OR. L.GT.NK1) GO TO 310	5000
L = L+1	5010
310 L0 = L-K2	5020
TERM1 = 0.	5030
TERM2 = 0.	5040
DO 320 J=1,K1	5050
L0 = L0+1	5060
TERM1 = TERM1+CX(L0)*Q(IT,J)	5070
TERM2 = TERM2+CY(L0)*Q(IT,J)	5080
320 CONTINUE	5090
FP = FP+W(IT)*((TERM1-X(IT))**2+(TERM2-Y(IT))**2)	5100
330 CONTINUE	5110
C TEST WHETHER THE APPROXIMATION X=SXP(Z),Y=SYN(Z) IS AN ACCEPTABLE	5120
C SOLUTION.	5130
FPMS = FP-S	5140
IF(ABS(FPMS).LT.ACC) GO TO 440	5150
C TEST WHETHER THE MAXIMAL NUMBER OF ITERATIONS IS REACHED.	5160
IF(ITER.EQ.MAXIT) GO TO 400	5170
C CARRY OUT ONE MORE STEP OF THE ITERATION PROCESS.	5180
P2 = P	5190

F2 = FPMS	5200
IF(ICHECK.NE.0) GO TO 340	5210
IF((F2-F3).GT.ACC) GO TO 335	5220
C OUR INITIAL CHOICE OF P IS TOO LARGE.	5230
P = P*0.1E-02	5240
P3 = P2	5250
F3 = F2	5260
GO TO 350	5270
335 IF((F1-F2).GT.ACC) GO TO 340	5280
C OUR INITIAL CHOICE OF P IS TOO SMALL	5290
TYPE *, "VALUE OF P", P	
P = P*0.1E+04	5300
P1 = P2	5310
F1 = F2	5320
GO TO 350	5330
C TEST WHETHER THE ITERATION PROCESS PROCEEDS AS THEORETICALLY	5340
C EXPECTED.	5350
340 IF(F2.GE.F1 .OR. F2.LE.F3) GO TO 410	5360
ICHECK = 1	5370
C FIND THE NEW VALUE FOR P.	5380
P = RATION(P1,F1,P2,F2,P3,F3)	5390
350 CONTINUE	5400
C ERROR CODES AND MESSAGES.	5410
400 IER = 3	5420
GO TO 440	5430
410 IER = 2	5440
GO TO 440	5450
420 IER = 1	5460
GO TO 440	5470
430 IER = -1	5480
440 RETURN	5490
END	5500
SUBROUTINE BSPLIN(T,N,K,X,L,H)	5510
C SUBROUTINE BSPLIN EVALUATES THE (K+1) NON-ZERO B-SPLINES OF	5520
C DEGREE K AT T(L) <= X < T(L+1) USING THE STABLE RECURRENCE	5530
C RELATION OF DE BOOR AND COX.	5540
C THE DIMENSION SPECIFICATIONS OF THE FOLLOWING ARRAYS MUST BE	5550
C AT LEAST H(K+1),HH(K).	5560
DIMENSION T(N),H(6),HH(5)	5570
H(1) = 1.	5580
DO 20 J=1,K	5590
DO 10 I=1,J	5600
HH(I) = H(I)	5610
10 CONTINUE	5620
H(1) = 0.	5630
DO 20 I=1,J	5640
LI = L+I	5650
LJ = LI-J	5660
F = HH(I)/(T(LI)-T(LJ))	5670
H(I) = H(I)+F*(T(LI)-X)	5680
H(I+1) = F*(X-T(LJ))	5690
20 CONTINUE	5700
RETURN	5710
END	5720
SUBROUTINE COSSIN(PIV,WI,WW,COS,SIN)	5730
C SUBROUTINE COSSIN CALCULATES THE PARAMETERS OF A GIVENS	5740
C TRANSFORMATION WITHOUT SQUARE ROOTS.	5750
STORE = PIV*WI	5760
OO = WW+STORE*PIV	
IF(ABS(OO).LT.1.E-36) OO=1.E-36	
COS = WW/OO	5770

SIN = STORE/OD	5790
WW = DD	5800
WI = COS*WI	5810
RETURN	5820
END	5830
SUBROUTINE ROTATE(PIV,COS,SIN,A,B)	5840
C SUBROUTINE ROTATE APPLIES A GIVEN ROTATION TO A AND B.	5850
STORE = B	5860
B = COS*STORE+SIN*A	5870
A = A-PIV*STORE	5880
RETURN	5890
END	5900
SUBROUTINE BACK(A,Z,N,K,C)	5910
C SUBROUTINE BACK CALCULATES THE SOLUTION OF THE SYSTEM OF	5920
C EQUATIONS $A \cdot C = Z$ WITH A AN $N \times N$ UNIT UPPER TRIANGULAR MATRIX	5930
C OF BANDWIDTH $K+1$ .	5940
C ATTENTION: THE FIRST DIMENSION SPECIFICATION OF MATRIX A MUST	5950
C BE THE SAME AS IN THE CALLING PROGRAM.	5960
DIMENSION A(200,K),Z(N),C(N)	5970
C(N) = Z(N)	5980
I = N-1	5990
IF(I.EQ.0) GO TO 30	6000
DO 20 J=2,N	6010
STORE = Z(I)	6020
I1 = K	6030
IF(J.LE.K) I1 = J-1	6040
M = I	6050
DO 10 L=1,I1	6060
M = M+1	6070
STORE = STORE-C(M)*A(I,L)	6080
10 CONTINUE	6090
C(I) = STORE	6100
I = I-1	6110
20 CONTINUE	6120
30 RETURN	6130
END	6140
SUBROUTINE NKNOT(X,M,T,N,FPINT,NRODATA,NRINT)	6150
C SUBROUTINE NKNOT LOCATES AN ADDITIONAL KNOT FOR A SPLINE OF DEGREE	6160
C K AND ADJUSTS THE CORRESPONDING PARAMETERS, I.E.	6170
C T : THE POSITION OF THE KNOTS.	6180
C N : THE NUMBER OF KNOTS.	6190
C NRINT : THE NUMBER OF KNOTINTERVALS.	6200
C FPINT : THE SUM OF SQUARES OF RESIDUAL RIGHT HAND SIDES	6210
C FOR EACH KNOT INTERVAL.	6220
C NRODATA: THE NUMBER OF DATA POINTS INSIDE EACH KNOT INTERVAL.	6230
C THE ARRAYS T,FPINT AND NRODATA MUST HAVE THE SAME DIMENSION	6240
C SPECIFICATIONS AS IN THE CALLING PROGRAM.	6250
DIMENSION X(M),T(200),FPINT(200),NRODATA(200)	
K = (N-NRINT-1)/2	6270
C SEARCH FOR KNOT INTERVAL $T(\text{NUMBER}+K) \leq X \leq T(\text{NUMBER}+K+1)$ WHERE	6280
C $FPINT(\text{NUMBER})$ IS MAXIMAL ON THE CONDITION THAT $NRODATA(\text{NUMBER})$	6290
C NOT EQUALS ZERO.	6300
FPMAX = 0.	6310
JBEGIN = 1	6320
DO 20 J=1,NRINT	6330
JPOINT = NRODATA(J)	6340
IF(FPMAX.GE.FPINT(J).OR.JPOINT.EQ.0) GO TO 10	6350
FPMAX = FPINT(J)	6360
NUMBER = J	6370
MAXPT = JPOINT	6380
MAXBEG = JBEGIN	6390

10	JBEGIN = JBEGIN+JPOINT+1	6400
20	CONTINUE	6410
C	LET COINCIDE THE NEW KNOT T(NUMBER+K+1) WITH A DATA POINT X(NRX)	6420
C	INSIDE THE OLD KNOT INTERVAL T(NUMBER+K) <= X <= T(NUMBER+K+1).	6430
	IHALF = MAXPT/2+1	6440
	NRX = MAXBEG+IHALF	6450
	NEXT = NUMBER+1	6460
	IF(NEXT.GT.NRINT) GO TO 40	6470
C	ADJUSTS THE DIFFERENT PARAMETERS.	6480
	DO 30 J=NEXT,NRINT	6490
	JJ = NEXT+NRINT-J	6500
	FPINT(JJ+1) = FPINT(JJ)	6510
	NRDATA(JJ+1) = NRDATA(JJ)	6520
	JK = JJ+K	6530
	T(JK+1) = T(JK)	6540
30	CONTINUE	6550
40	NRDATA(NUMBER) = IHALF-1	6560
	NRDATA(NEXT) = MAXPT-IHALF	6570
	FPINT(NUMBER) = FPMAX*FLOAT(NRDATA(NUMBER))/FLOAT(MAXPT)	6580
	FPINT(NEXT) = FPMAX*FLOAT(NRDATA(NEXT))/FLOAT(MAXPT)	6590
	JK = NEXT+K	6600
	T(JK) = X(NRX)	6610
	N = N+1	6620
	NRINT = NRINT+1	6630
	RETURN	6640
	END	6650
	SUBROUTINE DISCO(T,N,K2,B)	6660
C	SUBROUTINE DISCO CALCULATES THE DISCONTINUITY JUMPS OF THE KTH	6670
C	DERIVATIVE OF THE B-SPLINES OF DEGREE K AT THE KNOTS T(K+2)..T(N-K-1)	6680
C	THE FIRST DIMENSION SPECIFICATION OF THE MATRIX B MUST BE THE SAME AS	6690
C	IN THE CALLING PROGRAM; M MUST HAVE A DIMENSION SPECIFICATION AT	6700
C	LEAST 2*K+2.	6710
	DIMENSION T(N),B(200,K2),M(12)	6720
	K1 = K2-1	6730
	K = K1-1	6740
	NK1 = N-K1	6750
	DO 40 L=K2,NK1	6760
	LMK = L-K1	6770
	DO 10 J=1,K1	6780
	IK = J+K1	6790
	LJ = L+J	6800
	LK = LJ-K2	6810
	M(J) = T(L)-T(LK)	6820
	M(IK) = T(L)-T(LJ)	6830
10	CONTINUE	6840
	LP = LMK	6850
	DO 30 J=1,K2	6860
	JK = J+K	6870
	PROD = 1.	6880
	DO 20 I=J,JK	6890
	PROD = PROD*M(I)	6900
20	CONTINUE	6910
	LK = LP+K1	6920
	B(LMK,J) = (T(LK)-T(LP))/PROD	6930
	LP = LP+1	6940
30	CONTINUE	6950
40	CONTINUE	6960
	RETURN	6970
	END	6980
	FUNCTION RATION(P1,F1,P2,F2,P3,F3)	6990
C	GIVEN THREE POINTS (P1,F1),(P2,F2) AND (P3,F3), FUNCTION RATION	7000



C	GIVES THE VALUE OF P SUCH THAT THE RATIONAL INTERPOLATING FUNCTION	7010
C	OF THE FORM $R(P) = (U*P+V)/(P+W)$ EQUALS ZERO AT P.	7020
	IF(P3.GT.0.) GO TO 10	7030
C	VALUE OF P IN CASE P3 = INFINITY.	7040
	P = (P1*(F1-F3)*F2-P2*(F2-F3)*F1)/((F1-F2)*F3)	7050
	GO TO 20	7060
C	VALUE OF P IN CASE P3 ^= INFINITY.	7070
10	M1 = F1*(F2-F3)	7080
	M2 = F2*(F3-F1)	7090
	M3 = F3*(F1-F2)	7100
	P = -(P1*P2*M3+P2*P3*M1+P3*P1*M2)/(P1*M1+P2*M2+P3*M3)	7110
C	ADJUST THE VALUE OF P1,F1,P3 AND F3 SUCH THAT F1 > 0 AND F3 < 0.	7120
20	IF(F2.LT.0.) GO TO 30	7130
	P1 = P2	7140
	F1 = F2	7150
	GO TO 40	7160
30	P3 = P2	7170
	F3 = F2	7180
40	RATION = P	7190
	RETURN	7200
	END	7210
	FUNCTION DERIV(T,N,C,NK1,NU,ARG,L)	7220
C	FUNCTION DERIV EVALUATES A SPLINE S(X) OF DEGREE K WHICH IS	7230
C	GIVEN IN ITS NORMALIZED B-SPLINE REPRESENTATION OR CALCULATES	7240
C	DERIVATIVES OF ANY SPECIFIED ORDER NU.	7250
C		7260
C	CALLING SEQUENCE	7270
C	VALUE = DERIV(T,N,C,NK1,NU,ARG,L)	7280
C		7290
C	INPUT PARAMETERS:	7300
C	T : ARRAY, MINIMUM LENGTH N, WHICH CONTAINS THE POSITION	7310
C	OF THE KNOTS OF S(X), I.E. THE POSITION OF THE INTERIOR	7320
C	KNOTS T(K+2),...T(N-K-1) AS WELL AS THE POSITION OF THE	7330
C	KNOTS T(1),...T(K+1) AND T(N-K),...T(N) WHICH ARE NEEDED	7340
C	FOR THE B-SPLINE REPRESENTATION.	7350
C	N : INTEGER VALUE GIVING THE TOTAL NUMBER OF KNOTS OF S(X).	7360
C	C : ARRAY, LENGTH NK1, CONTAINING THE B-SPLINE COEFFICIENTS.	7370
C	NK1 : INTEGER VALUE, GIVING THE DIMENSION OF S(X), I.E. NK1 = N-K-1.	7380
C	NU : INTEGER VALUE WHICH GIVES THE ORDER OF THE DERIVATIVE.	7390
C	ARG : REAL VALUE, GIVING THE VALUE OF THE ARGUMENT.	7400
C	L : INTEGER VALUE WHICH SPECIFIES THE POSITION OF THE ARGUMENT	7410
C	I.E. T(L) <= ARG < T(L+1) OR	7420
C	L = NK1 IF ARG = T(NK1+1).	7430
C		7440
C	OUTPUT PARAMETER:	7450
C	VALUE: REAL VALUE, GIVING THE VALUE OF THE NUTH DERIVATIVE OF	7460
C	S(X) AT X = ARG.	7470
C		7480
C	OTHER SUBROUTINES REQUIRED: NONE.	7490
C		7500
C	RESTRICTIONS:	7510
C	1) NU >= 0	7520
C	2) T(K+1) <= ARG <= T(NK1+1)	7530
C	THE DIMENSION SPECIFICATION OF THE ARRAY M MUST BE AT LEAST K+1.	7540
	DIMENSION T(N), C(NK1), M(6)	7550
	DERIV = 0.	7560
	K1 = N-NK1	7570
	IF(NU.LT.0 .OR. NU.GE.K1) RETURN	7580
	DO 100 I=1,K1	7590
	IK = L+I-K1	7600
	M(I) = C(IK)	7610

100	CONTINUE	7620
	IF(NU.EQ.0) GO TO 300	7630
	NU1 = NU+1	7640
	DO 200 J=2,NU1	7650
	DO 200 JJ=J,K1	7660
	I = J+K1-JJ	7670
	LI = L+I-K1	7680
	LJ = L+I-J+1	7690
	H(I) = (H(I)-H(I-1))/(T(LJ)-T(LI))	7700
200	CONTINUE	7710
	IF(NU.EQ.K1-1) GO TO 500	7720
300	NU2 = NU+2	7730
	DO 400 J=NU2,K1	7740
	DO 400 JJ=J,K1	7750
	I = J+K1-JJ	7760
	LI = L+I-K1	7770
	LJ = L+I-J+1	7780
	H(I) = ((ARG-T(LI))*H(I)+(T(LJ)-ARG)*H(I-1))/(T(LJ)-T(LI))	7790
400	CONTINUE	7800
500	DERIV = H(K1)	7810
	IF(NU.EQ.0) RETURN	7820
	DO 600 I=1,NU	7830
	DERIV = DERIV*FLOAT(K1-I)	7840
600	CONTINUE	7850
	RETURN	7860
	END	7870

APPENDIX D  
THE PROGRAM TO FIND B-SPLINE COEFFICIENT

```
PROGRAM BSPLINE(INPUT,OUTPUT,INFILE,OUTFILE);
```

```
TYPE
```

```

  BYTE = 0..255;
  IMAGEROW1 = PACKED ARRAY [0..255] OF BYTE;
  ROW1 = PACKED ARRAY [0..257] OF BYTE;
  SHIP = PACKED ARRAY [0..63,0..255] OF BYTE;
  OA1 = PACKED ARRAY [0..512] OF REAL;
  OA2 = PACKED ARRAY [1..300] OF REAL;
VAR
  R,C,F0,F1,F2,F3,F4,F5,F6,F7,F8 : BYTE;
  F : ARRAY [0..65] OF ROW1;
  A,IMAGE : SHIP;
  INFILE : FILE OF IMAGEROW1;
  SPX,SPY,SPX1,SPY1 : PACKED ARRAY [0..512] OF REAL;
  R1,C1,R2,C2,COIR,I,J,J1,COUNT,NEG : INTEGER;
  TEMPX,TEMPY,RA : REAL;
  M,IOPT,K,IPAR,N,IER,ANS,NU,ANS1,ANS2,ANS3 : INTEGER;
  NK1,NEND,L,MET : INTEGER;
  W,Z : PACKED ARRAY [0..512] OF REAL;
  S,ZB,ZE,FP : REAL;
  ARG,THETA,TOLE : REAL;
  FLAG_BEGIN,FLAG_END,AK,LUMP : INTEGER;
  BEG,EN : PACKED ARRAY [1..5] OF REAL;
  BEGN,ENN : PACKED ARRAY [1..5] OF INTEGER;
  T,CX,CY : PACKED ARRAY [1..300] OF REAL;
  COL,ROW : PACKED ARRAY [0..512] OF REAL;
  X,Y : PACKED ARRAY [0..512] OF REAL;
  OCY,OCX,ARE,CY1,OMAX,OMIN : REAL;
  CYMIN,CYMAX,MAXCY : PACKED ARRAY [1..5] OF REAL;
  AREA : PACKED ARRAY [1..5] OF REAL;
  OUTFILE : FILE OF IMAGEROW1;
  NAME : PACKED ARRAY [1..20] OF CHAR;
  PEAK,STAR,TER : PACKED ARRAY [1..5] OF REAL;
```

```
(* FILTER THE POINTS *)
```

```
PROCEDURE STORE;
```

```
BEGIN
```

```
  TEMPX:=C-1; TEMPY:=64-(R-1);
```

```
  IF I>0 THEN BEGIN
```

```
    FOR J:=0 TO M-1 BEGIN
```

```
      IF (COL[J]=TEMPX) AND (ROW[J]=TEMPY)
```

```
      THEN I:=J;
```

```
    END;
```

```
    COL[I] :=TEMPX;
```

```
    ROW[I] :=TEMPY;
```

```
    M:=I;
```

```
    I:=I+1;
```

```
  END
```

```
  ELSE BEGIN
```

```
    COL[I] :=TEMPX;
```

```
    ROW[I] :=TEMPY;
```

```
    M:=0;
```

```
    I:=1;
```

```
  END;
```

```
END;
```

PROCEDURE CMOVE;

BEGIN

```
F0 := FCR-1,C]; F1:=FCR-1,C+1]; F2:=FCR,C+1];  
F3:=FCR+1,C+1]; F4:=FCR+1,C]; F5:=FCR+1,C-1];  
F6:=FCR,C-1]; F7:=FCR-1,C-1];
```

CASE COIR OF

0: BEGIN

```
IF (F3=0) AND (F2=255) THEN BEGIN C:=C+1;  
COIR:=1;
```

END

```
ELSE IF (F2=0) AND (F1=255) THEN BEGIN R:=R-1;  
STORE; C:=C+1; COIR:=0; END
```

```
ELSE IF (F1=0) AND (F0=255) THEN BEGIN R:=R-1;  
COIR:=0; END
```

```
ELSE IF (F0=0) AND (F7=255) THEN BEGIN C:=C-1;  
STORE; R:=R-1; COIR:=3; END
```

```
ELSE IF (F7=0) AND (F6=255) THEN BEGIN C:=C-1;  
COIR:=3; END
```

```
ELSE IF (F6=0) AND (F5=255) THEN BEGIN R:=R+1;  
STORE; C:=C-1; COIR:=3; END
```

```
ELSE BEGIN R:=R+1; COIR:=2; END;
```

END;

1: BEGIN

```
IF (F5=0) AND (F4=255) THEN BEGIN R:=R+1;  
COIR:=2; END
```

```
ELSE IF (F4=0) AND (F3=255) THEN BEGIN C:=C+1;  
STORE; R:=R+1; COIR:=1; END
```

```
ELSE IF (F3=0) AND (F2=255) THEN BEGIN C:=C+1;  
COIR:=1; END
```

```
ELSE IF (F2=0) AND (F1=255) THEN BEGIN R:=R-1;  
STORE; C:=C+1; COIR:=0; END
```

```
ELSE IF (F1=0) AND (F0=255) THEN BEGIN R:=R-1;  
COIR:=0; END
```

```
ELSE IF (F0=0) AND (F7=255) THEN BEGIN C:=C-1;  
STORE; R:=R-1; COIR:=0; END
```

```
ELSE BEGIN C:=C-1; COIR:=3; END;
```

END;

2: BEGIN

```
IF (F7=0) AND (F6=255) THEN BEGIN C:=C-1;  
COIR:=3; END
```

```
ELSE IF (F6=0) AND (F5=255) THEN BEGIN R:=R+1;  
STORE; C:=C-1; COIR:=3; END
```

```
ELSE IF (F5=0) AND (F4=255) THEN BEGIN R:=R+1;  
COIR:=2; END
```

```
ELSE IF (F4=0) AND (F3=255) THEN BEGIN C:=C+1;  
STORE; R:=R+1; COIR:=1; END
```

```
ELSE IF (F3=0) AND (F2=255) THEN BEGIN C:=C+1;  
COIR:=1; END
```

```
ELSE IF (F2=0) AND (F1=255) THEN BEGIN R:=R-1;  
STORE; C:=C+1; COIR:=1; END
```

```
ELSE BEGIN R:=R-1; COIR:=0; END;
```

END;

3: BEGIN

```
IF (F1=0) AND (F0=255) THEN BEGIN R:=R-1;  
COIR:=0; END
```

```
ELSE IF (F0=0) AND (F7=255) THEN BEGIN C:=C-1;
```

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```
        STORE; R:=R-1; COIR:=0; END
      ELSE IF (F7=0) AND (F6=255) THEN BEGIN C:=C-1;
        COIR:=3; END
      ELSE IF (F6=0) AND (F5=255) THEN BEGIN R:=R+1;
        STORE; C:=C-1; COIR:=2; END
      ELSE IF (F5=0) AND (F4=255) THEN BEGIN R:=R+1;
        COIR:=2; END
      ELSE IF (F4=0) AND (F3=255) THEN BEGIN C:=C+1;
        STORE; R:=R+1; COIR:=2; END
      ELSE BEGIN C:=C+1; COIR:=1; END;
    END;
  END;
END;
```

```
PROCEDURE INITIAL:
  BEGIN
    FOR I :=0 TO 255 DO BEGIN
      X[I] :=0;
      Y[I] :=0;
    END;
    I:=0;
  END;
```

```
PROCEDURE FIRST:
  VAR
    N,M :INTEGER;
  BEGIN
    N := 0; M:=0;
    FOR C :=0 TO 200 DO BEGIN
      FOR R :=0 TO 63 DO BEGIN
        IF (F[R,C]=255) AND (N=0) THEN BEGIN
          C1 := C; R1 :=R; N := N+1;
        END;
        IF (F[R,255-C]=255) AND (M=0) THEN BEGIN
          C2 := 255-C; R2 := R; M :=M+1;
        END;
      END;
    END;
  END;
```

```
PROCEDURE ROTATE:
  BEGIN
    NEG:=0;
    IF R1=R2 THEN BEGIN THETA:=0.0;
      FOR I:=0 TO M DO BEGIN
        X[I]:=COL[I]; Y[I]:=ROW[I];
      END;
    END
  ELSE THETA:=ABS(ARCTAN((R2-R1)/(C2-C1)));
  IF (THETA<>0) AND (R2>R1) THEN BEGIN
    FOR I:=0 TO M DO BEGIN
      X[I]:=COL[I]*COS(THETA)-ROW[I]*SIN(THETA);
      Y[I]:=COL[I]*SIN(THETA)+ROW[I]*COS(THETA);
      IF Y[I]>63.0 THEN NEG:=NEG+1;
    END;
  END;
```



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END
ELSE IF (THETA<>0) AND (R2<R1) THEN BEGIN
  FOR I:=0 TO M DO BEGIN
    X[I]:=COL[I]*COS(THETA)+ROW[I]*SIN(THETA);
    Y[I]:=-COL[I]*SIN(THETA)+ROW[I]*COS(THETA);
    IF Y[I]>=63.0 THEN NEG:=NEG+1;
  END;
END;
(* FILTER *)
I:=0;
FOR K:=0 TO M DO BEGIN
  TEMPX:=X[K]; TEMPY:=Y[K];
  IF I>0 THEN BEGIN
    FOR J:=0 TO M DO BEGIN
      IF (X[J]=TEMPX) AND (Y[J]=TEMPY) THEN I:=J;
    END;
    X[I]:=TEMPX; Y[I]:=TEMPY; M:=I; I:=I+1;
  END
  ELSE BEGIN X[I]:=TEMPX; Y[I]:=TEMPY; M:=0; I:=1;
  END;
END;
END;

PROCEDURE PARAM( X:DA1; Y:DA1; VAR Z:DA1; W:DA1;
  M:INTEGER; VAR ZB:REAL; VAR ZE:REAL; K:INTEGER;
  S:REAL; VAR N:INTEGER; VAR T:DA2; VAR CX:DA2;
  VAR CY:DA2; VAR FP:REAL; IOPT:INTEGER;
  IPAR:INTEGER; VAR IER:INTEGER); FORTRAN;
PROCEDURE INITT(SPEED:INTEGER); FORTRAN;
PROCEDURE VWINDOW(XMIN:REAL; X RANGE:REAL; YMIN:REAL;
  Y RANGE:REAL); FORTRAN;
PROCEDURE MOVEA(X:REAL; Y:REAL); FORTRAN;
PROCEDURE DRAWA(X:REAL; Y:REAL); FORTRAN;
PROCEDURE ANCHO(ICAR:INTEGER); FORTRAN;
PROCEDURE FINITT(I1:INTEGER; I2:INTEGER); FORTRAN;
PROCEDURE DASHA(X:REAL; Y:REAL; L:INTEGER); FORTRAN;

FUNCTION DERIV(%REF T:DA2; N:INTEGER; CX:DA2;
  NK1:INTEGER; NU:INTEGER; ARG:REAL;
  L:INTEGER) :REAL; FORTRAN;
PROCEDURE SPLCOEF;
BEGIN
  I:=5; LUMP:=0;
  WHILE (I<=N-5) AND (LUMP=0) DO BEGIN
    IF (CY[I-1]>CY[I]) AND (CY[I+1]>CY[I]) AND
      (CY[I+2]>CY[I+1]) THEN
      LUMP:=1;
    I:=I+1;
  END;
  IF LUMP=1 THEN BEGIN
    FLAG_BEGIN:=0; AK:=1; FLAG_END:=0; I:=5;
    WHILE I<=N-5 DO BEGIN
      IF FLAG_BEGIN=0 THEN BEGIN
        IF (CY[I-1]>CY[I]) AND (CY[I+1]>CY[I]) AND
          (CY[I+2]>CY[I+1]) THEN BEGIN
          BEG[AK]:=T[I]; FLAG_BEGIN:=1; BEG[AK]:=I;
        END;
      END;
    END;
  END;

```

```

END
ELSE IF FLAG_BEGIN=1 THEN BEGIN
  IF (CY[I-1]>=CY[I]) AND (CY[I]>=CY[I+1]) AND
    (CY[I+2]>=CY[I+1]) THEN BEGIN
    EN[AK]:=T[I+1]; FLAG_END:=1; ENN[AK]:=I+1;
  END;
END;
IF (FLAG_BEGIN=1) AND (FLAG_END=1) THEN BEGIN
  FLAG_BEGIN:=0; FLAG_END:=0; AK:=AK+1; I:=I-1;
END;
I:=I+1;
END;
END
ELSE IF LUMP=0 THEN BEGIN
  FLAG_BEGIN:=0; FLAG_END:=0; I:=5; AK:=1;
  WHILE I<=N-5 DO BEGIN
    IF FLAG_BEGIN=0 THEN BEGIN
      IF (CY[I-1]>=CY[I]) AND (CY[I+1]>CY[I]) AND
        (CY[I+2]>CY[I]) THEN BEGIN
        BEG[AK]:=T[I]; FLAG_BEGIN:=1; BEGN[AK]:=I;
      END;
    END
    ELSE IF FLAG_BEGIN=1 THEN BEGIN
      IF (CY[I-1]=CY[I]) AND ((CY[I]<=CY[I+1]+TOLE)
        AND (CY[I]>=CY[I+1]-TOLE)) AND
        ((CY[I]<=CY[I+2]+TOLE) AND
        (CY[I]>=CY[I+2]-TOLE)) THEN BEGIN
        EN[AK]:=T[I]; FLAG_END:=1; ENN[AK]:=I;
      END;
    END;
    IF (FLAG_BEGIN=1) AND (FLAG_END=1) THEN BEGIN
      FLAG_BEGIN:=0; FLAG_END:=0; AK:=AK+1; I:=I-1;
    END;
    I:=I+1;
  END;
  IF FLAG_BEGIN=1 THEN BEGIN
    I:=BEGN[AK];
    WHILE (I<=N-5) AND (FLAG_END=0) DO BEGIN
      IF (CY[I-1]=CY[I]) AND
        ((CY[I]<=CY[I+1]+TOLE)
        AND (CY[I]>=CY[I+1]-TOLE)) THEN BEGIN
        EN[AK]:=T[I]; ENN[AK]:=I; FLAG_END:=1;
      END;
      I:=I+1;
    END;
  END;
END;
END;
END;
PROCEDURE ALUMP;
BEGIN
  AK:=1;
  WHILE (ENN[AK])>0 DO BEGIN
    AREA[AK]:=0; CYMAX[AK]:=0.0; CYMIN[AK]:=1000.0;
    FOR I:=(BEGN[AK]) TO (ENN[AK]) DO BEGIN
      IF CYMAX[AK] <= CY[I] THEN BEGIN
        CYMAX[AK]:=CY[I]; MAXCY[AK]:=T[I]

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      END;
      IF CYMIN[AK] >= CY[I] THEN
        CYMIN[AK]:=CY[I];
      END;
      CY1:=CY[BEGN[AK]]-CYMIN[AK];
      FOR I:=(BEGN[AK]) TO (ENN[AK]-1) DO BEGIN
        DCY:=CY[I+1]-CY[I];
        DCX:=CX[I+1]-CX[I];
        AREA[AK]:=AREA[AK]+CY1*DCX+DCY*DCX/2.0;
        WRITELN("AREA=",AREA[AK], " I=",I, " AK=",AK);
        CY1:=CY1+DCY;
      END;
      AK:=AK+1;
    END;
  END;
  PROCEDURE POSINX;
  BEGIN
    I:=0; MET:=0;
    WHILE (I<=M-1) AND (MET=0) DO BEGIN
      IF TEMPX=Z[I] THEN BEGIN
        TEMPY:=X[I]; MET:=1;
      END;
      I:=I+1;
    END;
  END;
  END;

(*****
(* MAIN PROGRAM *)
  BEGIN
    WRITELN("INPUT CUT OR CONLINE FILENAME");
    READLN(NAME);
    OPEN (INFILE,NAME,HISTORY := OLD,
      ACCESS_METHOD := SEQUENTIAL,
      RECORD_LENGTH := 256,RECORD_TYPE := FIXED);
    RESET(INFILE);
    R := 0;
    WHILE NOT EOF (INFILE) DO
      BEGIN
        READ (INFILE,IMAGE[R]);
        FOR C := 0 TO 255 DO
          F[R+1,C+1] := IMAGE[R,C];
        R := R+1;
      END;
    CLOSE(INFILE);
    FIRST;
    R := R1; C := C1; COIR := 1;
    INITIAL;
    WHILE (C<>C2) DO BEGIN
      STORE;
      CMOVE;
    END;
    ROTATE;
    FOR I:=0 TO M DO BEGIN
      IF NEG<>0 THEN Y[I]:=Y[I]-20.0;
    END;
    M:=M+1; K:=2; S:=0.1; IOPT:=0; IPAR:=0;
    FOR I:=0 TO M DO

```

```

WCII:=1.0;
ANS:=1;
WHILE ANS=1 DO BEGIN
  WRITELN("OLD S=",S,"NEW S=");
  READ(S);
  WRITELN("OLD K=",K,"K=");
  READ(K);
  PARAM(X,Y,Z,W,M,ZB,ZE,K,S,N,T,CX,CY,FP,IOP,
    IPAR,IER);
  WRITELN(" S=",S," IER=",IER," M=",M," N=",N,
    " CX[N]=",CX[N]," CY[N-4]=",CY[N-4]);
  WRITELN(" PRINTING YES=1 PLOT X-Y=2,CX,CY=3 ",
    "X-Y-CX-CY=4"," NO=5");
  READ(ANS1);
  IF ANS1=1 THEN BEGIN
    FOR I:=1 TO N DO
      WRITELN("CX=",CX[I]," CY=",CY[I],
        " T=",T[I]);
    END
  ELSE IF ANS1=2 THEN BEGIN
    INITT(960);
    VWINDOW(0.0,Z[M-1],0.0,256.0);
    MOVEA(Z[0],X[0]);
    FOR I:=0 TO M-1 DO
      ORAWA(Z[I],X[I]);
    MOVEA(Z[0],Y[0]);
    FOR I:=0 TO M-1 DO
      OASHA(Z[I],Y[I],2);
    FINITT(0,0);
    END
  ELSE IF ANS1=4 THEN BEGIN
    INITT(960);
    VWINDOW(0.0,Z[M-1],0.0,256.0);
    MOVEA(Z[0],X[0]);
    FOR I:=0 TO M-1 DO
      ORAWA(Z[I],X[I]);
    MOVEA(Z[0],Y[0]);
    FOR I:=0 TO M-1 DO
      ORAWA(Z[I],Y[I]);
    MOVEA(T[4],CX[4]);
    FOR I:=4 TO N-4 DO
      OASHA(T[I],CX[I],2);
    FOR I:=4 TO N-4 DO BEGIN
      MOVEA(T[I]-1.5,CX[I]-1.5);
      ANCHO(111);
    END;
    MOVEA(T[4],CY[4]);
    FOR I:=4 TO N-4 DO
      OASHA(T[I],CY[I],2);
    FOR I:=4 TO N-4 DO BEGIN
      MOVEA(T[I]-1.5,CY[I]-1.5);
      ANCHO(111);
    END;
    OMIN:=0.0; OMAX:=256.0; TEMPX:=OMIN;
    WHILE TEMPX<=Z[M-1] DO BEGIN
      MOVEA(TEMPX,OMIN);
      ORAWA(TEMPX,OMIN);

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        DRAWA(TEMPX,DMAX);
        TEMPX:=TEMPX+10.0;
    END;
    OMIN:=0.0; OMAX:=Z[M-1]; TEMPX:=DMIN;
    WHILE TEMPX<=256.0 DO BEGIN
        MOVEA(DMIN,TEMPX);
        DRAWA(DMIN,TEMPX);
        DRAWA(DMAX,TEMPX);
        TEMPX:=TEMPX+10.0;
    END;
    FINITT(0,0);
END
ELSE IF ANS1=3 THEN BEGIN
    INITT(960);
    VWINDO(0.0,T[N],0.0,256.0);
    MOVEA(T[4],CX[4]);
    FOR I:=4 TO N-4 DO
        DRAWA(T[I],CX[I]);
    FOR I:=4 TO N-4 DO BEGIN
        MOVEA(T[I]-1.5,CX[I]-1.5);
        ANCHO(111);
    END;
    MOVEA(T[4],CY[4]);
    FOR I:=4 TO N-4 DO
        DASHA(T[I],CY[I],2);
    FOR I:=4 TO N-4 DO BEGIN
        MOVEA(T[I]-1.5,CY[I]-1.5);
        ANCHO(111);
    END;
    FINITT(0,0);
END;
(* IMPORTANT FORTRAN DECLEAR FROM 1 *)
NK1:=N-K-1; L:=K+1; NEND:=N-K; J1:=0;
FOR I:=L TO NEND DO BEGIN
    ARG:=T[I];
    WHILE(ARG>=T[L+1]) AND (L<NK1) DO
        L:=L+1;
    SPX[J1]:= DERIV(T,N,CX,NK1,0,ARG,L);
    SPY[J1]:= DERIV(T,N,CY,NK1,0,ARG,L);
    J1:=J1+1;
END;
J:=0; L:=K+1;
FOR I:=L TO NEND DO BEGIN
    ARG:=T[I];
    WHILE (ARG>=T[L+1]) AND (L<NK1) DO
        L:=L+1;
    TEMPX:=T[I+1]-ARG;
    IF TEMPX>=7.0 THEN BEGIN
        WHILE ARG<T[L+1] DO BEGIN
            SPX[J] := DERIV(T,N,CX,NK1,0,ARG,L);
            SPY[J] := DERIV(T,N,CY,NK1,0,ARG,L);
            ARG:=ARG+2.0; J:=J+1;
        END;
    END
ELSE BEGIN
    SPX[J] := DERIV(T,N,CX,NK1,0,ARG,L);
    SPY[J] := DERIV(T,N,CY,NK1,0,ARG,L);

```

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```
J:=J+1;
END;
END;
WRITELN("DO YOU WANT PLOTTING YES=1 ",
"PRINT=2 COMPARE=3 NO=4");
READ(ANS2);
IF ANS2=1 THEN BEGIN
  INITT(960);
  VWINDOW(0.0,256.0,0.0,256.0);
  MOVEA(X[0],Y[0]);
  FOR I:=0 TO M-1 DO
    DRAWA(X[I],Y[I]);
  FOR I:=0 TO J1-1 DO BEGIN
    MOVEA(SPX1[I]-1.5,SPY1[I]-1.5);
    ANCHO(111);
  END;
  MOVEA(SPX[0],SPY[0]);
  FOR I:=1 TO J-1 DO
    QASHA(SPX[I],SPY[I],2);
  FINITT(0,0);
END
ELSE IF ANS2=2 THEN BEGIN
  FOR I:=0 TO J1-1 DO
    WRITELN("SPX1=",SPX1[I],
"      SPY1=",SPY1[I]);
END
ELSE IF ANS2=3 THEN BEGIN
  WRITELN("TOL=");
  READ(TOLE);
  SPLCOEF;
  ALUMP;
  AK:=1; RA:=X[M-1]-X[0];
  WHILE ENCAK>0 DO BEGIN
    TEMPX:=BEGCAK; POSINX;
    STARCAK:=((TEMPY-X[0])-RA/2)/RA;
    TEMPX:=ENCAK; POSINX;
    TERCAK:=((TEMPY-X[0])-RA/2)/RA;
    TEMPX:=MAXCY[AK]; POSINX;
    PEAKCAK:=((TEMPY-X[0])-RA/2)/RA;
    AREACAK:=(AREACA[AK])/(RA**2);
    WRITELN("BEGIN=",STARCAK," ENO=",
TERCAK," TOTAL=",RA,
" AREA=",AREACA[AK],
" PEAK=",PEAKCAK);
    AK:=AK+1;
  END;
END;
WRITELN("DO YOU WANT RUN AGAIN YES=1",
" NO=2");
READ(ANS);
IF ANS=1 THEN BEGIN
  WRITELN("IOPT=");
  READ(IOPT);
END;
END; (* WHILE *)
END.
```



## LIST OF REFERENCES

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